

# *Analysis of Rigid Frames in Space by Applying Slope-Deflection Formulas*

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(Received December 23, 1961)

**Synopsis.** Rigid frames in space are successfully and exactly analysed by applying the slope-deflection equations to the flexural members and the torsion equations to the twisted members respectively. The results are compared to those obtained from the conventional two-dimensional analysis, the precision of which is numerically shown.

## **Introduction**

In designing a rigid frame in space, it is usual to divide it up into several constituent plane frames and analyse each separately. Although this conventional procedure greatly facilitates the calculations, it does not determine the torsion effects which exist and sometimes amount to considerable magnitude. It is hoped to give an exact and easily applicable method which treats the frame as a whole.

Several three-dimensional analyses ever proposed<sup>9), 10), 12)</sup> are almost based upon the principles of virtual work or the like which are taking the stress functions as redundants. Accordingly the calculations become so tedious that the practical applications are limited to the comparatively simple problems, e.g., the symmetrical frames under symmetrical loadings or, if not symmetrical, joints are assumed not to translate.

The author presents here a three-dimensional analysis using the slope-deflection equations together with the torsion equations. It is well known that the slope-deflection method is so superior to the classical methods in treating the plane frames.<sup>11)</sup> This circumstance agrees equally with the space frames,<sup>13)</sup> and many advantages will be claimed in the illustrations which follow.

The space frames mainly considered here are of single-storied and of multi-bayed longitudinally and laterally, whose members are all straight and prismatic meeting each other at right angles.

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The author is heartily grateful to Prof. Tadashi MURAKAMI of the Kyushu University for his kind instructions and constant encouragements in the whole course of this work.

## Chapter I. Fundamental Formulas

### 1. Notations

In dealing with a space frame, the slope-deflection method now familiar to us will also be the most convenient, since it takes the end-deformations of members for unknowns instead of stresses or reactions as in the other classical methods. On this account, the number of unknowns is greatly diminished, simple and clear sign conventions are established, the stress diagrams are made easy to draw, and moreover the deformed structure is readily visualized. Now, we proceed with illustrations of notations adopted.

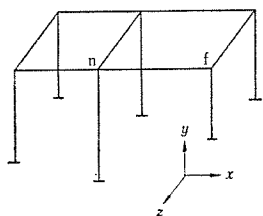


Fig. 1

The members of a space frame in Fig. 1 meet each other at right angles. In this connection, the orthogonal coordinate axes  $x$ ,  $y$  and  $z$  are set up as shown. Note that any member lies parallel to one of these three axes. The member will be designated by its ends, being called the near end  $n$  and the far end  $f$ .

Consider, for example, a member  $nf$  which is parallel to  $x$  axis, Fig. 1. Moments acting at its end are denoted by  $M$  with three subscripts such as  $M_{xnf}$ ,  $M_{ynf}$  and  $M_{znf}$ . The first subscript designates the axis about which  $M$  acts; the remaining two, the member and the end in question as usual. Thus, we readily tell that  $M_{xnf}$  is the moment at the near end about (an axis parallel to)  $x$  axis, i.e., the twisting moment;  $M_{ynf}$  and  $M_{znf}$  are the moments at the near end about  $y$  and  $z$  axes respectively, i.e., the bending moments.

The angular deflections produced by  $M$ 's are denoted by  $\theta$  with subscripts similarly as above. Thus,  $\theta_{xnf}$  is the angular deflection at  $n$  about  $x$  axis, i.e., the angle of twist;  $\theta_{ynf}$  and  $\theta_{znf}$  are the angular deflections at  $n$  about  $y$  and  $z$  axes respectively, i.e., the end-slopes. Since the joint rotations (§9) are always taken up instead of end-slopes in our problems, the last subscript of  $\theta$  designating the far end will often be omitted for convenience. For the member lying parallel to the axes other than  $x$ , the designations of  $M$ 's and of  $\theta$ 's will be made by reading their first subscripts.

$M$ 's are positive which agree with the "right-hand screw rule" when the

arrow-heads point negative directions of reference axes;  $\theta$ 's are positive which correspond to positive  $M$ 's.

The relative displacements of joints tend the member to revolve from its original unstrained position. These angles of revolution are denoted by  $R$ , similarly accompanied by three subscripts; thus, by  $R_{ynf}$  and  $R_{znf}$  are meant that the revolutions are around  $y$  and  $z$  axes respectively. A clockwise revolution is taken as positive when the "center line of screw driver" through the member-end points to the negative direction of the reference axis.

## 2. Slope-Deflection Equations

If the member  $nf$  lies parallel to  $x$  axis, then it undergoes bending deformations in the planes  $xz$  and  $xy$ , and for the member  $nf$  the well known slope-deflection equations hold. See Fig. 2. In  $xz$  plane, we have

$$M_{ynf} = 2E \frac{I_{ynf}}{l_{nf}} (2\theta_{yn} + \theta_{yf} - 3R_{ynf}) + C_{ynf}, \quad (1)$$

and in  $xy$  plane

$$M_{znf} = 2E \frac{I_{znf}}{l_{nf}} (2\theta_{zn} + \theta_{zf} - 3R_{znf}) + C_{znf}, \quad (2)$$

where  $E$  = the modulus of elasticity.

$I_y, I_z$  = the moments of inertia of the section referred to  $y$  and  $z$  axes respectively.

$C_y, C_z$  = the fixed-end bending moments about  $y$  and  $z$  axes respectively. Their values will be found in many texts or pocket-books.

Introducing the stiffness factors for flexure, these equations are modified:

$$M_{ynf} = 2EK_{ynf} (2\theta_{yn} + \theta_{yf} - 3R_{ynf}) + C_{ynf}, \quad (3)$$

$$M_{znf} = 2EK_{znf} (2\theta_{zn} + \theta_{zf} - 3R_{znf}) + C_{znf}, \quad (4)$$

where  $K_y = I_y/l_{nf}$  = the stiffness factor for flexure referred to  $y$  axis,

$K_z = I_z/l_{nf}$  = do. referred to  $z$  axis.

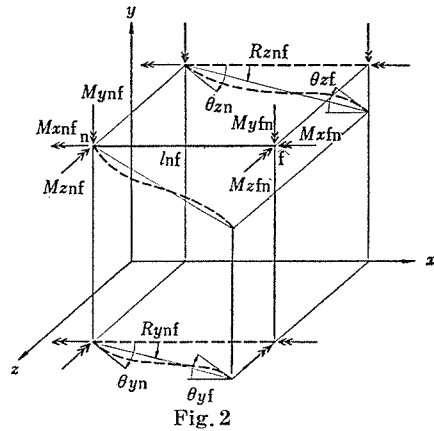


Fig. 2

### 3. Torsion Equations<sup>1), 2), 3), 4), 5)</sup>

We have, for a prismatic bar of length  $l$ , the relationship between the angle of twist  $\theta$  of an end relative to the other and the applied torque  $M_T$ :

$$\theta = \frac{M_T l}{GJ}, \quad (5)$$

where  $G$  = the modulus of rigidity,

$J$  = the torsional constant of the section.

Applying this to the member  $nf$  which lies, for example, parallel to  $x$  axis, Fig. 2, we obtain

$$M_{xnf} = \frac{GJ_{xnf}}{l_{nf}} (\theta_{xn} - \theta_{xf}). \quad (6)$$

If the member is loaded with torques as in Fig. 3, their effects must be added to the right-hand side of eq. (6), which will be called the fixed-end torque and denoted by  $C_{xnf}$ . Thus

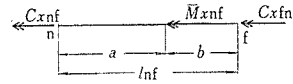


Fig. 3

$$M_{xnf} = \frac{GJ_{xnf}}{l_{nf}} (\theta_{xn} - \theta_{xf}) + C_{xnf}, \quad (7)$$

where

$$C_{xnf} = - \sum \frac{b}{l_{nf}} \bar{M}_{xnf}. \quad (8)$$

Eq. (7) is written in the form

$$M_{xnf} = GK_{xnf} (\theta_{xn} - \theta_{xf}) + C_{xnf}, \quad (9)$$

where  $K_{xnf} = J_{xnf}/l_{nf}$  = the stiffness factor for torsion.

### 4. Evaluation of $J$

The torsional constant  $J$  must be evaluated from the St. Venant's theory. We see that, if the section is circular,  $J$  is equal to  $I_p$ , the polar moment of inertia. But for non-circular sections, being the theory too long and complicated to evaluate it exactly, approximations are mostly being made. Only the formulas for the sections which seem to be important for our problems will be shown below:

1) Rectangular section of width  $a$  and depth  $b$  :

The St. Venant's exact expression is

$$J = \frac{a^3b}{3} \left( 1 - \frac{192}{\pi^5} \frac{a}{b} \sum_{n=1,3,5,\dots}^{\infty} \frac{1}{n^5} \tanh \frac{n\pi b}{2a} \right), \quad (10)$$

which is not so tedious to compute, because the series is very rapidly convergent. For practical purposes, we may take its first term only, thus

$$J = \frac{a^3b}{3} \left( 1 - \frac{192}{\pi^5} \frac{a}{b} \tanh \frac{\pi b}{2a} \right). \quad (11)$$

## 2) Narrow rectangular section:

If  $b/a$  is large, letting  $\tanh(n\pi b/2a) = 1$  and  $\sum_{n=1,3,5,\dots}^{\infty} \frac{1}{n^5} = 1.0046$ , we obtain from exp. (10)

$$J = \frac{a^3b}{3} \left( 1 - 0.630 \frac{a}{b} \right). \quad (12)$$

This gives good approximations for  $b/a > 1.6$ .

If  $b/a = 1 \sim 1.6$  the Föppl's formula below is rather accurate.

$$J = \frac{a^3b^3}{3.6(a^2+b^2)}. \quad (13)$$

In the case where the width is very thin, the second term inside the parentheses of exp. (12) becomes negligible, and we have

$$J = \frac{a^3b}{3}. \quad (14)$$

Note that this is the moment of inertia with respect to the longer side.

## 3) Square section:

For  $a = b$ , exp. (10) yields

$$J = 0.1406a^4. \quad (15)$$

See that eq. (13) gives the value close enough to this.

## 4) Rolled profile sections:

For the sections such as angles, channels and  $I$ 's, divide them up into

rectangular strips and apply exp. (14) to each strip. The summations of the results give the satisfactory values:

$$J = \frac{1}{3} \sum at^3, \quad (16)$$

where  $a$  = the width of the strip,  
 $t$  = the thickness of the strip.

### 5. Relative Stiffnesses

The torsional rigidity  $GJ$  of a member can be expressed by the flexural rigidity  $E\bar{I}$  of any member chosen as reference:

$$GJ = \frac{mE}{2(m+1)} J = \left\{ \frac{m}{2(m+1)} \frac{J}{\bar{I}} \right\} E\bar{I},$$

where  $m$  = Poisson's number,

$\bar{I}$  = the moment of inertia of the section of the reference member.

The transformation gives

$$\frac{GJ}{l} = \left\{ \frac{m}{2(m+1)} \frac{J/l}{\bar{I}/\bar{l}} \right\} \frac{E\bar{I}}{\bar{l}},$$

or

$$GK_t = \left\{ \frac{m}{2(m+1)} \frac{K_t}{\bar{K}_b} \right\} E\bar{K}_b,$$

where  $l, \bar{l}$  = the lengths of the member considered and of reference member respectively.

$K_t = J/l$  = the stiffness factor for the torsion of the member in question.

$\bar{K}_b = \bar{I}/\bar{l}$  = the stiffness factor for the flexure of the reference member.

This will be written in the simple form:

$$GK_t = 4\beta k_t E\bar{K}_b, \quad (17)$$

in which

$$\beta = \frac{G}{4E} = \frac{m}{8(m+1)}, \quad (18)$$

$$k_t = \frac{K_t}{\bar{K}_b}. \quad (19)$$

$k_t$  denotes the stiffness factor for the torsion measured by the reference stiffness factor for flexure, which is called the relative stiffness or stiffness ratio for torsion.

Similarly, measuring the stiffness factor for flexure  $K_b$  of the member considered by  $\bar{K}_b$ , the stiffness ratio for flexure is defined:

$$k_b = \frac{K_b}{\bar{K}_b}, \quad (20)$$

whence we have, corresponding to eq.(17)

$$EK_b = k_b E \bar{K}_b. \quad (21)$$

### 6. Equations for Practical Use

To obtain facilities for practical calculations, the foregoing equations will be simplified by using the stiffness ratios and the symbols below:

$$\left. \begin{aligned} \varphi &= 2E\bar{K}_b\theta, \\ \psi &= -6E\bar{K}_bR. \end{aligned} \right\} \quad (22)$$

Thus we have from the slope-deflection eqs.(3) and (4)

$$M_{ynf} = k_{ynf}(2\varphi_{yn} + \varphi_{yf} + \psi_{ynf}) + C_{ynf}, \quad (23)$$

$$M_{znf} = k_{znf}(2\varphi_{zn} + \varphi_{zf} + \psi_{znf}) + C_{znf}, \quad (24)$$

and from the torsion eq.(9)

$$M_{xnf} = 2\beta k_{xnf}(\varphi_{xn} - \varphi_{xf}) + C_{xnf}. \quad (25)$$

### 7. Members Free to Rotate at Far End

Consider the member is supported at the far end so as to rotate freely against bending but fixed against torsion. For such member the slope-deflection equations are also provided. For example, if the member in Fig.1 is bending-free at the far end f about y axis only, we have, in xz plane, the end-moment expression at the near end n eliminating  $\varphi_{yf}$  from eq.(23) that is using the relation  $M_{yfn}=0$ , thus

$$M_{ynf} = \frac{1}{2}k_{ynf}(3\varphi_{yn} + \psi_{ynf}) + H_{ynf}, \quad (26)$$

where

$$H_{ynf} = C_{ynf} - \frac{1}{2}C_{yfn}. \quad (27)$$

Further, if the far end of the member in question is torsion-free but bending-fixed, we readily find that

$$M_{xnf} = C_{xnf}. \quad (28)$$

This becomes zero when the member has no torque applied, therefore we need not consider such members in such a case.

### 8. End-Shears

If a member of the space frame acted upon by the loads, the reactions normal to its axis are induced in addition to the end-moments. These reactions, often called the end-shears, will be denoted by  $X$  with three subscripts as before. See Fig. 4.

Consider, for example, the member  $nf$  lying parallel to  $x$  axis, and write the equilibrium conditions in  $xy$  plane. Then we readily have the expressions for  $X$ 's:

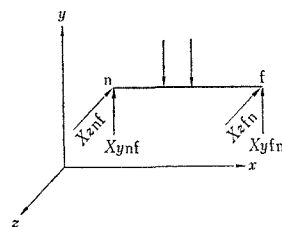


Fig. 4

$$\left. \begin{aligned} X_{ynf} &= -\frac{1}{l_{nf}}(M_{znf} + M_{zfn}) + \bar{X}_{ynf}, \\ X_{yfn} &= -\frac{1}{l_{nf}}(M_{znf} + M_{zfn}) + \bar{X}_{yfn}, \end{aligned} \right\} \quad (29)$$

where  $\bar{X}$ 's denote the normal reactions produced by the loads when the member is assumed to be a simple beam.

$X$ 's and  $\bar{X}$ 's are taken as positive when they have the moments about the far end which agree with the right-hand screw rule. For the other reaction components such as  $X_{znf}$  and  $X_{zfn}$ , the similar expressions will be obtained.

## Chapter II. Elastic Equations

### 9. Joint Equilibrium Equations

The condition of continuity at the joints states that the intersection angles between the member-axes remain unchanged even if the frame undergoes deformation, i.e., all the member-ends rotate by the same angle at the joint under consideration. This rotation angle, common to all member-ends, is



termed the joint rotation angle or briefly the joint rotation. To satisfy the condition of continuity, it is only necessary to put the respective component joint rotations in the places of the component end-slopes in the end-moment equations.

Having this done, we write the equilibrium equations expressing that the component end-moments total to zero at each joint. For example, at joint a, we have

$$\left. \begin{aligned} \text{about } x \text{ axis, } \sum(-M_{xai}) + \bar{M}_{xa} &= 0, \\ \text{about } y \text{ axis, } \sum(-M_{yai}) + \bar{M}_{ya} &= 0, \\ \text{about } z \text{ axis, } \sum(-M_{zai}) + \bar{M}_{za} &= 0, \end{aligned} \right\} \quad (30)$$

where  $i$  = the subscript designating the joint adjacent to a,

$\bar{M}$  = the component external moment existing at a.

Thus, we have equilibrium equations like these, each three at each joint, which coincide in total with the number of unknown  $\varphi$ 's.

#### 10. Horizontal-Shear Equations

A typical space frame is considered to be a system of plane frames standing parallel to  $xy$  or  $yz$  plane connected each other by a group of beams running parallel to  $z$  or  $y$  axis. In Fig. 5 the constituent plane frames are shown by heavy lines, the connecting beams by fine lines; in (a) the plane frames are arranged in row, and in (b) in column.

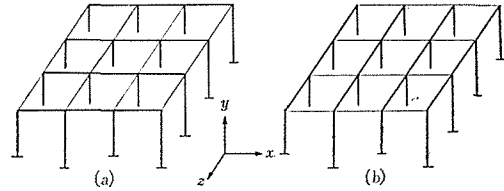


Fig. 5

Now consider a plane frame of any row as a free-body isolated from the whole assembly, and write the equilibrium of forces acting along the section through its column-tops. Considering the forces transmitted from the connecting beams, Fig. 6, we have

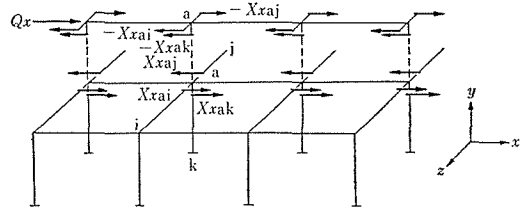


Fig. 6

$$\sum(-X_{xak}) + \sum(-X_{xai} - X_{xaj}) + Q_x = 0, \quad (31)$$

where  $i, j$  = subscripts designating the adjacent joints,

$k = \text{do. the column base,}$

$Q_x = \text{the } x\text{-component of the total horizontal load acting above the section considered.}$

This is written in the simple form:

$$\sum(-X_x) + Q_x = 0. \quad (32)$$

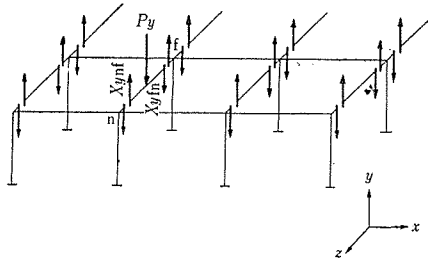
Similarly, observing a plane-frame of any column, we obtain

$$\sum(-X_z) + Q_z = 0. \quad (33)$$

These equations express the equilibrium condition of horizontal forces, and we will call them the horizontal-shear equations; see that the number of them equals the sum of numbers of rows and of columns.

### 11. Vertical-Shear Equations

The connecting beams spanning any two adjacent plane frames must be in equilibrium under the vertical forces. Considering, for example, any beam  $nf$  in row-arrangement shown in Fig. 7, we immediately have



$$X_{ynf} - X_{yfn} - P_y = 0, \quad (34)$$

where  $P = \text{the resultant vertical load acting on the beam.}$

Applying this relation to the beams in order and adding, we obtain

$$\sum(X_{ynf} - X_{yfn}) - \sum P_y = 0. \quad (35)$$

Fig. 7

The similar equations are also deduced in the case of column-arrangement. These will be referred to as the vertical-shear equations, the number of which agrees with the sum of numbers of bays of the plane frames in column-arrangement and of those in row-arrangement.

### 12. Compatibility Equations

When the members of the frame change their lengths due to the axial forces, temperature changes etc., the displacements of joints are resulted, which produce the revolutions of members. The matter is the same when the

supports happen to displace. Since the members can revolve in two directions, the number of unknown  $R$ 's (or  $\psi$ 's) is equal to twice the number of the members. These  $R$ 's must be compatible with the geometry of the frame to be mentioned below.

Take, for example, any space bounded by members either open or closed. Such a space is shown in Fig. 8(a), where the skeleton after deformation is drawn by broken lines assuming the left support to be fixed in position.

Let the initial and final dimensions of the skeletons be as follows:

	Initial	Final
Length of member	$s$	$s + \Delta s$
Inclination of member	$\alpha$	$\alpha_z - R_z^{(*)}$
Span	$l$	$l + \Delta l$
Relative height of the ends	$h$	$h + \Delta h$

(\*) Cf Fig. 8(d).

Then we have, by geometry; the relations after deformation:

$$\sum (s + \Delta s) \cos(\alpha_z - R_z) = l + \Delta l,$$

$$\sum (s + \Delta s) \sin(\alpha_z - R_z) = h + \Delta h.$$

Expand these equations considering the geometry before deformation,

$\sum s \cos \alpha_z = l$  and  $\sum s \sin \alpha_z = h$ ; let  $\sin R_z = R_z$  and  $\cos R_z = 1$ , and omit the small terms of higher order in the resulting expressions since  $\Delta s$  and  $R_z$  are very small. Then we finally have

$$\left. \begin{aligned} \sum \Delta s \cos \alpha_z + \sum R_z s \sin \alpha_z &= \Delta l, \\ \sum \Delta s \sin \alpha_z - \sum R_z s \cos \alpha_z &= \Delta h. \end{aligned} \right\} \quad (36)$$

These are the compatibility equations required. If we consider such spaces as in Fig. 8(b) and (c), change the subscripts of  $\alpha$  and  $R$  to  $x$  and  $y$  respectively.

On some particular cases it will be mentioned that:

- 1) When the supports do not displace, let  $\Delta l = \Delta h = 0$ .
- 2) For a closed space, also put  $\Delta l = \Delta h = 0$ .
- 3) For the case where the lengths of members remain unchanged and the

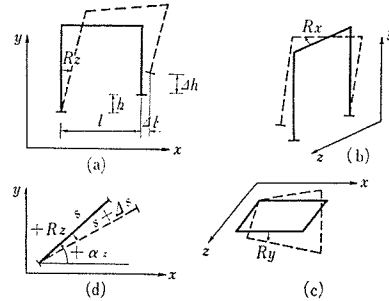


Fig. 8

supports are fixed in positions, eqs. (36) reduce to the simple form:

$$\left. \begin{aligned} \sum R s \sin \alpha &= 0, \\ \sum R s \cos \alpha &= 0. \end{aligned} \right\} \quad (37)$$

The subscripts are here omitted.

### 13. Establishment of Solution

In order to carry out the analysis of the rigid frames in space, the equilibrium equations—joint equilibrium equations, horizontal- and vertical-shear equations—should be utilized together with the compatibility equations. To make sure the possibility of solution, we will here investigate the numbers of unknowns sought and of condition equations available. Take, for simplicity, a space frame of single story, and let  $m$  and  $n$  be the numbers of its constituent plane frames in row- and column-arrangements respectively.

Then, the joints, the numbers and the spaces are respectively enumerated as shown in Table 1.

Table 1

Item		No.
Joint		$mn$
Member	Column	$mn$
	Beam, parallel to $x$ axis	$m(n-1)$
	Do., " $z$ "	$n(m-1)$
		$3mn - (m+n)$
Space	Parallel to $yz$ plane	$m'n-1$
	" $xz$ "	$(m-1)n-1$
	" $xy$ "	$n'm-1$
		$3mn - 2(m+n) + 1$

With this, we can tell the numbers of condition equations as follows:

Joint equilibrium equations... $3mn$  (three times of no. of joints).

Horizontal-shear equations... $m+n$  (sum of nos. of rows and columns).

Vertical-shear equations... $(m-1) + (n-1) = m+n-2$  (sum of nos. of bays of constituent plane frames in both arrangements).

Compatibility equations... $6mn - 4(m+n) + 2$  (twice the nos. of spaces)

Totals... $9mn - 2(m+n)$ .

On the other hand, the number of unknown  $\varphi$ 's is  $3mn$ , and of  $\psi$ 's  $6mn - 2(m+n)$ , which evidently agrees, in all, with that of condition equations shown

above. Therefore, we can conclude that the solution is always possible.

The similar discussion will hold for the frames of multi-story.

If we assume that the supports are immovable and the member-lengths are immutable, which is the case usually considered, the compatibility equations become needless, because we can tell, by inspection, as follows:

1) The beams can revolve about  $y$  axis only. All the beams spanning adjacent two plane frames have the same  $\phi$ . Hence, the number of  $\phi$ 's of them becomes  $(m - 1) + (n - 1) = m + n - 2$ .

2) All the columns in any constituent plane frame have the same  $R$ , if their heights are all equal. Hence, the number of  $\phi$ 's of columns is  $m + n$ .

3) Thus, the number of  $\phi$ 's totals to  $2(m + n - 1)$ , which just comes to the number of shear equations.

4) The matter is the same, if there is irregularity in the length of columns and beams.

Furthermore, if the joints do not displace due to the symmetry or to the lateral supports,  $\phi$ 's become zeros, hence the solution is carried out by the use of the joint equilibrium equations only.

#### 14. Comparison with Method of Redundancies

In order to obtain a statically determinate rigid connection in space by assembling  $k$  bars, we have to apply six constraints at each knot. Since there are  $k - 1$  knots, the required constraints must be  $6(k - 1)$  in all. The frame thus fabricated is then attached to the foundation to complete a statically determinate structure. For this purpose, we are again in need of six external constraints. Thus, we see that a statically determinate rigid frame should have the constraints of  $6(k - 1) + 6 = 6k$  in total.

If the frame under consideration is statically indeterminate, it has more constraints than  $6k$ , and its degree of redundancy  $N$  is found to be

$$N = r + j - 6k,$$

where  $r$  = total external constraints,

$j$  = total internal constraints. At a rigid joint,  $j$  denotes the number of meeting members less than one multiplied by six.

The methods of redundancies, the classical methods, take the end-forces for unknowns, which are the constraints above mentioned. Therefore,  $N$  denotes the number of redundant forces to be found.

Let, for example, a single storied frame of  $m$  rows and  $n$  columns be



$$\begin{aligned}
&\text{About } x \text{ axis, } M_{xEF} = k_{xEF}(\varphi_{xF} + \phi_{xEF}) = k_{xEF}(\varphi_{xF} + (H/h)\phi_{xAB}), \\
&\text{Do. , } M_{xFE} = k_{xEF}(2\varphi_{xF} + \phi_{xEF}) = k_{xEF}(2\varphi_{xF} + (H/h)\phi_{xAB}), \\
&\text{About } z \text{ axis, } M_{zEF} = k_{zEF}\varphi_{zF}, \\
&\text{Do. , } M_{zFE} = 2k_{zEF}\varphi_{zF}.
\end{aligned}$$

For member FG:

$$\text{About } z \text{ axis, } M_{zFG} = k_{zFG}\varphi_{zF} + C_{zFG}.$$

## 2) Elastic Equations

Condition equations to be satisfied by these end-moments are:

### i) Joint Equilibrium Equations

At joint B:

$$\begin{aligned}
&\text{About } x \text{ axis, } M_{xBA} + M_{xBF} = 0, \\
&\text{About } z \text{ axis, } M_{zBA} + M_{zBC} + M_{zBF} = 0.
\end{aligned}$$

At joint F:

$$\begin{aligned}
&\text{About } x \text{ axis, } M_{xFB} + M_{xFE} = 0, \\
&\text{About } z \text{ axis, } M_{zFB} + M_{zFE} + M_{zFG} = 0.
\end{aligned}$$

### ii) Horizontal Shear Equations

$$X_{zBA} + X_{zFE} = 0.$$

Substituting the above end-moments (a) into these equilibrium equations (b), we get the following simultaneous equations.

$$\begin{aligned}
&2(k_{xAB} + k_{xBF})\varphi_{xB} + k_{xBF}\varphi_{xF} + k_{xAB}\phi_{xAB} + C_{xBF} = 0, \\
&k_{xBF}\varphi_{xB} + 2(k_{xBF} + k_{xEF})\varphi_{xF} + (H/h)k_{xEF}\phi_{xAB} + C_{xFB} = 0, \\
&(2k_{xAB} + k_{zBC} + 2\beta k_{zBF})\varphi_{zB} - 2\beta k_{zBF}\varphi_{zF} + C_{zBC} = 0, \\
&(2k_{zEF} + k_{zFG} + 2\beta k_{zBF})\varphi_{zF} - 2\beta k_{zBF}\varphi_{zB} + C_{zFG} = 0, \\
&3hk_{xAB}\varphi_{xB} + 3Hk_{xEF}\varphi_{xF} + (2hk_{xAB} + 2(H^2/h)k_{xEF})\phi_{xAB} = 0.
\end{aligned}$$

Solving these, we have:

Joint rotations at B,

$$\begin{aligned}
\varphi_{xB} &= (-1/\nu)\{[4k_{xBF} + k_{xEF}](hk_{xAB} + (H^2/h)k_{xEF}) - 3(H^2/h)k_{xEF}^2\}C_{xBF} \\
&\quad - [2k_{xBF}(hk_{xAB} + (H^2/h)k_{xEF}) - 3Hk_{xAB}k_{xEF}]C_{xFB}, \\
\varphi_{zB} &= (-1/\mu)\{(2k_{zEF} + k_{zFG} + 2\beta k_{zBF})C_{zBC} + 2\beta k_{zBF}C_{zFG}\}.
\end{aligned}$$

Joint rotations at F,

$$\begin{aligned}
\varphi_{xF} &= (1/\nu)\{[2k_{xBF}(hk_{xAB} + (H^2/h)k_{xEF}) - 3Hk_{xAB}k_{xEF}]C_{xBF} \\
&\quad - [4k_{xAB} + k_{xBF}](hk_{xAB} + (H^2/h)k_{xEF}) - 3hk_{xAB}^2\}C_{xFB},
\end{aligned}$$

$$\varphi_{zF} = (-1/\mu)\{(2k_{zAB} + k_{zBC} + 2\beta k_{zBF})C_{zFG} + 2\beta k_{zBF}C_{zBC}\}.$$

Revolution of member AB,

$$\begin{aligned} \psi_{xAB} = & (-1/\nu)\{[3Hk_{xBF}k_{xEF} - 6hk_{xAB}(k_{xBF} + k_{xEF})]C_{xBF} \\ & - [6Hk_{xEF}(k_{xAB} + k_{xBF}) + 3hk_{xAB}k_{xBF}]C_{xBF}\}, \end{aligned}$$

where,

$$\begin{aligned} \mu = & (2k_{zAB} + k_{zBC} + 2\beta k_{zBF})(2k_{xEF} + k_{zFG} + 2\beta k_{zBF}) - 4\beta^2 k_{zBF}^2, \\ \nu = & 2(k_{xAB} + k_{xBF})\{4(k_{xBF} + k_{xEF})(hk_{xAB} + (H^2/h)k_{xEF}) - 3(H^2/h)k_{xEF}^2\} \\ & - k_{xBF}\{2k_{xBF}(hk_{xAB} + (H^2/h)k_{xEF}) - 3Hk_{xAB}k_{xEF}\} \\ & + 3hk_{xAB}\{(H/h)k_{xBF}k_{xEF} - 2k_{xAB}(k_{xBF} + k_{xEF})\}. \end{aligned}$$

(c)

Finally, introducing these  $\varphi$  - and  $\psi$  - values into the expressions of end moments (a), we arrive at the solutions.

Consider now the case in which  $k_{xBF} = k_{zBF} = 0$  and  $C_{xBF} = C_{xFB} = 0$ , then we have from (c)

$$\begin{aligned} \varphi_{xB} = \varphi_{xF} = \psi_{xAB} = \psi_{xEF} = 0, \\ \varphi_{zB} = -C_{zBC}/(2k_{zAB} + k_{zBC}), \quad \varphi_{zF} = -C_{zFG}/(2k_{xEF} + k_{zFG}). \end{aligned}$$

These coincide with the solutions by the conventional two-dimensional method, that is, the solutions for the constituent plane frames ABCD or EFGH.

In the followings, we will consider in detail about  $M_{zBA}$ ,  $M_{zBC}$  and  $M_{zBF}$ .

### 3) Relations between $k_{xEF}$ and $M$ 's

Assuming that  $p_1 = p_2$ ,  $k_{zAB} = k_{zBC} = k_{zBF} = k_{zFG} = 1$  and that  $\beta = 0.11$ , putting  $m = 6$  in eq.(18), we obtain  $\mu = 6.44 k_{xEF} + 3.88$ ,  $C_{zBC} = C_{zFG}$  and

$$\begin{aligned} \varphi_{zB} = & -(2k_{xEF} + 1.44)C_{zBC}/(6.44k_{xEF} + 3.88), \\ \varphi_{zF} = & -3.44C_{zBC}/(6.44k_{xEF} + 3.88). \end{aligned}$$

Hence we get, from eqs.(a),

$$\begin{aligned} M_{zBA} = & -(2k_{xEF} + 1.44)C_{zBC}/(3.22k_{xEF} + 1.94), \\ M_{zBC} = & (2.22k_{xEF} + 1.22)C_{zBC}/(3.22k_{xEF} + 1.94), \\ M_{zBF} = & -0.22(k_{xEF} - 1)C_{zBC}/(3.22k_{xEF} + 1.94). \end{aligned}$$

Thus we have the end-moments expressed by  $k_{xEF}$ , the relations between them are shown in Table 2.



**Table 2**  $k_{zEF} \sim M$ 

$M \backslash k_{zEF}$	0	0.2	0.4	0.6	0.8	1.0	1.2	1.4	1.6	1.8	2.0
$M_{zBA}$	-0.74	-0.71	-0.69	-0.68	-0.67	-0.67	-0.66	-0.66	-0.65	-0.65	-0.65
$M_{zBC}$	0.63	0.64	0.65	0.66	0.66	0.67	0.67	0.67	0.67	0.67	0.67
$M_{zBF}$	0.11	0.07	0.04	0.02	0.01	0	-0.01	-0.01	-0.02	-0.02	-0.02

Multiplier:  $C_{zBC} = -p_l l^2/12$ 

See that as  $k_{zEF}$  increases,  $M_{zBC}$  increases and  $M_{zBA}$  and  $M_{zBF}$  decrease, but for  $k_{zEF} > 1$  they remain almost constant. Again, when  $k_{zEF}$  becomes unity, i.e., the frame becomes also symmetrical about  $xz$  plane, the twisting moments vanish and the frame ABCD may be isolated and treated two-dimensionally.

#### 4) Relations between $C_{zFG}/C_{zBC}$ and $M$ 's

For convenience sake, letting all stiffness ratios be unities and  $\beta$  be 0.11, we have

$$\varphi_{zB} = -(0.31C_{zBC} + 0.02C_{zFG}),$$

$$\varphi_{zF} = -(0.31C_{zFG} + 0.02C_{zBC}),$$

and

$$M_{zBA} = -C_{zBC}(0.62 + 0.04C_{zFG}/C_{zBC}),$$

$$M_{zBC} = C_{zBC}(0.69 - 0.02C_{zFG}/C_{zBC}),$$

$$M_{zBF} = -0.06C_{zBC}(1 - C_{zFG}/C_{zBC}).$$

The relations between  $C_{zFG}/C_{zBC}$  and  $M$ 's are shown in Table 3.

**Table 3**  $C_{zFG}/C_{zBC} \sim M$ 

$M \backslash C_{zFG}/C_{zBC}$	0	0.2	0.4	0.6	0.8	1.0	1.2	1.4	1.6	1.8	2.0
$M_{zBA}$	-0.62	-0.63	-0.64	-0.64	-0.65	-0.67	-0.67	-0.68	-0.68	-0.69	-0.70
$M_{zBC}$	0.69	0.69	0.68	0.68	0.67	0.67	0.67	0.66	0.66	0.65	0.65
$M_{zBF}$	-0.07	-0.06	-0.04	-0.02	-0.01	0	0.01	0.02	0.04	0.05	0.06

Multiplier:  $C_{zBC} = -p_l l^2/12$ 

From this table it is perceived that  $M_{zBA}$  and  $M_{zBC}$  are not so different from the two-dimensional solution,  $0.67 C_{zBC}$ , and that the magnitudes of twisting moments in the range considered are very small compared with those of bending moments. Thus, it may be concluded that the variation of  $C_{zFG}/C_{zBC}$  does not effect so much upon the end-moments analysed either three-dimen-

sionally or two-dimensionally.

5) Relations between  $k_{zBF}$  and  $M$ 's

To show the effects of the torsional rigidity of a member, consider the frame in Fig. 10 which is the special case in Fig. 9. For this frame we can put  $k_{zEF} = k_{zFG} = \infty$  and  $C_{zFG} = 0$  in the foregoing solution.

To investigate the relations between  $k_{zBF}$  and the end-moments, let, for example,  $k_{zAB} = k_{zBC} = 1$ . Then, we have

$$\varphi_{zB} = -C_{zBC}/(3 + 2\beta k_{zBF})$$

and

$$M_{zBA} = -2C_{zBC}/(3 + 0.22k_{zBF}),$$

$$M_{zBC} = 2(1 + 0.11k_{zBF})C_{zBC}/(3 + 0.22k_{zBF}),$$

$$M_{zBF} = -0.22k_{zBF}C_{zBC}/(3 + 0.22k_{zBF}).$$

Table 4 shows the relations thus obtained which are plotted in Fig. 11.

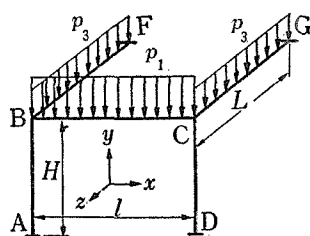


Fig. 10

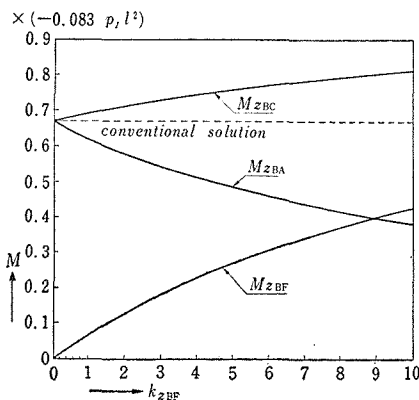


Fig. 11

Table 4  $k_{zBF} \sim M$

$M \backslash k_{zBF}$	0	0.2	0.4	0.6	0.8	1.0	2.0	4.0	6.0	8.0	10.0	$\infty$
$M_{zBA}$	-0.67	-0.66	-0.65	-0.64	-0.63	-0.62	-0.58	-0.52	-0.46	-0.42	-0.38	0
$M_{zBC}$	0.67	0.67	0.68	0.68	0.69	0.69	0.71	0.74	0.77	0.79	0.81	1
$M_{zBF}$	0	-0.014	-0.028	-0.042	-0.055	-0.068	-0.128	-0.227	-0.306	-0.370	-0.423	—

Multiplier:  $C_{zBC} = -p_1 l^2 / 12$



In the followings, we allot 0.107 to  $\beta$  assuming  $m = 6$  in eq.(18).

For member AD :

$$\begin{aligned} M_{xAD} &= \varphi_{xD} + \phi_{xAD}, & M_{xDA} &= 2\varphi_{xD} + \phi_{xAD}. \\ M_{yAD} &= -0.21\varphi_{yD}, & M_{yDA} &= 0.21\varphi_{yD}. \\ M_{zAD} &= \varphi_{zD} + \phi_{zAD}, & M_{zDA} &= 2\varphi_{zD} + \phi_{zAD}. \end{aligned}$$

For member BE :

$$\begin{aligned} M_{xBE} &= 1.2\varphi_{xE} + 1.2\phi_{xBE}, & M_{xEB} &= 2.4\varphi_{xE} + 1.2\phi_{xBE}. \\ M_{yBE} &= -0.21\varphi_{yE}, & M_{yEB} &= 0.21\varphi_{yE}. \\ M_{zBE} &= 1.2\varphi_{zE} + 0.93\phi_{zAD}, & M_{zEB} &= 2.4\varphi_{zE} + 0.93\phi_{zAD}. \end{aligned}$$

For member CF :

$$\begin{aligned} M_{xCF} &= 0.8\varphi_{xF} + 0.8\phi_{xCF}, & M_{xFC} &= 1.6\varphi_{xF} + 0.8\phi_{xCF}. \\ M_{yCF} &= -0.21\varphi_{yF}, & M_{yFC} &= 0.21\varphi_{yF}. \\ M_{zCF} &= 0.8\varphi_{zF} + 1.12\phi_{zAD}, & M_{zFC} &= 1.6\varphi_{zF} + 1.12\phi_{zAD}. \end{aligned}$$

For member DE :

$$\begin{aligned} M_{xDE} &= 0.30(\varphi_{xD} - \varphi_{xE}), & M_{xED} &= 0.30(\varphi_{xE} - \varphi_{xD}). \\ M_{yDE} &= 1.2(2\varphi_{yD} + \varphi_{yE} + 1.05\phi_{xAD} - 1.35\phi_{xBE}), \\ M_{yED} &= 1.2(2\varphi_{yE} + \varphi_{yD} + 1.05\phi_{xAD} - 1.35\phi_{xBE}). \\ M_{zDE} &= 2\varphi_{zD} + \varphi_{zE}, & M_{zED} &= 2\varphi_{zE} + \varphi_{zD}. \end{aligned}$$

For member EF :

$$\begin{aligned} M_{xEF} &= 0.3(\varphi_{xE} - \varphi_{xF}), & M_{xFE} &= 0.3(\varphi_{xF} - \varphi_{xE}). \\ M_{yEF} &= 1.2(2\varphi_{yE} + \varphi_{yF} + 1.35\phi_{xBE} - 0.75\phi_{xCF}), \\ M_{yFE} &= 1.2(2\varphi_{yF} + \varphi_{yE} + 1.35\phi_{xBE} - 0.75\phi_{xCF}). \\ M_{zEF} &= 2\varphi_{zE} + \varphi_{zF}, & M_{zFE} &= 2\varphi_{zF} + \varphi_{zE}. \end{aligned}$$

For member DJ :

$$\begin{aligned} M_{xDJ} &= 1.5(2\varphi_{xD} + \varphi_{xJ}), & M_{xJD} &= 1.5(2\varphi_{xJ} + \varphi_{xD}). \\ M_{yDJ} &= (2\varphi_{yD} + \varphi_{yJ} + 0.724\phi_{zGJ} - 0.84\phi_{zAD}), \\ M_{yJD} &= (2\varphi_{yJ} + \varphi_{yD} + 0.724\phi_{zGJ} - 0.84\phi_{zAD}). \\ M_{zDJ} &= 0.3(\varphi_{zD} - \varphi_{zJ}), & M_{zJD} &= 0.3(\varphi_{zJ} - \varphi_{zD}). \end{aligned}$$

For member EK :

$$\begin{aligned} M_{xEK} &= 1.5(2\varphi_{xE} + \varphi_{xK}), & M_{xKE} &= 1.5(2\varphi_{xK} + \varphi_{xE}). \\ M_{yEK} &= (2\varphi_{yE} + \varphi_{yK} + 0.72\phi_{zGJ} - 0.84\phi_{zAD}), \\ M_{yKE} &= (2\varphi_{yK} + \varphi_{yE} + 0.72\phi_{zGJ} - 0.84\phi_{zAD}). \\ M_{zEK} &= 0.3(\varphi_{zE} - \varphi_{zK}), & M_{zKE} &= 0.3(\varphi_{zK} - \varphi_{zE}). \end{aligned}$$

For member FL :

$$\begin{aligned} M_{xFL} &= 1.5(2\varphi_{xF} + \varphi_{xL}), & M_{xLF} &= 1.5(2\varphi_{xL} + \varphi_{xF}). \\ M_{yFL} &= (2\varphi_{yF} + \varphi_{yL} + 0.72\phi_{zGJ} - 0.84\phi_{zAD}), \\ M_{yLF} &= (2\varphi_{yL} + \varphi_{yF} + 0.72\phi_{zGJ} - 0.84\phi_{zAD}). \\ M_{zFL} &= 0.3(\varphi_{zF} - \varphi_{zL}), & M_{zLF} &= 0.3(\varphi_{zL} - \varphi_{zF}). \end{aligned}$$

For member JK :

$$\begin{aligned} M_{xJK} &= 0.3(\varphi_{xJ} - \varphi_{xK}), & M_{xKJ} &= 0.3(\varphi_{xK} - \varphi_{xJ}). \\ M_{yJK} &= 1.2(2\varphi_{yJ} + \varphi_{yK} + 1.05\phi_{xAD} - 1.35\phi_{xBE}), \\ M_{yKJ} &= 1.2(2\varphi_{yK} + \varphi_{yJ} + 1.05\phi_{xAD} - 1.35\phi_{xBE}). \\ M_{zJK} &= 0.8(2\varphi_{zJ} + \varphi_{zK}), & M_{zKJ} &= 0.8(2\varphi_{zK} + \varphi_{zJ}). \end{aligned}$$

For member KL :

$$\begin{aligned} M_{xKL} &= 0.3(\varphi_{xK} - \varphi_{xL}), & M_{xLK} &= 0.3(\varphi_{xL} - \varphi_{xK}). \\ M_{yKL} &= 1.2(2\varphi_{yK} + \varphi_{yL} + 1.35\phi_{xBE} - 0.75\phi_{xCF}), \\ M_{yLK} &= 1.2(2\varphi_{yL} + \varphi_{yK} + 1.35\phi_{xBE} - 0.75\phi_{xCF}). \\ M_{zKL} &= 0.8(2\varphi_{zK} + \varphi_{zL}), & M_{zLK} &= 0.8(2\varphi_{zL} + \varphi_{zK}). \end{aligned}$$

For member GJ :

$$\begin{aligned} M_{xGJ} &= \varphi_{xJ} + 1.17\phi_{xAD}, & M_{xJG} &= 2\varphi_{xJ} + 1.17\phi_{xAD}. \\ M_{yGJ} &= -0.21\varphi_{yJ}, & M_{yJG} &= 0.21\varphi_{yJ}. \\ M_{zGJ} &= \varphi_{zJ} + \phi_{zGJ}, & M_{zJG} &= 2\varphi_{zJ} + \phi_{zGJ}. \end{aligned}$$

For member HK :

$$M_{xHK} = 1.2\varphi_{xK} + 1.54\phi_{xBE}, \quad M_{xKH} = 2.4\varphi_{xK} + 1.54\phi_{xBE}.$$

$$M_{yHK} = -0.26\varphi_{yK},$$

$$M_{yKH} = 0.26\varphi_{yK}.$$

$$M_{zHK} = \varphi_{zK} + 0.86\psi_{zGJ},$$

$$M_{zKH} = 2\varphi_{zK} + 0.86\psi_{zGJ}.$$

For member IL :

$$M_{xIL} = 0.8\varphi_{xL} + 0.8\psi_{xCF},$$

$$M_{xLI} = 1.6\varphi_{xF} + 0.8\psi_{xCF}.$$

$$M_{yIL} = -0.17\varphi_{yL},$$

$$M_{yLI} = 0.17\varphi_{yL}.$$

$$M_{zIL} = \varphi_{zL} + 1.2\psi_{zGJ},$$

$$M_{zLI} = 2\varphi_{zL} + 1.2\psi_{zGJ}.$$

## 2) Expressions of End-Shears

$$X_{zDA} = -(0.71\varphi_{xD} + 0.48\psi_{xAD}),$$

$$X_{zDE} = -(0.9\varphi_{yD} + \varphi_{yE} + 0.63\psi_{xAD} - 0.81\psi_{xBE}),$$

$$X_{zJG} = -(0.83\varphi_{xJ} + 0.65\psi_{xAD}),$$

$$X_{zDE} = X_{zED},$$

$$X_{zEB} = -(0.67\varphi_{xE} + 0.44\psi_{xBE}),$$

$$X_{zKJ} = -(0.9\varphi_{yJ} + 0.9\varphi_{yK} + 0.63\psi_{xAD} - 0.81\psi_{xBE}),$$

$$X_{zJK} = X_{zKJ},$$

$$X_{zKH} = -(0.86\varphi_{xK} + 0.73\psi_{xBE}),$$

$$X_{zFE} = -(0.9\varphi_{yE} + 0.9\varphi_{yF} + 0.81\psi_{xBE} - 0.45\psi_{xCF}),$$

$$X_{zEF} = X_{zFE},$$

$$X_{zFC} = -(0.8\varphi_{xF} + 0.53\psi_{xCF}),$$

$$X_{zLK} = -(0.9\varphi_{yK} + 0.9\varphi_{yL} + 0.81\psi_{xBE} - 0.45\psi_{xCF}),$$

$$X_{zKL} = X_{zLK},$$

$$X_{zLI} = -(0.8\varphi_{xL} + 0.53\psi_{xCF}),$$

$$X_{xDA} = -(0.71\varphi_{xD} + 0.48\psi_{xAD}),$$

$$X_{xEB} = -(0.67\varphi_{zE} + 0.34\psi_{zAD}),$$

$$X_{xFC} = -(0.8\varphi_{zF} + 0.75\psi_{zAD}),$$

$$X_{xDJ} = -(0.6\varphi_{yD} + 0.6\varphi_{yJ} + 0.29\varphi_{zGJ} - 0.34\psi_{zAD}),$$

$$X_{xDJ} = X_{xDJ},$$

$$X_{xEK} = -(0.6\varphi_{yE} + 0.6\varphi_{yK} + 0.29\varphi_{zGJ} - 0.34\psi_{zAD}),$$

$$X_{xEK} = X_{xEK},$$

$$X_{xFL} = -(0.6\varphi_{yF} + 0.6\varphi_{yL} + 0.29\varphi_{zGJ} - 0.34\psi_{zAD}),$$

$$X_{xFL} = X_{xFL},$$

$$X_{xJG} = -(0.83\varphi_{zJ} + 0.56\psi_{zGJ}),$$

$$X_{xKH} = -(0.71\varphi_{zK} + 0.41\psi_{zGJ}),$$

$$X_{xLI} = -(\varphi_{zL} + 0.80\psi_{zGJ}).$$

## 3) Elastic Equations

## i) Joint Equilibrium Equations

At joint D :

$$M_{xDA} + M_{xDE} + M_{xDJ} = 0. \quad M_{yDA} + M_{yDE} + M_{yDJ} = 0.$$

$$M_{zDA} + M_{zDE} + M_{zDJ} = 0.$$

At joint E :

$$M_{xED} + M_{xEB} + M_{xEF} + M_{xEK} = 0. \quad M_{yED} + M_{yEB} + M_{yEF} + M_{yEK} = 0.$$

$$M_{zED} + M_{zEB} + M_{zEF} + M_{zEK} = 0.$$

At joint F :

$$M_{xFE} + M_{xFC} + M_{xFI} = 0. \quad M_{yFE} + M_{yFC} + M_{yFI} = 0.$$

$$M_{zFE} + M_{zFC} + M_{zFI} = 0.$$

At joint J :

$$M_{xJG} + M_{xJK} + M_{xJD} = 0. \quad M_{yJG} + M_{yJK} + M_{yJD} = 0.$$

$$M_{zJG} + M_{zJK} + M_{zJD} = 0.$$

At joint K :

$$M_{xKJ} + M_{xKH} + M_{xKL} + M_{xKE} = 0. \quad M_{yKJ} + M_{yKH} + M_{yKL} + M_{yKE} = 0.$$

$$M_{zKJ} + M_{zKH} + M_{zKL} + M_{zKE} = 0.$$

At joint L :

$$M_{xLK} + M_{xLI} + M_{xLF} = 0. \quad M_{yLK} + M_{yLI} + M_{yLF} = 0.$$

$$M_{zLK} + M_{zLI} + M_{zLF} = 0.$$

These conditions give eqs.(1)~(18) in Table 7.

## ii) Shear Equations

For the 1st column-frame :  $X_{xDA} + X_{xDE} + X_{zJG} + X_{zJK} - 3 = 0,$ 

$$\text{“} \quad 2\text{nd} \quad \text{“} \quad : \quad X_{xED} - X_{xEB} - X_{xEF} + X_{xKJ} - X_{xKH} - X_{xKL} + 5 = 0,$$

$$\text{“} \quad 3\text{rd} \quad \text{“} \quad : \quad X_{zFE} - X_{zFC} + X_{zLK} - X_{zLI} + 4 = 0,$$

For the 1st row-frame :  $X_{xDA} + X_{xEB} + X_{xFC} - X_{xDJ} - X_{xEK} - X_{xFL} = 0,$ 

$$\text{“} \quad 2\text{nd} \quad \text{“} \quad : \quad X_{xJG} + X_{xKH} + X_{xLI} + X_{xJD} + X_{xKE} + X_{xLF} = 0.$$

From these conditions we get eqs.(19)~(23) in Table 7.

Table 7 Elastic equations

Eq.	Left-hand side											
	$\varphi_{xD}$	$\varphi_{xE}$	$\varphi_{xF}$	$\varphi_{xJ}$	$\varphi_{xK}$	$\varphi_{xL}$	$\varphi_{yD}$	$\varphi_{yE}$	$\varphi_{yF}$	$\varphi_{yJ}$	$\varphi_{yK}$	$\varphi_{yL}$
(1)	5.30	-0.30		1.50								
(2)	-0.30	6.00	-0.30		1.50							
(3)		-0.03	4.90			1.50						
(4)	1.50			5.30	-0.30							
(5)		1.50		-0.30	6.00	-0.30						
(6)			1.50		-0.30	4.90						
(7)							4.61	1.20		1.00		
(8)							1.20	7.01	1.20		1.00	
(9)								1.20	4.61			1.00
(10)							1.00			4.61	1.20	
(11)								1.00		1.20	7.06	1.20
(12)									1.00		1.20	4.57
(13)												
(14)												
(15)												
(16)												
(17)												
(18)												
(19)	0.71			0.83			0.90	0.90		0.90	0.90	
(20)		0.67			0.86		-0.90		0.90	-0.90		0.90
(21)			0.80			0.80		-0.90	-0.90		-0.90	-0.90
(22)							-0.60	-0.60	-0.60	-0.60	-0.60	-0.60
(23)							0.60	0.60	0.60	0.60	0.60	0.60

Thus, we finally have twenty-three simultaneous equations in total, which give the solutions as follows:

$$\begin{aligned}
 &\text{Joint rotations at D, } \varphi_{xD} = +0.615, & \varphi_{yD} = -0.175, & \varphi_{zD} = -0.008, 5, \\
 &\text{Do. at E, } \varphi_{xE} = +0.627, & \varphi_{yE} = +0.040, & \varphi_{zE} = -0.002, 9, \\
 &\text{Do. at F, } \varphi_{xF} = +0.733, & \varphi_{yF} = +0.239, & \varphi_{zF} = -0.010, 6, \\
 &\text{Do. at J, } \varphi_{xJ} = +0.833, & \varphi_{yJ} = -0.175, & \varphi_{zJ} = +0.008, 9, \\
 &\text{Do. at K, } \varphi_{xK} = +0.944, & \varphi_{yK} = +0.039, & \varphi_{zK} = +0.003, 2, \\
 &\text{Do. at L, } \varphi_{xL} = +0.761, & \varphi_{yL} = +0.243, & \varphi_{zL} = +0.010, 8,
 \end{aligned}$$



Table 7—continued—

$\varphi_{zD}$	$\varphi_{zE}$	$\varphi_{zF}$	$\varphi_{zJ}$	$\varphi_{zK}$	$\varphi_{zL}$	$\phi_{xAD}$	$\phi_{xBE}$	$\phi_{xCF}$	$\phi_{zAD}$	$\phi_{zGJ}$	Right-hand side
						1.00	1.20	0.80			
						1.17	1.54				
						1.26	-1.62	0.80	-0.84	0.72	
						1.26		-0.90	-0.84	0.72	
						1.62	-0.90	-0.84	-0.84	0.72	
						1.26	-1.62		-0.84	0.72	
4.30	1.00		-0.30			1.26		-0.90	-0.84	0.72	
1.00	6.70	1.00		-0.30		1.62	-0.90	-0.84	1.00		
1.00	3.90				-0.30				0.93		
									1.12		
-0.30			3.90	0.80						1.00	
	-0.30		0.80	5.50	0.80					0.86	
		-0.30		0.80	3.90					1.20	
						2.39	-1.62				-3.00
						-1.26	4.41	-0.90			5.00
0.71	0.67	0.80					-1.62	1.96			4.00
			0.83	0.71	1.00				2.58	-0.86	
									-1.01	2.63	

and revolutions

$$\phi_{xAD} = -4.32, \quad \phi_{xBE} = -3.98, \quad \phi_{xCF} = -5.68,$$

$$\phi_{zAD} = +0.042, 2, \quad \phi_{zGJ} = -0.039, 7.$$

The end-moments are found substituting these  $\varphi$ 's and  $\phi$ 's into the expressions in 1). The results are shown in Table 8 and in Fig. 13~16.

The horizontal and vertical reactions at supports are determined from the end-moments by statics.

Table 8 Values of end-moments (t-m)

Member	A D		B E		C F		D E		E F
End	A	D	B	E	C	F	D	E	E
$M_x$	-3.705	-3.090	-4.024	-3.271	-3.958	-3.371	-0.004	+0.004	-0.032
$M_y$	+0.037	-0.037	-0.008	+0.008	-0.050	+0.050	+0.632	+0.890	-0.953
$M_z$	+0.034	+0.025	+0.036	+0.032	+0.039	+0.030	-0.020	-0.014	-0.016

Member	E F	D J		E K		F L		J K	
End	F	D	J	E	K	F	L	J	K
$M_x$	+0.032	+3.095	+3.422	+3.297	+3.773	+3.341	+3.383	-0.033	+0.033
$M_y$	-0.714	-0.589	-0.588	+0.055	+0.054	+0.657	+0.660	+0.632	+0.888
$M_z$	-0.024	-0.005	+0.005	-0.002	+0.002	-0.006	+0.006	+0.017	+0.012

Member	K L		G J		H K		I L	
End	K	L	G	J	H	K	I	L
$M_x$	+0.055	-0.055	-4.221	-3.388	-4.996	-3.864	-3.935	-3.326
$M_y$	-0.951	-0.707	+0.037	-0.037	-0.010	+0.010	-0.041	+0.041
$M_z$	+0.014	+0.020	-0.031	-0.022	-0.031	-0.028	-0.037	-0.026

## 4) End-Moments Analysed Two-Dimensionally

The usual two-dimensional analysis treats the constituent plane frames ADJG, BEKH and CFLI separately. The solutions are easily carried out, and we have the results shown by broken lines in Fig. 13, which compares with the exact values, shown by full lines. Note that the two-dimensional analysis does not produce the values of  $M_y$ 's and  $M_z$ 's.

In Table 9 the end-moments are compared, and the errors produced by the conventional analysis are shown. It should be mentioned that the errors are remarkable and that they appear either on the safe side or on the dangerous side. The moments in Figs. 14, 15 and 16 are never been found by the conventional analysis, which leads us to the conclusion that the economical design can only be attained through the rigorous three-dimensional analysis. Especially, the rigid frames in space accompanying side-sways should be

analysed three-dimensionally, otherwise, the design will come far from economy and considerable dangers will arise at times.

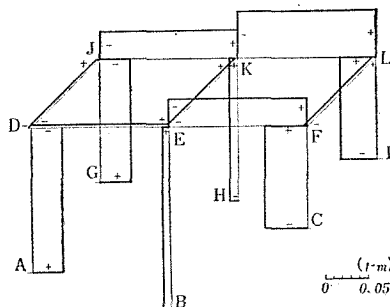
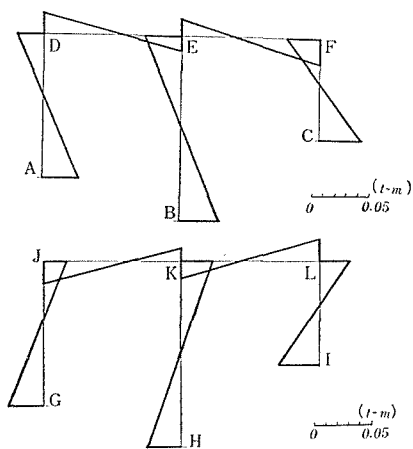
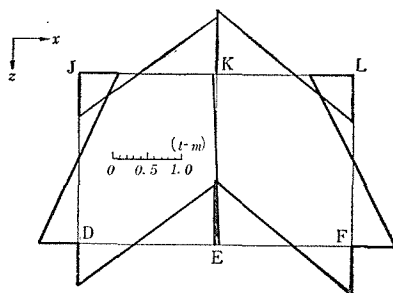
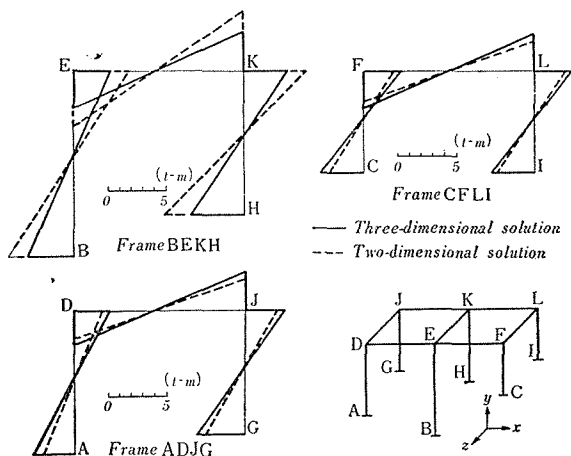


Table 9 Percentage errors

(a) Frame ADJG

	$M_{xAD}$	$M_{xDA}$	$M_{xDJ}$	$M_{xJD}$	$M_{xJG}$	$M_{xGJ}$
Three-dim. (t-m)	-3.71	-3.10	+3.10	+3.42	-3.39	-4.22
Two-dim. (t-m)	-2.98	-2.48	+2.48	+2.73	-2.73	-3.39
Error (%)	-19.7	-20.0	-20.0	-20.0	-19.5	-19.7

(b) Frame BEKH

	$M_{xBE}$	$M_{xEB}$	$M_{xEK}$	$M_{xKE}$	$M_{xKH}$	$M_{xHK}$
Three-dim. (t-m)	-4.02	-3.27	+3.30	+3.77	-3.86	-5.00
Two-dim. (t-m)	-5.84	-4.77	+4.77	+5.54	-5.53	-7.21
Error (%)	+45.2	+45.9	+44.6	+47.0	+43.2	+44.2

(c) Frame CFLI

	$M_{xCF}$	$M_{xFC}$	$M_{xFL}$	$M_{xLF}$	$M_{xLI}$	$M_{xIL}$
Three-dim. (t-m)	-3.96	-3.37	+3.34	+3.38	-3.33	-3.94
Two-dim. (t-m)	-3.24	-2.76	+2.75	+2.75	-2.75	-3.24
Error (%)	-18.2	-18.1	-17.7	-18.7	-17.4	-17.8

Note: Three-dim. : Three-dimensional solution  
Two-dim. : Two-dimensional solution

### 17. Unsymmetrical Rigid Frame in Space (2)

Fig. 17 shows an unsymmetrical rigid frame in space. The member-lengths and stiffnesses are tabulated in Table 10. The calculations will be carried out assuming, for simplicity,  $\beta = 0.1$ .

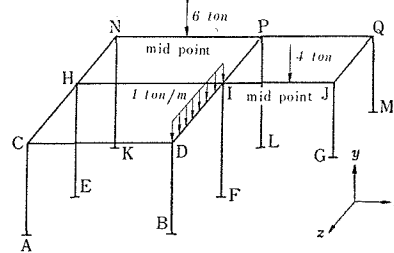


Fig. 17

Table 10

Member	AC	BD	CD	CH	CI	EH	FI	GJ	HI	IJ	HN	IP	JQ	KN	LP	MQ	NP	PQ
Length	5.0	5.0	8.0	5.0	5.0	6.0	6.0	4.0	8.0	6.0	4.0	4.0	4.0	6.0	6.0	4.0	8.0	6.0(m)
$k_x$	0.8	0.8	1.0	1.5	1.0	1.2	1.0	1.1	1.2	1.5	0.7	1.0	1.6	1.2	1.2	1.3	1.2	1.0
$k_y$	1.2	1.1	0.9	1.2	0.7	1.4	1.0	1.3	1.0	1.3	0.9	0.8	1.4	0.8	1.0	0.8	0.6	0.6
$k_z$	1.0	1.0	0.7	0.8	0.9	1.2	1.0	0.9	0.8	1.4	1.3	0.9	0.9	1.2	1.2	1.4	1.2	1.0

Though the frame seems to have twenty-six unknown  $\phi$ 's, they can be reduced to six by inspection or by referring to the compatibility condition in § 12. Taking  $\phi_{xAC}$ ,  $\phi_{xBD}$ ,  $\phi_{xGJ}$ ,  $\phi_{zAC}$ ,  $\phi_{zEH}$  and  $\phi_{zKN}$  for the independent unknowns, the others are denoted as follows:

$$\phi_{xEH} = (5/6) \times \phi_{xAC} = 0.833, 3\phi_{xAC},$$

$$\phi_{xKN} = (5/6) \times \phi_{xAC} = 0.833, 3\phi_{xAC},$$

$$\phi_{xFI} = (5/6) \times \phi_{xBD} = 0.833, 3\phi_{xBD},$$

$$\phi_{xLP} = (5/6) \times \phi_{xBD} = 0.833, 3\phi_{xBD},$$

$$\phi_{xMQ} = (4/4) \times \phi_{xGJ} = \phi_{xGJ},$$

$$\phi_{zBD} = (5/5) \times \phi_{zAC} = \phi_{zAC},$$

$$\phi_{zFI} = (6/6) \times \phi_{zEH} = \phi_{zEH},$$

$$\phi_{zGJ} = (6/4) \times \phi_{zEH} = 1.5\phi_{zEH},$$

$$\phi_{zLP} = (6/6) \times \phi_{zKN} = \phi_{zKN},$$

$$\phi_{zMQ} = (6/4) \times \phi_{zKN} = 1.5\phi_{zKN},$$

$$\phi_{yCD} = (5/8) \phi_{xAC} - (5/8) \phi_{xBD} = 0.625\phi_{xAC} - 0.625\phi_{xBD},$$

$$\phi_{yHI} = \phi_{yCD} = 0.625\phi_{xAC} - 0.625\phi_{xBD},$$

$$\phi_{yNP} = \phi_{yCD} = 0.625\phi_{xAC} - 0.625\phi_{xBD},$$

$$\phi_{yIJ} = (6/6) \phi_{xEI} - (4/6) \phi_{xGJ} = \phi_{xFI} - 0.666, 7\phi_{xGJ} = 0.833, 3\phi_{xBD} - 0.666, 7\phi_{xGJ},$$

$$\phi_{yPQ} = \phi_{yIJ} = 0.833, 3\phi_{xBD} - 0.666, 7\phi_{xGJ},$$

$$\phi_{yCH} = (6/5) \phi_{zEH} - (5/5) \phi_{zAC} = 1.2\phi_{zEH} - \phi_{zAC},$$

$$\phi_{yDI} = \phi_{yCH} = 1.2\phi_{zEH} - \phi_{zAC},$$

$$\phi_{yHN} = (6/4) \phi_{zKN} - (6/4) \phi_{zEH} = 1.5\phi_{zKN} - 1.5\phi_{zEH},$$

$$\phi_{yIP} = \phi_{yHN} = 1.5\phi_{zKN} - 1.5\phi_{zEH},$$

$$\phi_{yJQ} = \phi_{yHN} = 1.5\phi_{zKN} - 1.5\phi_{zEH}.$$

Using these, we have the necessary expressions which follow:

1) Expressions of End-Moments

For member AC :

$$M_{xAC} = 0.8(\varphi_{xC} + \phi_{xAC}),$$

$$M_{xCA} = 0.8(2\varphi_{xC} + \phi_{xAC}),$$

$$M_{yAC} = -0.24\varphi_{yC},$$

$$M_{yCA} = 0.24\varphi_{yC},$$

$$M_{zAC} = \varphi_{zC} + \phi_{zAC},$$

$$M_{zCA} = 2\varphi_{zC} + \phi_{zAC}.$$

For member BD :

$$M_{xBD} = 0.8(\varphi_{xD} + \phi_{xBD}),$$

$$M_{xDB} = 0.8(2\varphi_{xD} + \phi_{xBD}),$$

$$\begin{aligned}
M_{yBD} &= -0.22\varphi_{yD}, & M_{yDB} &= 0.22\varphi_{yD}, \\
M_{zBD} &= \varphi_{zD} + \phi_{zAC}, & M_{zDB} &= 2\varphi_{zD} + \phi_{zAC}.
\end{aligned}$$

For member CD :

$$\begin{aligned}
M_{xCD} &= 0.2(\varphi_{xC} - \varphi_{xD}), & M_{xDC} &= 0.2(\varphi_{xD} - \varphi_{xC}), \\
M_{yCD} &= 0.9(2\varphi_{yC} + \varphi_{yD} + 0.625\phi_{xAC} - 0.625\phi_{xBD}), \\
M_{yDC} &= 0.9(\varphi_{yC} + 2\varphi_{yD} + 0.625\phi_{xAC} - 0.625\phi_{xBD}), \\
M_{zCD} &= 0.7(2\varphi_{zC} + \varphi_{zD}), & M_{zDC} &= 0.7(\varphi_{zC} + 2\varphi_{zD}).
\end{aligned}$$

For member CH :

$$\begin{aligned}
M_{xCH} &= 1.5(2\varphi_{xC} + \varphi_{xH}), & M_{xHC} &= 1.5(\varphi_{xC} + 2\varphi_{xH}), \\
M_{yCH} &= 1.2(2\varphi_{yC} + \varphi_{yH} + 1.2\phi_{zEH} - \phi_{zAC}), \\
M_{yHC} &= 1.2(\varphi_{yC} + 2\varphi_{yH} + 1.2\phi_{zEH} - \phi_{zAC}), \\
M_{zCH} &= 0.16(\varphi_{zC} - \varphi_{zH}), & M_{zHC} &= 0.16(\varphi_{zH} - \varphi_{zC}).
\end{aligned}$$

For member DI :

$$\begin{aligned}
M_{xDI} &= 2\varphi_{xD} + \varphi_{xI} - 2.083, 3, & M_{xID} &= \varphi_{xD} + 2\varphi_{xI} + 2.083, 3, \\
M_{yDI} &= 0.7(2\varphi_{yD} + \varphi_{yI} + 1.2\phi_{zEH} - \phi_{zAC}), \\
M_{yID} &= 0.7(\varphi_{yD} + 2\varphi_{yI} + 1.2\phi_{zEH} - \phi_{zAC}), \\
M_{zDI} &= 0.18(\varphi_{zD} - \varphi_{zI}), & M_{zID} &= 0.18(\varphi_{zI} - \varphi_{zD}).
\end{aligned}$$

For member EH :

$$\begin{aligned}
M_{xEH} &= 1.2(\varphi_{xH} + 0.833, 3\phi_{xAC}), & M_{xHE} &= 1.2(2\varphi_{xH} + 0.833, 3\phi_{xAC}), \\
M_{yEH} &= -0.28\varphi_{yH}, & M_{yHE} &= 0.28\varphi_{yH}, \\
M_{zEH} &= 1.2(\varphi_{zH} + \phi_{zEH}), & M_{zHE} &= 1.2(2\varphi_{zH} + \phi_{zEH}).
\end{aligned}$$

For member FI :

$$\begin{aligned}
M_{xFI} &= \varphi_{xI} + 0.833, 3\phi_{xBD}, & M_{xIF} &= 2\varphi_{xI} + 0.833, 3\phi_{xBD}, \\
M_{yFI} &= -0.2\varphi_{yI}, & M_{yIF} &= 0.2\varphi_{yI}, \\
M_{zFI} &= \varphi_{zI} + \phi_{zEH}, & M_{zIF} &= 2\varphi_{zI} + \phi_{zEH}.
\end{aligned}$$

For member GJ :

$$\begin{aligned}
M_{xGJ} &= 1.1(\varphi_{xJ} + \phi_{xGI}), & M_{xJG} &= 1.1(2\varphi_{xJ} + \phi_{xGI}), \\
M_{yGJ} &= -0.26\varphi_{yJ}, & M_{yJG} &= 0.26\varphi_{yJ},
\end{aligned}$$

$$M_{zGJ} = 0.9(\varphi_{zJ} + 1.5\phi_{zEH}), \quad M_{zJG} = 0.9(2\varphi_{zJ} + 1.5\phi_{zEH}).$$

For member HI :

$$\begin{aligned} M_{xHI} &= 0.24(\varphi_{xH} - \varphi_{xI}), & M_{xIH} &= 0.24(\varphi_{xI} - \varphi_{xH}), \\ M_{yHI} &= 2\varphi_{yH} + \varphi_{yI} + 0.625\phi_{xAC} - 0.625\phi_{xBD}, \\ M_{yIH} &= \varphi_{yH} + 2\varphi_{yI} + 0.625\phi_{xAC} - 0.625\phi_{xBD}, \\ M_{zHI} &= 0.8(2\varphi_{zH} + \varphi_{zI}), & M_{zIH} &= 0.8(\varphi_{zH} + 2\varphi_{zI}). \end{aligned}$$

For member IJ :

$$\begin{aligned} M_{xIJ} &= 0.30(\varphi_{xI} - \varphi_{xJ}), & M_{xJI} &= 0.3(\varphi_{xJ} - \varphi_{xI}), \\ M_{yIJ} &= 1.3(2\varphi_{yI} + \varphi_{yJ} + 0.833, 3\phi_{xBD} - 0.666, 7\phi_{xGI}), \\ M_{yJI} &= 1.3(\varphi_{yI} + 2\varphi_{yJ} + 0.833, 3\phi_{xBD} - 0.666, 7\phi_{xGI}), \\ M_{zIJ} &= 1.4(2\varphi_{zI} + \varphi_{zJ}) - 3, & M_{zJI} &= 1.4(\varphi_{zI} + 2\varphi_{zJ}) + 3. \end{aligned}$$

For member HN :

$$\begin{aligned} M_{xHN} &= 0.7(2\varphi_{xH} + \varphi_{xN}), & M_{xNH} &= 0.7(\varphi_{xH} + 2\varphi_{xN}), \\ M_{yHN} &= 0.9(2\varphi_{yH} + \varphi_{yN} + 1.5\phi_{zKN} - 1.5\phi_{zEH}), \\ M_{yNH} &= 0.9(\varphi_{yH} + 2\varphi_{yN} + 1.5\phi_{zKN} - 1.5\phi_{zEH}), \\ M_{zHN} &= 0.26(\varphi_{zH} - \varphi_{zN}), & M_{zNH} &= 0.26(\varphi_{zN} - \varphi_{zH}). \end{aligned}$$

For member IP :

$$\begin{aligned} M_{xIP} &= 2\varphi_{xI} + \varphi_{xP}, & M_{xPI} &= \varphi_{xI} + 2\varphi_{xP}, \\ M_{yIP} &= 0.8(2\varphi_{yI} + \varphi_{yP} + 1.5\phi_{zKN} - 1.5\phi_{zEH}), \\ M_{yPI} &= 0.8(\varphi_{yI} + 2\varphi_{yP} + 1.5\phi_{zKN} - 1.5\phi_{zEH}), \\ M_{zIP} &= 0.18(\varphi_{zI} - \varphi_{zP}), & M_{zPI} &= 0.18(\varphi_{zP} - \varphi_{zI}). \end{aligned}$$

For member JQ :

$$\begin{aligned} M_{xJQ} &= 1.6(2\varphi_{xJ} + \varphi_{xQ}), & M_{xQJ} &= 1.6(\varphi_{xJ} + 2\varphi_{xQ}), \\ M_{yJQ} &= 1.4(2\varphi_{yJ} + \varphi_{yQ} + 1.5\phi_{zKN} - 1.5\phi_{zEH}), \\ M_{yQJ} &= 1.4(\varphi_{yJ} + 2\varphi_{yQ} + 1.5\phi_{zKN} - 1.5\phi_{zEH}), \\ M_{zJQ} &= 0.18(\varphi_{zJ} - \varphi_{zQ}), & M_{zQJ} &= 0.18(\varphi_{zQ} - \varphi_{zJ}). \end{aligned}$$

For member KN :

$$M_{xKN} = 1.2(\varphi_{xN} + 0.833, 3\phi_{xAC}), \quad M_{xNK} = 1.2(2\varphi_{xN} + 0.833, 3\phi_{xAC}),$$

$$\begin{aligned}
M_{yKN} &= -0.16\varphi_{yN}, & M_{yNK} &= 0.16\varphi_{yN}, \\
M_{zKN} &= 1.2(\varphi_{zN} + \phi_{zKN}), & M_{zNK} &= 1.2(2\varphi_{zN} + \phi_{zKN}).
\end{aligned}$$

For member LP :

$$\begin{aligned}
M_{xLP} &= 1.2(\varphi_{xP} + 0.833, 3\phi_{xBD}), & M_{xPL} &= 1.2(2\varphi_{xP} + 0.833, 3\phi_{xBD}), \\
M_{yLP} &= -0.2\varphi_{yP}, & M_{yPL} &= 0.2\varphi_{yP}, \\
M_{zLP} &= 1.2(\varphi_{zP} + \phi_{zKN}), & M_{zPL} &= 1.2(2\varphi_{zP} + \phi_{zKN}).
\end{aligned}$$

For member MQ :

$$\begin{aligned}
M_{xMQ} &= 1.3(\varphi_{xQ} + \phi_{xGJ}), & M_{xQM} &= 1.3(2\varphi_{xQ} + \phi_{xGJ}), \\
M_{yMQ} &= -0.16\varphi_{yQ}, & M_{yQM} &= 0.16\varphi_{yQ}, \\
M_{zMQ} &= 1.4(\varphi_{zQ} + 1.5\phi_{zKN}), & M_{zQM} &= 1.4(2\varphi_{zQ} + 1.5\phi_{zKN}).
\end{aligned}$$

For member NP :

$$\begin{aligned}
M_{xNP} &= 0.24(\varphi_{xN} - \varphi_{xP}), & M_{xPN} &= 0.24(\varphi_{xP} - \varphi_{xN}), \\
M_{yNP} &= 0.6(2\varphi_{yN} + \varphi_{yP} + 0.625\phi_{xAC} - 0.625\phi_{xBD}), \\
M_{yPN} &= 0.6(\varphi_{yN} + 2\varphi_{yP} + 0.625\phi_{xAC} - 0.625\phi_{xBD}), \\
M_{zNP} &= 1.2(2\varphi_{zH} + \varphi_{zP}) - 6, & M_{zPN} &= 1.2(\varphi_{zN} + 2\varphi_{zP}) + 6.
\end{aligned}$$

For member PQ :

$$\begin{aligned}
M_{xPQ} &= 0.2(\varphi_{xP} - \varphi_{xQ}), & M_{xQP} &= 0.2(\varphi_{xQ} - \varphi_{xP}), \\
M_{yPQ} &= 0.6(2\varphi_{yP} + \varphi_{yQ} + 0.833, 3\phi_{xBD} - 0.666, 7\phi_{xGJ}), \\
M_{yQP} &= 0.6(\varphi_{yP} + 2\varphi_{yQ} + 0.833, 3\phi_{xBD} - 0.666, 7\phi_{xGJ}), \\
M_{zPQ} &= 2\varphi_{zP} + \varphi_{zQ}, & M_{zQP} &= \varphi_{zP} + 2\varphi_{zQ}.
\end{aligned}$$

## 2) Expressions of End-Shears

Observing constituent frames in row:

$$\begin{aligned}
X_{xCA} &= -(1/l_{CA})(M_{zAC} + M_{zCA}) = -(1/5)(3\varphi_{zC} + 2\phi_{zAC}) = -0.6\varphi_{zC} - 0.4\phi_{zAC}, \\
X_{xDB} &= -(1/l_{DB})(M_{zBD} + M_{zDB}) = -(1/5)(3\varphi_{zD} + 2\phi_{zAC}) = -0.6\varphi_{zD} - 0.4\phi_{zAC}, \\
X_{xCH} &= -(1/l_{CH})(M_{yCH} + M_{yHC}) = -(1/5)(3.6\varphi_{yC} + 3.6\varphi_{yH} + 2.88\phi_{zEH} - 2.4\phi_{zAC}) \\
&= -0.72\varphi_{yC} - 0.72\varphi_{yH} - 0.576\phi_{zEH} + 0.48\phi_{zAC}, \\
X_{xDI} &= -(1/l_{DI})(M_{yDI} + M_{yID}) = -(1/5)(2.1\varphi_{yD} + 2.1\varphi_{yI} + 1.68\phi_{zEH} - 1.4\phi_{zAC}) \\
&= -0.42\varphi_{yD} - 0.42\varphi_{yI} - 0.336\phi_{zEH} + 0.28\phi_{zAC},
\end{aligned}$$



$$\begin{aligned}
X_{xHE} &= -(1/l_{HE})(M_{zHE} + M_{zEH}) = -(1.2/6)(3\varphi_{zH} + 2\phi_{zEH}) = -0.6\varphi_{zH} - 0.4\phi_{zEH}, \\
X_{xIF} &= -(1/l_{IF})(M_{zIF} + M_{zFI}) = -(1/6)(3\varphi_{zI} + 2\phi_{zEH}) = -0.5\varphi_{zI} - 0.333, 3\phi_{zEH}, \\
X_{xJG} &= -(1/l_{JG})(M_{zJG} + M_{zGJ}) = -(0.9/4)(3\varphi_{zJ} + 3\phi_{zEH}) = -0.675\varphi_{zJ} - 0.675\phi_{zEH}, \\
X_{xHC} &= X_{xCH} = -0.72\varphi_{yC} - 0.72\varphi_{yH} - 0.576\phi_{zEH} + 0.48\phi_{zAC}, \\
X_{xID} &= X_{xDI} = -0.42\varphi_{yD} - 0.42\varphi_{yI} - 0.336\phi_{zEH} + 0.28\phi_{zAC}, \\
X_{xHN} &= -(1/l_{HN})(M_{yHN} + M_{yNH}) = -(0.9/4)(3\varphi_{yH} + 3\varphi_{yN} + 3\phi_{zKN} - 3\phi_{zEH}) \\
&= -0.675\varphi_{yH} - 0.675\varphi_{yN} - 0.675\phi_{zKN} + 0.675\phi_{zEH}, \\
X_{xIP} &= -(1/l_{IP})(M_{yIP} + M_{yPI}) = -(0.8/4)(3\varphi_{yI} + 3\varphi_{yP} + 3\phi_{zKN} - 3\phi_{zEH}) \\
&= -0.6\varphi_{yI} - 0.6\varphi_{yP} - 0.6\phi_{zKN} + 0.6\phi_{zEH}, \\
X_{xJQ} &= -(1/l_{JQ})(M_{yJQ} + M_{yQJ}) = -(1.4/4)(3\varphi_{yJ} + 3\varphi_{yQ} + 3\phi_{zKN} - 3\phi_{zEH}) \\
&= -1.05\varphi_{yJ} - 1.05\varphi_{yQ} - 1.05\phi_{zKN} + 1.05\phi_{zEH}, \\
X_{xNK} &= -(1/l_{NK})(M_{zNK} + M_{zKN}) = -(1.2/6)(3\varphi_{zN} + 2\phi_{zKN}) = -0.6\varphi_{zN} - 0.4\phi_{zKN}, \\
X_{xPL} &= -(1/l_{PL})(M_{zPL} + M_{zLP}) = -(1.2/6)(3\varphi_{zP} + 2\phi_{zKN}) = -0.6\varphi_{zP} - 0.4\phi_{zKN}, \\
X_{xQM} &= -(1/l_{QM})(M_{zQM} + M_{zMQ}) = -(1.4/4)(3\varphi_{zQ} + 3\phi_{zKN}) \\
&= -1.05\varphi_{zQ} - 1.05\phi_{zKN}, \\
X_{xNH} &= X_{xHN} = -0.675\varphi_{yH} - 0.675\varphi_{yN} - 0.675\phi_{zKN} + 0.675\phi_{zEH}, \\
X_{xPI} &= X_{xIP} = -0.6\varphi_{yI} - 0.6\varphi_{yP} - 0.6\phi_{zKN} + 0.6\phi_{zEH}, \\
X_{xQJ} &= X_{xJQ} = -1.05\varphi_{yJ} - 1.05\varphi_{yQ} - 1.05\phi_{zKN} + 1.05\phi_{zEH}.
\end{aligned}$$

Similarly in column :

$$\begin{aligned}
X_{zCA} &= -(1/l_{CA})(M_{xAC} + M_{xCA}) = -(0.8/5)(3\varphi_{xC} + 2\phi_{xAC}) \\
&= -0.48\varphi_{xC} - 0.32\phi_{xAC}, \\
X_{zHE} &= -(1/l_{HE})(M_{xHE} + M_{xEH}) = -(1.2/6)(3\varphi_{xH} + 1.667\phi_{xAC}) \\
&= -0.6\varphi_{xH} - 0.333, 3\phi_{xAC}, \\
X_{zNK} &= -(1/l_{NK})(M_{xNK} + M_{xKN}) = -(1.2/6)(3\varphi_{xN} + 1.667\phi_{xAC}) \\
&= -0.6\varphi_{xN} - 0.333, 3\phi_{xAC}, \\
X_{zCD} &= -(1/l_{CD})(M_{yCD} + M_{yDC}) = -(0.9/8)(3\varphi_{yC} + 3\varphi_{yD} + 1.25\phi_{xAC} - 1.25\phi_{xBD}) \\
&= -0.337, 5\varphi_{yC} - 0.337, 5\varphi_{yD} - 0.140, 6\phi_{xAC} + 0.140, 6\phi_{xBD},
\end{aligned}$$

$$\begin{aligned} X_{zHI} &= -(1/l_{HI})(M_{yHI} + M_{yIH}) = -(1/8)(3\varphi_{yH} + 3\varphi_{yI} + 1.25\phi_{xAC} - 1.25\phi_{xBD}) \\ &= -0.375\varphi_{yH} - 0.375\varphi_{yI} - 0.156, 25\phi_{xAC} + 0.156, 25\phi_{xBD}, \end{aligned}$$

$$\begin{aligned} X_{zNP} &= -(1/l_{NP})(M_{yNP} + M_{yPN}) = -(0.6/8)(3\varphi_{yN} + 3\varphi_{yP} + 1.25\phi_{xAC} - 1.25\phi_{xBD}) \\ &= -0.225\varphi_{yN} - 0.225\varphi_{yP} - 0.093, 75\phi_{xAC} + 0.093, 75\phi_{xBD}, \end{aligned}$$

$$\begin{aligned} X_{zDB} &= -(1/l_{BD})(M_{xBD} + M_{xDB}) = -(0.8/5)(3\varphi_{xD} + 2\phi_{xBD}) \\ &= -0.48\varphi_{xD} - 0.32\phi_{xBD}, \end{aligned}$$

$$\begin{aligned} X_{zIF} &= -(1/l_{IF})(M_{xIF} + M_{xFI}) = -(1/6)(3\varphi_{xI} + 1.667\phi_{xBD}) \\ &= -0.5\varphi_{xI} - 0.277, 8\phi_{xBD}, \end{aligned}$$

$$\begin{aligned} X_{zPL} &= -(1/l_{PL})(M_{xPL} + M_{xLP}) = -(1.2/6)(3\varphi_{xP} + 1.667\phi_{xBD}) \\ &= -0.6\varphi_{xP} - 0.333, 3\phi_{xBD}, \end{aligned}$$

$$X_{zDC} = X_{zCD} = -0.337, 5\varphi_{yC} - 0.337, 5\varphi_{yD} - 0.140, 6\phi_{xAC} + 0.140, 6\phi_{xBD},$$

$$X_{zIH} = X_{zHI} = -0.375\varphi_{yH} - 0.375\varphi_{yI} - 0.156, 25\phi_{xAC} + 0.156, 25\phi_{xBD},$$

$$X_{zPN} = X_{zNP} = -0.225\varphi_{yN} - 0.225\varphi_{yP} - 0.093, 75\phi_{xAC} + 0.093, 75\phi_{xBD},$$

$$\begin{aligned} X_{zIJ} &= -(1/l_{IJ})(M_{yIJ} + M_{yJI}) = -(1.3/6)(3\varphi_{yI} + 3\varphi_{yJ} + 1.666, 7\phi_{xBD} - 1.333, 3\phi_{xGJ}) \\ &= -0.65\varphi_{yI} - 0.65\varphi_{yJ} - 0.361, 1\phi_{xBD} + 0.288, 9\phi_{xGJ}, \end{aligned}$$

$$\begin{aligned} X_{zPQ} &= -(1/l_{PQ})(M_{yPQ} + M_{yQP}) = -(0.6/6)(3\varphi_{yP} + 3\varphi_{yQ} + 1.667\phi_{xBD} - 1.333\phi_{xGJ}) \\ &= -0.3\varphi_{yP} - 0.3\varphi_{yQ} - 0.166, 7\phi_{xBD} + 0.133, 3\phi_{xGJ}, \end{aligned}$$

$$X_{zJG} = -(1/l_{JG})(M_{xJG} + M_{xGJ}) = -(1.1/4)(3\varphi_{xJ} + 2\phi_{xGJ}) = -0.825\varphi_{xJ} - 0.55\phi_{xGJ},$$

$$\begin{aligned} X_{zQM} &= -(1/l_{QM})(M_{xQM} + M_{xMQ}) = -(1.3/4)(3\varphi_{xQ} + 2\phi_{xGJ}) \\ &= -0.975\varphi_{xQ} - 0.65\phi_{xGJ}, \end{aligned}$$

$$X_{zJI} = X_{zIJ} = -0.65\varphi_{yI} - 0.65\varphi_{yJ} - 0.361, 1\phi_{xBD} + 0.288, 9\phi_{xGJ},$$

$$X_{zQP} = X_{zPQ} = -0.3\varphi_{yP} - 0.3\varphi_{yQ} - 0.166, 7\phi_{xBD} + 0.133, 3\phi_{xGJ}.$$

### 3) Elastic Equations

#### i) Joint Equilibrium Equations

About  $x$  axis :

$$\text{At joint C, } M_{xCA} + M_{xCD} + M_{xCH} = 0.$$

$$\text{〃 D, } M_{xDB} + M_{xDC} + M_{xDI} = 0.$$

At joint H,  $M_{xHC} + M_{xHE} + M_{xHI} + M_{xHN} = 0$ .

// I,  $M_{xID} + M_{xIF} + M_{xIH} + M_{xIJ} + M_{xIP} = 0$ .

// J,  $M_{xJG} + M_{xJI} + M_{xJQ} = 0$ .

// N,  $M_{xNH} + M_{xNK} + M_{xNP} = 0$ .

// P,  $M_{xPI} + M_{xPL} + M_{xPN} + M_{xPQ} = 0$ .

// Q,  $M_{xQJ} + M_{xQM} + M_{xQP} = 0$ .

About y axis :

At joint C,  $M_{yCA} + M_{yCD} + M_{yCH} = 0$ .

// D,  $M_{yDB} + M_{yDC} + M_{yDI} = 0$ .

// H,  $M_{yHC} + M_{yHE} + M_{yHI} + M_{yHN} = 0$ .

// I,  $M_{yID} + M_{yIF} + M_{yIH} + M_{yIJ} + M_{yIP} = 0$ .

// J,  $M_{yJG} + M_{yJI} + M_{yJQ} = 0$ .

// N,  $M_{yNH} + M_{yNK} + M_{yNP} = 0$ .

// P,  $M_{yPI} + M_{yPL} + M_{yPN} + M_{yPQ} = 0$ .

// Q,  $M_{yQJ} + M_{yQM} + M_{yQP} = 0$ .

About z axis :

At joint C,  $M_{zCA} + M_{zCD} + M_{zCH} = 0$ .

// D,  $M_{zDB} + M_{zDC} + M_{zDI} = 0$ .

// H,  $M_{zHC} + M_{zHE} + M_{zHI} + M_{zHN} = 0$ .

// I,  $M_{zID} + M_{zIF} + M_{zIH} + M_{zIJ} + M_{zIP} = 0$ .

// J,  $M_{zJG} + M_{zJI} + M_{zJQ} = 0$ .

// N,  $M_{zNH} + M_{zNK} + M_{zNP} = 0$ .

// P,  $M_{zPI} + M_{zPL} + M_{zPN} + M_{zPQ} = 0$ .

// Q,  $M_{zQJ} + M_{zQM} + M_{zQP} = 0$ .

## ii) Horizontal Shear Equations

For the 1st column-frame :

$$X_{zCA} + X_{zHE} + X_{zNK} + X_{zCD} + X_{zHI} + X_{zNP} = 0.$$

Table 11 Elastic equations

Eq.	Left-hand side															
	$\varphi_{xC}$	$\varphi_{xD}$	$\varphi_{xH}$	$\varphi_{xI}$	$\varphi_{xJ}$	$\varphi_{xN}$	$\varphi_{xP}$	$\varphi_{xQ}$	$\varphi_{yC}$	$\varphi_{yD}$	$\varphi_{yH}$	$\varphi_{yI}$	$\varphi_{yJ}$	$\varphi_{yN}$	$\varphi_{yP}$	$\varphi_{yQ}$
1	4.800	-0.200	1.500													
2	-0.200	3.800		1.000												
3	1.500		7.040	-0.240		0.700										
4		1.000	-0.240	6.540	-0.300		1.000									
5			-0.300	5.700				1.600								
6			0.700													
7				1.000		4.040	-0.240									
8					1.600	-0.240	4.840	-0.200								
9						-0.200	6.000									
10									4.440	0.900	1.200	0.700				
11									1.200		6.480	1.000		0.900		
12										0.700	1.000	7.800	1.300		0.800	
13											1.300	5.660				1.400
14											0.900			3.160	0.600	
15											0.800			0.600	4.200	0.600
16													1.400		0.600	4.160
17																
18																
19																
20																
21																
22																
23																
24																
25	0.800		1.000			1.000			0.563	0.563	0.625	0.625		0.375	0.375	
26		0.800		0.833			1.000	1.300	-0.563	-0.563	-0.625	0.458	1.083	-0.375	0.125	0.500
27					1.100				-1.200	-0.700	-1.200	-0.867	-0.867		-0.400	-0.400
28									1.440	0.840	0.090	-0.360	-2.100	-1.350	-1.200	-2.100
29											1.350	1.200	2.100	1.350	1.200	2.100
30																

Table 11—continued—

$\phi_{zC}$	$\phi_{zD}$	$\phi_{zH}$	$\phi_{zI}$	$\phi_{zJ}$	$\phi_{zN}$	$\phi_{zP}$	$\phi_{zQ}$	$\phi_{xAC}$	$\phi_{xBD}$	$\phi_{xGJ}$	$\phi_{zAC}$	$\phi_{zEH}$	$\phi_{zKN}$	Right-hand side
								0.800 1.000	0.800 0.833	1.100				2.083 -2.083
								1.000 0.563 0.563	1.000 -0.563 -0.563	1.300	-1.200 -0.700	1.440 0.840		
								0.625 0.625 1.083 0.375 0.375	-0.625 0.458 -0.867 -0.375 0.125	-0.867 -0.867 -0.400	-1.200 -0.700	0.090 -0.360 -2.100 -1.350 -1.200	1.350 1.200 2.100 1.350 1.200	
3.560 0.700 -0.160	0.700 3.580 -0.180	-0.160 4.420 0.800	-0.180 0.800 6.760	1.400	-0.260	-0.180			0.550	-0.400	1.000 1.000	-2.100 1.200 1.000	2.100	3.000
		-0.260	1.400 -0.180	4.780 -0.180	5.060 1.200	1.200 6.980 1.000	-0.180 1.000 4.980	2.296	-0.651			1.350	1.200 1.200 2.100	-3.000 6.000 -6.000
1.000	1.000	1.200	1.000	1.350	1.200	1.200	2.100	-0.651	3.083 -0.704	-0.704 2.163	2.600 -1.520	-1.520 9.290 -4.650	-4.650 8.350	

For the 2nd column-frame:

$$X_{zDB} + X_{zIF} + X_{zPL} - X_{zDC} - X_{zIH} - X_{zPN} + X_{zIJ} + X_{zPQ} = 0.$$

For the 3rd column-frame:

$$X_{zJG} + X_{zQM} - X_{zJI} - X_{zQP} = 0.$$

For the 1st row-frame :

$$X_{xCA} + X_{xDB} - X_{xCH} - X_{xDI} = 0.$$

For the 2nd row frame:

$$X_{xHE} + X_{xIF} + X_{xJG} + X_{xHC} + X_{xID} - X_{xHN} - X_{xIP} - X_{xJQ} = 0.$$

For the 3rd row frame:

$$X_{xNK} + X_{xPL} + X_{xQM} + X_{xNH} + X_{xPI} + X_{xQJ} = 0.$$

Substituting the expressions in 1) and 2), the required elastic equations are obtained as shown in Table 11, in which twenty-four from the beginning are the joint equilibrium equations and the remainders are the horizontal shear equations.

#### 4) Solutions Obtained

Simultaneous equations in Table 11 are solved, and we have the joint rotations;

at joint C,	$\varphi_{xC} = +0.049, 75,$	$\varphi_{yC} = -0.016, 63,$	$\varphi_{zC} = -0.012, 36,$
“ D,	$\varphi_{xD} = +0.697, 2,$	$\varphi_{yD} = -0.017, 36,$	$\varphi_{zC} = +0.025, 36,$
“ H,	$\varphi_{xH} = -0.014, 73,$	$\varphi_{yH} = +0.021, 97,$	$\varphi_{zH} = -0.027, 17,$
“ I,	$\varphi_{xI} = -0.425, 0,$	$\varphi_{yI} = +0.030, 89,$	$\varphi_{zI} = +0.577, 9,$
“ J,	$\varphi_{xJ} = -0.029, 68,$	$\varphi_{yJ} = +0.082, 26,$	$\varphi_{zJ} = -0.796, 9,$
“ N,	$\varphi_{xN} = +0.033, 81,$	$\varphi_{yN} = +0.065, 21,$	$\varphi_{zN} = +1.488, 1,$
“ P,	$\varphi_{xP} = +0.123, 7,$	$\varphi_{yP} = +0.046, 74,$	$\varphi_{zP} = -1.109, 2,$
“ Q,	$\varphi_{xQ} = +0.005, 594,$	$\varphi_{yQ} = +0.094, 59,$	$\varphi_{zQ} = +0.266, 2,$

and revolutions

$\psi_{xAC} = -0.096, 59,$	$\psi_{xBD} = -0.164, 4,$	$\psi_{xGJ} = +0.029, 73,$
$\psi_{zAC} = +0.021, 90,$	$\psi_{zEH} = +0.035, 56,$	$\psi_{zKN} = -0.171, 3.$

## 5) Results Obtained

Using these  $\phi$ 's and  $\psi$ 's, end-moments in 1) are now determined. The results are found in Table 12. In Fig. 18~20 the moment diagrams are shown by full lines. The reactions at supports are also computed and are shown in Fig. 21.

**Table 12** Values of end-moments (t-m)

Member	A C		B D		C D		C H	
End	A	C	B	D	C	D	C	H
$M_x$	-0.038	+0.002	+0.426	+0.984	-0.130	+0.130	+0.127	+0.030
$M_y$	+0.004	-0.004	+0.004	-0.004	-0.007	-0.008	+0.011	+0.058
$M_z$	+0.010	-0.003	+0.047	+0.073	+0.000	+0.027	+0.002	-0.002

Member	D I		E H		F I		G J	
End	D	I	E	H	F	I	G	J
$M_x$	-1.114	+1.931	-0.114	-0.132	-0.562	-0.987	+0.000	-0.033
$M_y$	+0.012	+0.046	-0.006	+0.006	-0.006	+0.006	-0.021	+0.021
$M_z$	-0.100	+0.100	+0.010	-0.023	+0.613	+1.191	-0.669	-1.386

Member	H I		I J		H N		I P	
End	H	I	I	J	H	N	I	P
$M_x$	+0.099	-0.099	-0.119	+0.119	+0.003	+0.037	-0.726	-0.178
$M_y$	+0.117	+0.126	-0.017	+0.050	-0.181	-0.142	-0.161	-0.149
$M_z$	+0.419	+0.903	-2.500	+1.578	-0.394	+0.394	+0.304	-0.304

Member	J Q		K N		L P		M Q	
End	J	Q	K	N	L	P	M	Q
$M_x$	-0.086	-0.030	-0.056	-0.015	-0.016	+0.133	+0.046	+0.053
$M_y$	-0.072	-0.056	-0.010	+0.010	-0.009	+0.009	-0.015	+0.015
$M_z$	-0.191	+0.191	+1.580	+3.370	-1.537	-2.870	+0.013	+0.386

Table 12—continued—

Member	N P		P Q	
End	N	P	P	Q
$M_x$	-0.022	+0.022	+0.024	-0.024
$M_y$	+0.132	+0.121	+0.019	+0.046
$M_z$	-3.760	+5.120	-1.952	-0.577

## 6) Comparison with the Current Solutions

The frame is analysed two-dimensionally and the results are compared with the values above obtained, see Table 13. Also, they are shown by broken lines in Fig. 18~20. Serious errors produced by the two-dimensional analysis are found in Table 13. Especially, the percentages enclosed by the brackets are noticeable. In the 2nd-and the 3rd-row-frames, the differences between exact values ( $M_z$ ) and those obtained by the conventional solution ( $M$ ) are remarkable. The reason will be such that while the 2nd-row-frame tends to deflect towards negative  $x$  and the 3rd-row-frame tends to positive  $x$ , the connecting girders tend to prevent these swayings. The existence of the connecting girders are not taken into account in the two-dimensional analysis.

Table 13 Percentage errors

(a) The 2nd-row-frame

Member	E H		F I		G J		H I		I J	
End	E	H	F	I	G	J	H	I	I	J
$M$	+0.109	-0.162	+0.971	+1.625	-0.422	-1.272	+0.162	+0.866	-2.491	+1.272
$M_z$	+0.010	-0.023	+0.613	+1.191	-0.669	-1.386	+0.419	+0.903	-2.500	+1.578
$(M-M_z)/M_z$	(+990)	(+604)	+58.4	+36.4	-36.9	-8.2	-61.3	-4.1	-0.4	-19.4(%)
$H$	-0.009		+0.433		-0.424					
$H_x$	-0.002		+0.300		-0.514					
$(H-H_x)/H_x$	(+350)		+33.3		-17.5					(%)
$V$	-0.129		+5.238		+1.797					
$V_y$	-0.144		+5.208		+1.875					
$(V-V_y)/V_y$	-10.4		+0.6		-4.2					(%)

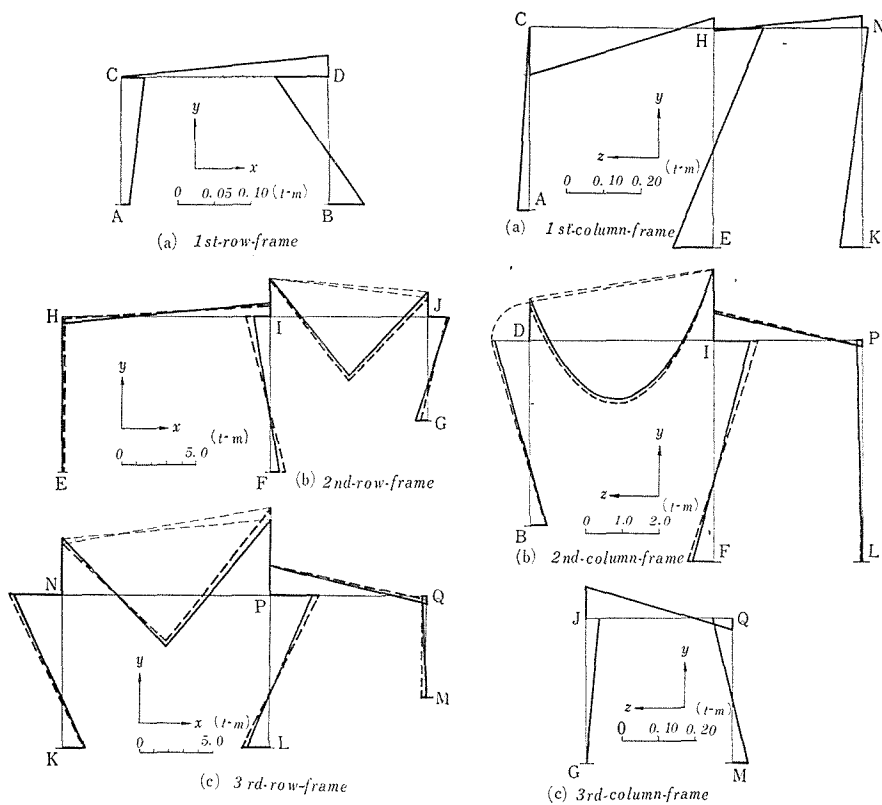
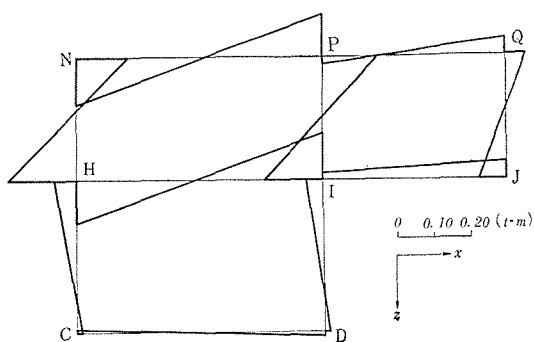


(b) The 3rd-row-frame

Member	K N		L P		M Q		N P		P Q	
End	K	N	L	P	M	Q	N	P	P	Q
$M$	+1.500	+3.465	-1.861	-3.258	-0.237	+0.339	-3.465	+5.173	-1.915	-0.340
$M_z$	+1.580	+3.370	-1.537	-2.870	+0.013	+0.386	-3.760	+5.120	-1.952	-0.577
$(M-M_z)/M_z$	-5.1	+2.8	+21.1	+13.5	—	-12.2	-7.8	+1.0	-1.9	-41.1(%)
$H$	+0.828		-0.853		+0.026					
$H_x$	+0.825		-0.735		+0.100					
$(H-H_x)/H_x$	+0.4		+16.1		-74.0					(%)
$V$	+3.214		+2.933		-0.376					
$V_y$	+2.840		+3.366		-0.450					
$(V-V_y)/V_y$	+13.2		-12.9		-16.4					(%)

(c) The 2nd-column-frame

Member	B D		F I		L P		D I		I P	
End	B	D	F	I	L	P	D	I	I	P
$M$	+0.413	+1.024	-0.673	-1.141	-0.052	+0.143	-1.024	+1.912	-0.772	-0.142
$M_x$	+0.426	+0.984	-0.562	-0.987	-0.016	+0.133	-1.114	+1.931	-0.726	-0.178
$(M-M_x)/M_x$	-3.1	+4.1	+19.8	+15.6	(+225)	+7.5	-8.1	-1.0	+6.3	-25.4(%)
$H$	-0.287		+0.302		-0.015					
$H_z$	-0.282		+0.258		-0.019					
$(H-H_z)/H_z$	+1.8		+17.1		-21.1					(%)
$V$	+2.322		+5.238		+2.933					
$V_y$	+2.340		+5.208		+3.366					
$(V-V_y)/V_y$	-0.8		+0.6		-12.9					(%)

Fig. 18  $M_z$  diagramFig. 19  $M_z$  diagramFig. 20  $M_y$  diagram

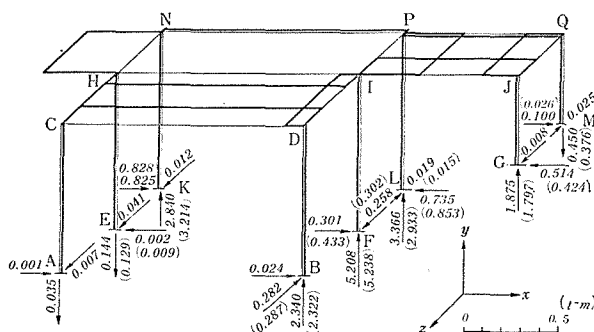


Fig. 21 Torsional moment diagram and reactions, values in ( ) are two-dimensional ones

### 18. Temperature Effects

To show the analysis of temperature effects, take, for example, the frame in Fig. 22. Let the girder BC and the column CG be subjected to the temperature rises of  $t_1 = 10^\circ\text{C}$  and  $t_2 = 15^\circ\text{C}$  respectively. The coefficient of the thermal expansion is  $\epsilon$ . The member-lengths and their relative stiffnesses are as shown in Table 14.

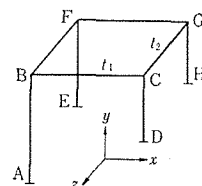


Fig. 22

Table 14

Member	AB	BC	CD	BF	CG	EF	FG	GH
Length	5.4	4.0	3.0	5.0	5.0	4.2	4.0	3.0 (m)
$k_x$	1.2	1.4	0.8	1.5	1.5	1.2	1.4	0.8
$k_y$	1.0	1.2	1.0	1.0	1.0	1.2	1.2	0.8
$k_z$	1.2	1.0	0.8	1.4	1.4	1.0	0.8	1.0

To analyse this frame, we have to find at first the relations among deflection angles by applying the conditions of compatibility described in §12, which are to be read from Table 15, and we have

$$R_{zCD} = 1.8R_{zAB} + 0.000,16, \quad R_{zBC} = 0,$$

$$R_{xGH} = R_{xCD} + 0.000,30, \quad R_{xCG} = 0,$$

$$R_{yCG} = R_{yBF} - 0.000,096, \quad R_{yFG} = R_{yBC} - 0.000,225.$$

Table 15 Compatibility conditions

Space	Member	Length(s) $x \quad y \quad z$			Direction angle ( $\alpha$ )	$\sin \alpha$	$\cos \alpha$	Revolution angle	Change in length ( $\Delta s$ )	$\Delta s \cos \alpha$	$R s \sin \alpha$	$\Delta s \sin \alpha$	$R s \cos \alpha$
ABCD ( $xy$ plane)	A B	0	5.4	0	$\alpha z = \pi/2$	1	0	$R_{zAB}$	0	0	$5.4R_{zAB}$	0	0
	B C	4.0	0	0	$\alpha z = 0$	0	1	$R_{zBC}$	$48 \times 10^{-5}$	$48 \times 10^{-5}$	0	0	$4.0R_{zBC}$
	C D	0	3.0	0	$\alpha z = 3\pi/2$	-1	0	$R_{zCD}$	0	0	$-3.0R_{zCD}$	0	0
									$\Sigma$	$48 \times 10^{-5} + 5.4R_{zAB} - 3.0R_{zCD} = \Delta l = 0$		$0 - 4.0R_{zBC} = \Delta h = 0$	
DCGH ( $yz$ plane)	D C	0	3.0	0	$\alpha x = \pi/2$	1	0	$R_{xDC}$	0	0	$3.0R_{xDC}$	0	0
	C G	0	0	5.0	$\alpha x = 0$	0	1	$R_{xCG}$	$90 \times 10^{-5}$	$90 \times 10^{-5}$	0	0	$5.0R_{xCG}$
	G H	0	3.0	0	$\alpha x = 3\pi/2$	-1	0	$R_{xGH}$	0	0	$-3.0R_{xGH}$	0	0
									$\Sigma$	$90 \times 10^{-5} + 3.0R_{xDC} - 3.0R_{xGH} = \Delta l = 0$		$0 - 5.0R_{xCG} = \Delta h = 0$	
BFCG ( $xz$ plane)	B F	0	0	5.0	$\alpha y = \pi/2$	1	0	$R_{yBF}$	0	0	$5.0R_{yBF}$	0	0
	F G	4.0	0	0	$\alpha y = 0$	0	1	$R_{yFG}$	0	0	0	0	$4.0R_{yFG}$
	G C	0	0	5.0	$\alpha y = 3\pi/2$	-1	0	$R_{yGC}$	$90 \times 10^{-5}$	0	$-5.0R_{yGC}$	$-90 \times 10^{-5}$	0
	C B	4.0	0	0	$\alpha y = \pi$	0	-1	$R_{yCB}$	$48 \times 10^{-5}$	$-48 \times 10^{-5}$	0	0	$-4.0R_{yCB}$
									$\Sigma$	$-48 \times 10^{-5} + 5.0R_{yBF} - 5.0R_{yGC} = \Delta l = 0$		$-90 \times 10^{-5} - 4.0R_{yFG} + 4.0R_{yCB} = \Delta h = 0$	

For remaining spaces ABFE and EFGH, the relations among deflection angles can easily be known without using the compatibility conditions. The relations thus found, taking  $R_{xAB}$ ,  $R_{xCD}$ ,  $R_{zAB}$  and  $R_{zEF}$  as the independent unknowns will be summarized in the following :

$$R_{yBC} = 1.35R_{xAB} - 0.75R_{xCD}, \quad R_{yBF} = 0.84R_{zEF} - 1.08R_{zAB},$$

$$R_{xEF} = 1.286R_{xAB}, \quad R_{zGH} = 1.4R_{zEF}.$$

Multiplying  $\mu = 6E\bar{K}/1000$  and transforming we have :

For member AB :  $\phi_{xAB}$ ,  $\phi_{zAB}$ .

$$\text{// BC : } \phi_{yBC} = 1.35\phi_{xAB} - 0.75\phi_{xCD}.$$

$$\text{// CD : } \phi_{xCD}, \phi_{zCD} = 1.8\phi_{zAB} - 0.16\mu.$$

$$\text{// BF : } \phi_{yBF} = 0.84\phi_{zEF} - 1.08\phi_{zAB}.$$

$$\text{// CG : } \phi_{yCG} = 0.84\phi_{zEF} - 1.08\phi_{zAB} + 0.096\mu.$$

$$\text{// EF : } \phi_{xEF} = 1.286\phi_{xAB}, \phi_{zEF}.$$

$$\text{// FG : } \phi_{yFG} = 1.35\phi_{xAB} - 0.75\phi_{xCD} + 0.225\mu.$$

$$\text{// GH : } \phi_{xGH} = \phi_{xCD} - 0.30\mu, \phi_{zGH} = 1.4\phi_{zEF}.$$

Now the analysis goes on as before.

#### 1) Expressions of End-Moments

For member AB :

$$M_{xAB} = 1.2(\varphi_{xB} + \phi_{xAB}),$$

$$M_{xBA} = 1.2(2\varphi_{xB} + \phi_{xAB}),$$

$$M_{yAB} = -0.2\varphi_{yB},$$

$$M_{yBA} = 0.2\varphi_{yB},$$

$$M_{zAB} = 1.2(\varphi_{zB} + \phi_{zAB}),$$

$$M_{zBA} = 1.2(2\varphi_{zB} + \phi_{zAB}).$$

For member BC :

$$M_{xBC} = 0.28(\varphi_{xB} - \varphi_{xC}),$$

$$M_{xCB} = 0.28(\varphi_{xC} - \varphi_{xB}),$$

$$M_{yBC} = 1.2(2\varphi_{yB} + \varphi_{yC} + 1.35\phi_{xAB} - 0.75\phi_{xCD}),$$

$$M_{yCB} = 1.2(\varphi_{yB} + 2\varphi_{yC} + 1.35\phi_{xAB} - 0.75\phi_{xCD}),$$

$$M_{zBC} = 2\varphi_{zB} + \varphi_{zC},$$

$$M_{zCB} = \varphi_{zB} + 2\varphi_{zC}.$$

For member CD :

$$M_{xCD} = 0.8(2\varphi_{xC} + \phi_{xCD}),$$

$$M_{xDC} = 0.8(\varphi_{xC} + \phi_{xCD}),$$

$$\begin{aligned}
M_{yCD} &= 0.2\varphi_{yC}, & M_{yDC} &= -0.2\varphi_{yC}, \\
M_{zCD} &= 0.8(2\varphi_{zC} + 1.8\phi_{zAB} - 0.16\mu), & M_{zDC} &= 0.8(\varphi_{zC} + 1.8\phi_{zAB} - 0.16\mu).
\end{aligned}$$

For member BF :

$$\begin{aligned}
M_{xBF} &= 1.5(2\varphi_{xB} + \varphi_{xF}), & M_{xFB} &= 1.5(\varphi_{xB} + 2\varphi_{xF}), \\
M_{yBF} &= 2\varphi_{yB} + \varphi_{yF} + 0.84\phi_{zEF} - 1.08\phi_{zAB}, \\
M_{yFB} &= \varphi_{yB} + 2\varphi_{yF} + 0.84\phi_{zEF} - 1.08\phi_{zAB}, \\
M_{zBF} &= 0.28(\varphi_{zB} - \varphi_{zF}), & M_{zFB} &= 0.28(\varphi_{zF} - \varphi_{zB}).
\end{aligned}$$

For member CG :

$$\begin{aligned}
M_{xCG} &= 1.5(2\varphi_{xC} + \varphi_{xG}), & M_{xGC} &= 1.5(\varphi_{xC} + 2\varphi_{xG}), \\
M_{yCG} &= 2\varphi_{yC} + \varphi_{yG} + 0.84\phi_{zEF} - 1.08\phi_{zAB} + 0.096\mu, \\
M_{yGC} &= \varphi_{yC} + 2\varphi_{yG} + 0.84\phi_{zEF} - 1.08\phi_{zAB} + 0.096\mu, \\
M_{zCG} &= 0.28(\varphi_{zC} - \varphi_{zG}), & M_{zGC} &= 0.28(\varphi_{zG} - \varphi_{zC}).
\end{aligned}$$

For member EF :

$$\begin{aligned}
M_{xEF} &= 1.2(\varphi_{xF} + 1.286\phi_{xAB}), & M_{xFE} &= 1.2(2\varphi_{xF} + 1.286\phi_{xAB}), \\
M_{yEF} &= -0.24\varphi_{yF}, & M_{yFE} &= 0.24\varphi_{yF}, \\
M_{zEF} &= \varphi_{zF} + \phi_{zEF}, & M_{zFE} &= 2\varphi_{zF} + \phi_{zEF}.
\end{aligned}$$

For member FG :

$$\begin{aligned}
M_{xFG} &= 0.28(\varphi_{xF} - \varphi_{xG}), & M_{xGF} &= 0.28(\varphi_{xG} - \varphi_{xF}), \\
M_{yFG} &= 1.2(2\varphi_{yF} + \varphi_{yG} + 1.35\phi_{xAB} - 0.75\phi_{xCD} + 0.225\mu), \\
M_{yGF} &= 1.2(\varphi_{yF} + 2\varphi_{yG} + 1.35\phi_{xAB} - 0.75\phi_{xCD} + 0.225\mu), \\
M_{zFG} &= 0.8(2\varphi_{zF} + \varphi_{zG}), & M_{zGF} &= 0.8(\varphi_{zF} + 2\varphi_{zG}).
\end{aligned}$$

For member GH :

$$\begin{aligned}
M_{xGH} &= 0.8(2\varphi_{xG} + \phi_{xCD} - 0.30\mu), & M_{xHG} &= 0.8(\varphi_{xG} + \phi_{xCD} - 0.30\mu), \\
M_{yGH} &= 0.16\varphi_{yG}, & M_{yHG} &= -0.16\varphi_{yG}, \\
M_{zGH} &= 2\varphi_{zG} + 1.4\phi_{zEF}, & M_{zHG} &= \varphi_{zG} + 1.4\phi_{zEF}.
\end{aligned}$$

## 2) Expressions of End-Shears

In  $x$  direction:

$$\begin{aligned} X_{xBA} &= -(1/l_{AB})(M_{zAB} + M_{zBA}) = -(1.2/5.4)(3\varphi_{zB} + 2\phi_{zAB}) \\ &= -0.667\varphi_{zB} - 0.444\phi_{zAB}, \end{aligned}$$

$$\begin{aligned} X_{xCD} &= -(1/l_{CD})(M_{zCD} + M_{zDC}) = -(0.8/3.0)(3\varphi_{zC} + 3.6\phi_{zAB} - 0.32\mu) \\ &= -0.8\varphi_{zC} - 0.96\phi_{zAB} + 0.085, 3\mu, \end{aligned}$$

$$\begin{aligned} X_{xBF} &= -(1/l_{BF})(M_{yBF} + M_{yFB}) = -(1/5.0)(3\varphi_{yB} + 3\varphi_{yF} + 1.68\phi_{zEF} - 2.16\phi_{zAB}) \\ &= -0.6\varphi_{yB} - 0.6\varphi_{yF} - 0.336\phi_{zEF} + 0.432\phi_{zAB}, \end{aligned}$$

$$\begin{aligned} X_{xCG} &= -(1/l_{CG})(M_{yCG} + M_{yGC}) = -(1/5.0)(3\varphi_{yC} + 3\varphi_{yG} + 1.68\phi_{zEF} - 2.16\phi_{zAB} \\ &\quad + 0.192\mu) = -0.6\varphi_{yC} - 0.6\varphi_{yG} - 0.336\phi_{zEF} + 0.432\phi_{zAB} - 0.038, 4\mu, \end{aligned}$$

$$X_{xFE} = -(1/l_{EF})(M_{zEF} + M_{zFE}) = -(1/4.2)(3\varphi_{zF} + 2\phi_{zEF}) = -0.714\varphi_{zF} - 0.476\phi_{zEF},$$

$$X_{xGH} = -(1/l_{GH})(M_{zGH} + M_{zHG}) = -(1/3.0)(3\varphi_{zG} + 2.8\phi_{zEF}) = -\varphi_{zG} - 0.933\phi_{zEF},$$

$$X_{xFB} = X_{xBF} = -0.6\varphi_{yB} - 0.6\varphi_{yF} - 0.336\phi_{zEF} + 0.432\phi_{zAB},$$

$$X_{xGC} = X_{xCG} = -0.6\varphi_{yC} - 0.6\varphi_{yG} - 0.336\phi_{zEF} + 0.432\phi_{zAB} - 0.038, 4\mu.$$

In  $z$  direction :

$$\begin{aligned} X_{zBA} &= -(1/l_{AB})(M_{xAB} + M_{xBA}) = -(1.2/5.4)(3\varphi_{xB} + 2\phi_{xAB}) \\ &= -0.667\varphi_{xB} - 0.444\phi_{xAB}, \end{aligned}$$

$$\begin{aligned} X_{zFE} &= -(1/l_{EF})(M_{xEF} + M_{xFE}) = -(1.2/4.2)(3\varphi_{xF} + 2.571\phi_{xAB}) \\ &= -0.857\varphi_{xF} - 0.735\phi_{xAB}, \end{aligned}$$

$$\begin{aligned} X_{zBC} &= -(1/l_{BC})(M_{yBC} + M_{yCB}) = -(1.2/4.0)(3\varphi_{yB} + 3\varphi_{yC} + 2.7\phi_{xAB} - 1.5\phi_{xCD}) \\ &= -0.9\varphi_{yB} - 0.9\varphi_{yC} - 0.81\phi_{xAB} + 0.45\phi_{xCD}, \end{aligned}$$

$$\begin{aligned} X_{zFG} &= -(1/l_{FG})(M_{yFG} + M_{yGF}) = -(1.2/4.0)(3\varphi_{yF} + 3\varphi_{yG} + 2.7\phi_{xAB} - 1.5\phi_{xCD} \\ &\quad + 0.45\mu) = -0.9\varphi_{yF} - 0.9\varphi_{yG} - 0.81\phi_{xAB} + 0.45\phi_{xCD} - 0.135\mu. \end{aligned}$$

$$\begin{aligned} X_{zCD} &= -(1/l_{CD})(M_{xCD} + M_{xDC}) = -(0.8/3.0)(3\varphi_{xC} + 2\phi_{xCD}) \\ &= -0.8\varphi_{xC} - 0.533\phi_{xCD}, \end{aligned}$$

$$\begin{aligned} X_{zGH} &= -(1/l_{GH})(M_{xGH} + M_{xHG}) = -(0.8/3.0)(3\varphi_{xG} + 2\phi_{xCD} - 0.6\mu) \\ &= -0.8\varphi_{xG} - 0.533\phi_{xCD} + 0.16\mu, \end{aligned}$$

$$X_{zCB} = X_{zBC} = -0.9\varphi_{yB} - 0.9\varphi_{yC} - 0.81\phi_{xAB} + 0.45\phi_{xCD},$$

$$X_{zGF} = X_{zFG} = -0.9\varphi_{yF} - 0.9\varphi_{yG} - 0.81\phi_{xAB} + 0.45\phi_{xCD} - 0.135\mu.$$

## 3) Elastic Equations

## i) Joint Equilibrium Equations

About  $x$  axis :

At joint B,  $M_{xBA} + M_{xBC} + M_{xBF} = 0.$

// C,  $M_{xCB} + M_{xCD} + M_{xCG} = 0.$

// F,  $M_{xFB} + M_{xFE} + M_{xFG} = 0.$

// G,  $M_{xGC} + M_{xGF} + M_{xGH} = 0.$

About  $y$  axis :

At joint B,  $M_{yBA} + M_{yBC} + M_{yBF} = 0.$

// C,  $M_{yCB} + M_{yCD} + M_{yCG} = 0.$

// F,  $M_{yFB} + M_{yFE} + M_{yFG} = 0.$

// G,  $M_{yGC} + M_{yGF} + M_{yGH} = 0.$

About  $z$  axis :**Table 16** Elastic equations

Eq.	Left-hand side							
	$\varphi_{xB}$	$\varphi_{xC}$	$\varphi_{xF}$	$\varphi_{xG}$	$\varphi_{yB}$	$\varphi_{yC}$	$\varphi_{yF}$	$\varphi_{yG}$
1	5.680	-0.280	1.500					
2	-0.280	4.880		1.500				
3	1.500		5.680	-0.280				
4		1.500	-0.280	4.880				
5					4.600	1.200	1.000	
6					1.200	4.600		1.000
7					1.000		4.640	1.200
8						1.000	1.200	4.560
9								
10								
11								
12								
13	1.200		1.543		1.620	1.620	1.620	1.620
14		0.800		0.800	-0.900	-0.900	-0.900	-0.900
15					-1.080	-1.080	-1.080	-1.080
16					0.840	0.840	0.840	0.840



At joint B,  $M_{zBA} + M_{zBC} + M_{zBF} = 0$ .

// C,  $M_{zCB} + M_{zCD} + M_{zCG} = 0$ .

// F,  $M_{zFB} + M_{zFE} + M_{zFG} = 0$ .

// G,  $M_{zGC} + M_{zGF} + M_{zGH} = 0$ .

## ii) Horizontal Shear Equations

In  $z$  direction :

For the 1st column-frame :  $X_{zBA} + X_{zFE} + X_{zBC} + X_{zFG} = 0$ .

// 2nd // :  $X_{zCD} + X_{zGH} - X_{zCB} - X_{zGF} = 0$ .

In  $x$  direction :

For the 1st row-frame :  $-X_{xBA} - X_{xCD} + X_{xBF} + X_{xCG} = 0$ .

// 2nd // :  $-X_{xFE} - X_{xGH} - X_{xFB} - X_{xGC} = 0$ .

These are summarized in Table 16.

**Table 16**—continued—

								Right-hand side
$\varphi_{zB}$	$\varphi_{zC}$	$\varphi_{zF}$	$\varphi_{zG}$	$\psi_{xAB}$	$\psi_{xCD}$	$\psi_{zAB}$	$\psi_{zEF}$	(multiplier: $\mu$ )
				1.200	0.800			
				1.543	0.800			
				1.620	-0.900	-1.080	0.840	0.240
				1.620	-0.900	-1.080	0.840	-0.096
				1.620	-0.900	-1.080	0.840	-0.270
				1.620	-0.900	-1.080	0.840	-0.366
4.680	1.000	-0.280				1.200		
1.000	3.880		-0.280			1.440		0.128
-0.280		3.880	0.800				1.000	
	-0.280	0.800	3.880				1.400	
				5.038	-1.620			-0.243
				-1.620	1.967			0.295
1.200	1.440					4.082	-1.210	0.223
		1.000	1.400			-1.210	2.913	-0.054

Solving these simultaneously, we obtain:

$$\begin{aligned}
 \text{At joint B, } & \varphi_{xB} = -0.001,865, \quad \varphi_{yB} = +0.042,13, \quad \varphi_{zB} = -0.018,57, \\
 // \quad \text{C, } & \varphi_{xC} = -0.036,07, \quad \varphi_{yC} = +0.020,51, \quad \varphi_{zC} = +0.016,12, \\
 // \quad \text{F, } & \varphi_{xF} = +0.003,216, \quad \varphi_{yF} = -0.013,79, \quad \varphi_{zF} = -0.002,133, \\
 // \quad \text{G, } & \varphi_{xG} = +0.035,36, \quad \varphi_{yG} = -0.036,25, \quad \varphi_{zG} = +0.000,693,3,
 \end{aligned}$$

and

$$\begin{aligned}
 \phi_{xAB} &= -0.003,609, & \phi_{xCD} &= +0.153,1, & \phi_{zAB} &= +0.058,48, \\
 \phi_{zEF} &= +0.002,521.
 \end{aligned}$$

Replacing these  $\varphi$ 's and  $\phi$ 's in the expressions of end-moments 2), we find the solutions shown in Table 17.

**Table 17** Values of end-moments (t-m)

Member	A B		B C		C D		B F	
End	A	B	B	C	C	D	B	F
$M_x$	-6.57	-8.81	+9.58	-9.58	+64.77	+93.62	-0.77	+6.85
$M_y$	-8.43	+8.43	-17.91	-43.86	+4.10	-4.10	+9.43	-46.49
$M_z$	+47.89	+25.61	-21.02	+13.67	-18.00	-30.89	-4.60	+4.60

Member	C G		E F		F G		G H	
End	C	G	E	F	F	G	G	H
$M_x$	-55.17	+51.98	-1.71	+2.15	-9.00	+9.00	-60.94	-89.23
$M_y$	+39.73	-17.03	+3.31	-3.31	+49.77	+22.82	-5.80	+5.80
$M_z$	+4.32	-4.32	+0.39	-1.75	-2.86	-0.60	+4.92	+4.22

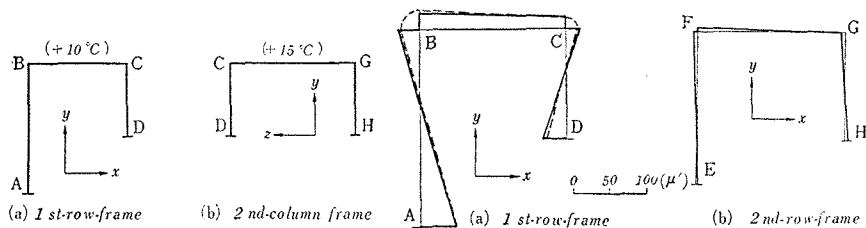
#### 4) Comparison with the Current Solutions

In the two-dimensional analysis, we treat only the two plane frames shown in Fig. 23 and the results obtained are compared with the exact values as shown in Table 18 and in Fig. 24~27. See that the two-dimensional analysis does not give the moments about  $y$  axis in Fig. 26 and that the magnitudes of them are as large as those of moments about the axes  $x$  and  $z$ , so that they can never be neglected. The deflection diagram is shown in Fig. 28.

**Table 18** Percentage errors

(a) The 1st-row-frame						Multiplier: $\mu'$
Member	A B		B C		C D	
End	A	B	B	C	C	D
$M$	+48.8	+24.4	-24.4	+13.2	-13.2	-26.6
$M_x$	+47.9	+25.6	-21.0	+13.7	-18.0	-30.9
$(M-M_x)/M_x$	+1.9	-4.7	+14.3	-3.6	-26.7	-13.9 (%)
$H$	+13.4					-13.3
$H_x$	+13.6					-16.3
$(H-H_x)/H_x$	-1.5					-18.4 (%)
$V$	+2.7					-2.7
$V_y$	+0.6					-1.2
$(V-V_y)/V_y$	(+350)					(+125) (%)

(b) The 2nd-column-frame						Multiplier: $\mu'$
Member	D C		C G		G H	
End	D	C	C	G	G	H
$M$	+89.0	+58.1	-58.1	+58.1	-58.1	-89.0
$M_x$	+93.6	+64.8	-55.2	+52.0	-60.9	-89.2
$(M-M_x)/M_x$	-4.9	-10.3	+5.3	+11.7	-4.6	-0.2 (%)
$H$	-49.0					+49.0
$H_x$	-52.8					+50.1
$(H-H_x)/H_x$	-7.2					-2.2 (%)
$V$	-2.7					0.0
$V_y$	-1.2					-1.5
$(V-V_y)/V_y$	(+125)					(-100) (%)

**Fig. 23****Fig. 24**

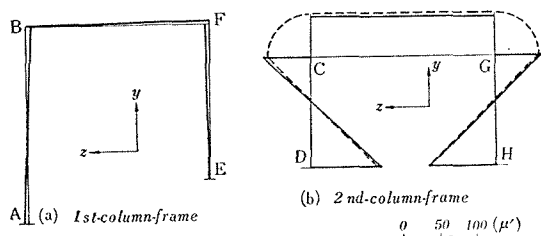
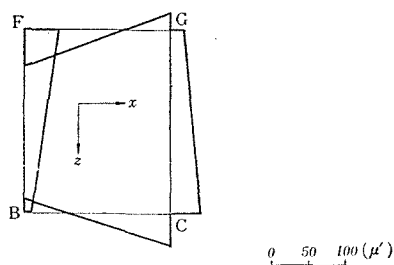
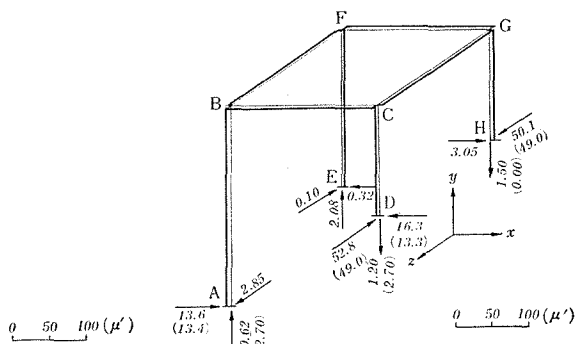
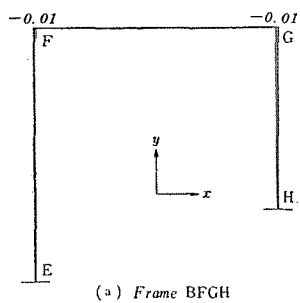
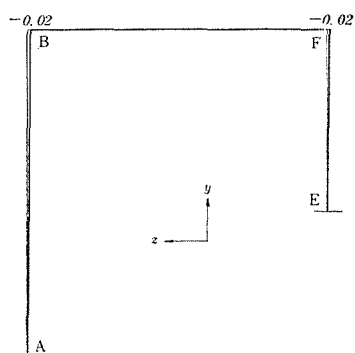
Fig. 25  $M_x$  diagramFig. 26  $M_y$  diagram

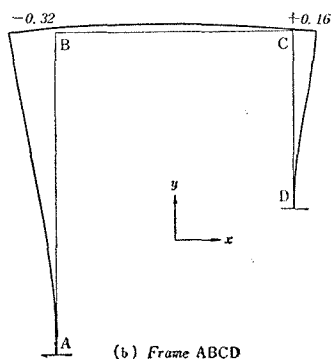
Fig. 27 Torsional moment diagram and reactions



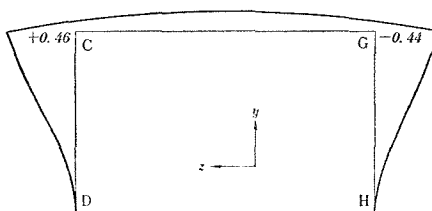
(a) Frame BFGH



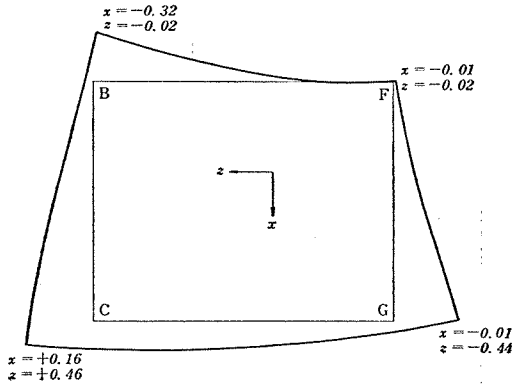
(c) Frame ABFE



(b) Frame ABCD



(d) Frame DCGH



(e) Frame BFGC

Scale of deflection :  $0 \quad 0.2 \quad 0.4 \text{ (mm)}$ 

Fig. 28 Deflection diagram

## 19. Settlements of Supports

The frame in Fig. 29 undergoes settlements at supports; they are 12 cm at C in negative  $z$  direction horizontally, and 6 cm at I vertically upward. The dimensions and stiffnesses are to be read from Table 19.

Table 19

Member	AD	BE	CF	DE	EF	DJ	EK	FL	JK	KL	GJ	HK	IL
Length	4.2	5.4	5.4	4.0	6.0	5.0	5.0	5.0	4.0	6.0	3.6	4.2	4.2 (m)
$k_x$	1.0	1.2	1.2	1.4	1.2	1.5	1.5	1.5	1.4	1.2	1.0	1.2	0.8
$k_y$	1.0	1.0	1.0	1.2	1.0	1.0	1.0	1.0	1.2	1.0	1.0	1.2	0.8
$k_z$	1.0	1.2	1.2	1.0	1.0	1.4	1.4	1.4	1.0	1.0	1.0	1.0	1.0

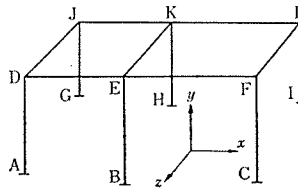


Fig. 29

As in the manner previously shown, the compatibility equations, Table 20, give the following relations :

$$\phi_{xIL} = 1.29\phi_{xCF} - 28.6\mu, \quad \phi_{xFL} = 12\mu,$$

$$\phi_{zIL} = 0.857\phi_{zGJ}, \quad \phi_{zKL} = 10\mu,$$

where  $\mu = 6EK/1000$ .

**Table 20** Compatibility conditions

Space	Mem- ber	Length ( $s$ )			Direction angle ( $\alpha$ )	$\sin$ $\alpha$	$\cos$ $\alpha$	Revo- lution angle	$Rs \sin \alpha$	$Rs \cos \alpha$
HKLI ( $xy$ plane)	HK	0	4.2	0	$\alpha z = \pi/2$	1	0	$R_{zHK}$	$4.2R_{zHK}$	0
	KL	6.0	0	0	$\alpha z = 0$	0	1	$R_{zKL}$	0	$6.0R_{zKL}$
	LI	0	4.2	0	$\alpha z = 3\pi/2$	-1	0	$R_{zLI}$	$-4.2R_{zLI}$	0
								$\Sigma$	$4.2R_{zHK} - 4.2R_{zLI}$ $= \Delta l = 0$	$-6.0R_{zKL} = \Delta h$ $= 0.06$
CFLI ( $yz$ plane)	CF	0	5.4	0	$\alpha x = \pi/2$	1	0	$R_{xCF}$	$5.4R_{xCF}$	0
	FL	0	0	5.0	$\alpha x = 0$	0	1	$R_{xFL}$	0	$5.0R_{xFL}$
	LI	0	4.2	0	$\alpha x = 3\pi/2$	-1	0	$R_{xLI}$	$-4.2R_{xLI}$	0
								$\Sigma$	$5.4R_{xCF} - 4.2R_{xLI}$ $= \Delta l = -0.12$	$-5.0R_{xFL} = \Delta h$ $= 0.06$

The relations below are readily found by inspecting the remaining spaces:

$$\begin{aligned}
 \phi_{zBE} &= 0.778\phi_{zAD}, & \phi_{zCF} &= \phi_{zBE}, & \phi_{yDE} &= 1.05\phi_{xAD} - 1.35\phi_{xBE}, \\
 \phi_{yEF} &= 0.9\phi_{xBE} - 0.9\phi_{xCF}, & \phi_{yDJ} &= -0.84\phi_{zAD} + 0.72\phi_{zGJ}, \\
 \phi_{yEK} &= \phi_{yDJ}, & \phi_{yJK} &= \phi_{yDE}, & \phi_{yKL} &= \phi_{yEF}, \\
 \phi_{xGJ} &= 1.167\phi_{xAD}, & \phi_{xHK} &= 1.286\phi_{xBE}, & \phi_{zHK} &= 0.857\phi_{zGJ}.
 \end{aligned}$$

### 1) Expressions of End-Moments

For member AD :

$$\begin{aligned}
 M_{xAD} &= \varphi_{xD} + \phi_{xAD}, & M_{xDA} &= 2\varphi_{xD} + \phi_{xAD}, \\
 M_{yAD} &= -0.2\varphi_{yD}, & M_{yDA} &= 0.2\varphi_{yD}, \\
 M_{zAD} &= \varphi_{zD} + \phi_{zAD}, & M_{zDA} &= 2\varphi_{zD} + \phi_{zAD}.
 \end{aligned}$$

For member BE :

$$\begin{aligned}
 M_{xBE} &= 1.2(\varphi_{xE} + \phi_{xBE}), & M_{xEB} &= 1.2(2\varphi_{xE} + \phi_{xBE}), \\
 M_{yBE} &= -0.2\varphi_{yE}, & M_{yEB} &= 0.2\varphi_{yE}, \\
 M_{zBE} &= 1.2(\varphi_{zE} + 0.778\phi_{zAD}), & M_{zEB} &= 1.2(2\varphi_{zE} + 0.778\phi_{zAD}).
 \end{aligned}$$

For member CF :

$$\begin{aligned}
 M_{xCF} &= 1.2(\varphi_{xF} + \phi_{xCF}), & M_{xFC} &= 1.2(2\varphi_{xF} + \phi_{xCF}), \\
 M_{yCF} &= 0.2\varphi_{yF}, & M_{yFC} &= 0.2\varphi_{yF},
 \end{aligned}$$

$$M_{zCF} = 1.2(\varphi_{zF} + 0.778\phi_{zAD}), \quad M_{zFC} = 1.2(2\varphi_{zF} + 0.778\phi_{zAD}),$$

For member DE :

$$M_{xDE} = 0.28(\phi_{xD} - \varphi_{xE}), \quad M_{xED} = 0.28(\varphi_{xE} - \phi_{xD}),$$

$$M_{yDE} = 1.2(2\varphi_{yD} + \varphi_{yE} + 1.05\phi_{xAD} - 1.35\phi_{xBE}),$$

$$M_{yED} = 1.2(\varphi_{yD} + 2\varphi_{yE} + 1.05\phi_{xAD} - 1.35\phi_{xBE}),$$

$$M_{zDE} = 2\varphi_{zD} + \varphi_{zE}, \quad M_{zED} = \varphi_{zD} + 2\varphi_{zE}.$$

For member EF :

$$M_{xEF} = 0.24(\varphi_{xE} - \varphi_{xF}), \quad M_{xFE} = 0.24(\varphi_{xF} - \varphi_{xE}),$$

$$M_{yEF} = 2\varphi_{yE} + \varphi_{yF} + 0.9\phi_{xBE} - 0.9\phi_{xCF},$$

$$M_{yFE} = \varphi_{yE} + 2\varphi_{yF} + 0.9\phi_{xBE} - 0.9\phi_{xCF},$$

$$M_{zEF} = 2\varphi_{zE} + \varphi_{zF}, \quad M_{zFE} = \varphi_{zE} + 2\varphi_{zF}.$$

For member DJ :

$$M_{xDJ} = 1.5(2\varphi_{xD} + \varphi_{xJ}), \quad M_{xJD} = 1.5(\varphi_{xD} + 2\varphi_{xJ}),$$

$$M_{yDJ} = 2\varphi_{yD} + \varphi_{yJ} - 0.84\phi_{zAD} + 0.72\phi_{zGJ},$$

$$M_{yJD} = \varphi_{yD} + 2\varphi_{yJ} - 0.84\phi_{zAD} + 0.72\phi_{zGJ},$$

$$M_{zDJ} = 0.28(\varphi_{zD} - \varphi_{zJ}), \quad M_{zJD} = 0.28(\varphi_{zJ} - \varphi_{zD}).$$

For member EK :

$$M_{xEK} = 1.5(2\varphi_{xE} + \varphi_{xK}), \quad M_{xKE} = 1.5(\varphi_{xE} + 2\varphi_{xK}),$$

$$M_{yEK} = 2\varphi_{yE} + \varphi_{yK} - 0.84\phi_{zAD} + 0.72\phi_{zGJ},$$

$$M_{yKE} = \varphi_{yE} + 2\varphi_{yK} - 0.84\phi_{zAD} + 0.72\phi_{zGJ},$$

$$M_{zEK} = 0.28(\varphi_{zE} - \varphi_{zK}), \quad M_{zKE} = 0.28(\varphi_{zK} - \varphi_{zE}).$$

For member FL :

$$M_{xFL} = 1.5(2\varphi_{xF} + \varphi_{xL} + 12\mu), \quad M_{xLF} = 1.5(\varphi_{xF} + 2\varphi_{xL} + 12\mu),$$

$$M_{yFL} = 2\varphi_{yF} + \varphi_{yL} - 0.84\phi_{zAD} + 0.72\phi_{zGJ},$$

$$M_{yLF} = \varphi_{yF} + 2\varphi_{yL} - 0.84\phi_{zAD} + 0.72\phi_{zGJ},$$

$$M_{zFL} = 0.28(\varphi_{zF} - \varphi_{zL}), \quad M_{zLF} = 0.28(\varphi_{zL} - \varphi_{zF}).$$

For member JK :

$$M_{xJK} = 0.28(\varphi_{xJ} - \varphi_{xK}), \quad M_{xKJ} = 0.28(\varphi_{xK} - \varphi_{xJ}),$$

$$M_{yJK} = 1.2(2\varphi_{yJ} + \varphi_{yK} + 1.05\phi_{xAD} - 1.35\phi_{xBE}),$$

$$M_{yKJ} = 1.2(\varphi_{yJ} + 2\varphi_{yK} + 1.05\phi_{xAD} - 1.35\phi_{xBE}),$$

$$M_{zJK} = 2\varphi_{zJ} + \varphi_{zK},$$

$$M_{zKJ} = \varphi_{zJ} + 2\varphi_{zK}.$$

For member KL :

$$M_{xKL} = 0.24(\varphi_{xK} - \varphi_{xL}),$$

$$M_{xLK} = 0.24(\varphi_{xL} - \varphi_{xK}),$$

$$M_{yKL} = 2\varphi_{yK} + \varphi_{yL} + 0.9\phi_{xBE} - 0.9\phi_{xCF},$$

$$M_{yLK} = \varphi_{yK} + 2\varphi_{yL} + 0.9\phi_{xBE} - 0.9\phi_{xCF},$$

$$M_{zKL} = 2\varphi_{zK} + \varphi_{zL} + 10\mu,$$

$$M_{zLK} = \varphi_{zK} + 2\varphi_{zL} + 10\mu.$$

For member GJ :

$$M_{xGJ} = \varphi_{xJ} + 1.167\phi_{xAD},$$

$$M_{xJG} = 2\varphi_{xJ} + 1.167\phi_{xAD},$$

$$M_{yGJ} = -0.2\varphi_{yJ},$$

$$M_{yJG} = 0.2\varphi_{yJ},$$

$$M_{zGJ} = \varphi_{zJ} + \phi_{zGJ},$$

$$M_{zJG} = 2\varphi_{zJ} + \phi_{zGJ}.$$

For member HK :

$$M_{xHK} = 1.2(\varphi_{xK} + 1.286\phi_{xBE}),$$

$$M_{xKH} = 1.2(2\varphi_{xK} + 1.286\phi_{xBE}),$$

$$M_{yHK} = -0.24\varphi_{yK},$$

$$M_{yKH} = 0.24\varphi_{yK},$$

$$M_{zHK} = \varphi_{zK} + 0.857\phi_{zGJ},$$

$$M_{zKH} = 2\varphi_{zK} + 0.857\phi_{zGJ}.$$

For member IL :

$$M_{xIL} = 0.8(\varphi_{xL} + 1.286\phi_{xCF} - 28.6\mu), \quad M_{xLI} = 0.8(2\varphi_{xL} + 1.286\phi_{xCF} - 28.6\mu),$$

$$M_{yIL} = -0.16\varphi_{yL},$$

$$M_{yLI} = 0.16\varphi_{yL}$$

$$M_{zIL} = \varphi_{zL} + 0.857\phi_{zGJ},$$

$$M_{zLI} = 2\varphi_{zL} + 0.857\phi_{zGJ}.$$

## 2) Expressions of End-Shears

In  $x$  direction :

$$\begin{aligned} X_{xDA} &= -(1/l_{AD})(M_{xAD} + M_{zDA}) = -(1.0/4.2)(3\varphi_{zD} + 2\phi_{zAD}) \\ &= -0.714\varphi_{zD} - 0.476\phi_{zAD}, \end{aligned}$$

$$\begin{aligned} X_{xEB} &= -(1/l_{BE})(M_{zBE} + M_{xEB}) = -(1.2/5.4)(3\varphi_{zE} + 1.556\phi_{zAD}) \\ &= -0.667\varphi_{zE} - 0.346\phi_{zAD}, \end{aligned}$$

$$\begin{aligned} X_{xFC} &= -(1/l_{CF})(M_{zCF} + M_{zFC}) = -(1.2/5.4)(3\varphi_{zF} + 1.556\phi_{zAD}) \\ &= -0.667\varphi_{zF} - 0.346\phi_{zAD}, \end{aligned}$$



$$\begin{aligned}
X_{xDJ} &= - (1/l_{DJ})(M_{yDJ} + M_{yJD}) = - (1/5.0)(3\varphi_{yD} + 3\varphi_{yJ} - 1.68\phi_{zAD} + 1.44\phi_{zGJ}) \\
&= - 0.6\varphi_{yD} - 0.6\varphi_{yJ} + 0.336\phi_{zAD} - 0.288\phi_{zGJ}, \\
X_{xEK} &= - (1/l_{EK})(M_{yEK} + M_{yKE}) = - (1/5.0)(3\varphi_{yE} + 3\varphi_{yK} - 1.68\phi_{zAD} + 1.44\phi_{zGJ}) \\
&= - 0.6\varphi_{yE} - 0.6\varphi_{yK} + 0.336\phi_{zAD} - 0.288\phi_{zGJ}, \\
X_{xFL} &= - (1/l_{FL})(M_{yFL} + M_{yLF}) = - (1/5.0)(3\varphi_{yF} + 3\varphi_{yL} - 1.68\phi_{zAD} + 1.44\phi_{zGJ}) \\
&= - 0.6\varphi_{yF} - 0.6\varphi_{yL} + 0.336\phi_{zAD} - 0.288\phi_{zGJ}, \\
X_{xJG} &= - (1/l_{JG})(M_{zGJ} + M_{zJG}) = - (1/3.6)(3\varphi_{zJ} + 2\phi_{zGJ}) = - 0.833\varphi_{zJ} - 0.556\phi_{zGJ}, \\
X_{xKH} &= - (1/l_{HK})(M_{zHK} + M_{zKH}) = - (1/4.2)(3\varphi_{zK} + 1.714\phi_{zGJ}) \\
&= - 0.714\varphi_{zK} - 0.408\phi_{zGJ}, \\
X_{xLI} &= - (1/l_{LI})(M_{zIL} + M_{zLI}) = - (1/4.2)(3\varphi_{zL} + 1.714\phi_{zGJ}) \\
&= - 0.714\varphi_{zL} - 0.408\phi_{zGJ}, \\
X_{xJD} &= X_{xDJ} = - 0.6\varphi_{yD} - 0.6\varphi_{yJ} + 0.336\phi_{zAD} - 0.288\phi_{zGJ}, \\
X_{xKE} &= X_{xEK} = - 0.6\varphi_{yE} - 0.6\varphi_{yK} + 0.336\phi_{zAD} - 0.288\phi_{zGJ}, \\
X_{xLF} &= X_{xFL} = - 0.6\varphi_{yF} - 0.6\varphi_{yL} + 0.336\phi_{zAD} - 0.288\phi_{zGJ}.
\end{aligned}$$

In  $z$  direction :

$$\begin{aligned}
X_{xDA} &= - (1/l_{DA})(M_{xDA} + M_{xAD}) = - (1/4.2)(3\varphi_{xD} + 2\phi_{xAD}) \\
&= - 0.714\varphi_{xD} - 0.476\phi_{xAD}, \\
X_{zJG} &= - (1/l_{JG})(M_{xJG} + M_{xGJ}) = - (1/3.6)(3\varphi_{xJ} + 2.333\phi_{xAD}) \\
&= - 0.833\varphi_{xJ} - 0.648\phi_{xAD}, \\
X_{zDE} &= - (1/l_{DE})(M_{yDE} + M_{yED}) = - (1.2/4.0)(3\varphi_{yD} + 3\varphi_{yE} + 2.10\phi_{xAD} - 2.70\phi_{xBE}) \\
&= - 0.9\varphi_{yD} - 0.9\varphi_{yE} - 0.63\phi_{xAD} + 0.81\phi_{xBE}, \\
X_{zJK} &= - (1/l_{JK})(M_{yJK} + M_{yKJ}) = - (1.2/4.0)(3\varphi_{yJ} + 3\varphi_{yK} + 2.10\phi_{xAD} - 2.70\phi_{xBE}) \\
&= - 0.9\varphi_{yJ} - 0.9\varphi_{yK} - 0.63\phi_{xAD} + 0.81\phi_{xBE}, \\
X_{xEB} &= - (1/l_{BE})(M_{xBE} + M_{xEB}) = - (1.2/5.4)(3\varphi_{xE} + 2\phi_{xBE}) \\
&= - 0.667\varphi_{xE} - 0.444\phi_{xBE}, \\
X_{zKH} &= - (1/l_{HK})(M_{xHK} + M_{xKH}) = - (1.2/4.2)(3\varphi_{xK} + 2.572\phi_{xBE}) \\
&= - 0.857\varphi_{xK} - 0.735\phi_{xBE},
\end{aligned}$$

$$X_{zED} = X_{zDE} = -0.9\varphi_{yD} - 0.9\varphi_{yE} - 0.63\phi_{xAD} + 0.81\phi_{xBE},$$

$$X_{zKJ} = X_{zJK} = -0.9\varphi_{yJ} - 0.9\varphi_{yK} - 0.63\phi_{xAD} + 0.81\phi_{xBE},$$

$$\begin{aligned} X_{zEF} &= -(1/l_{EF})(M_{yEF} + M_{yFE}) = -(1/6.0)(3\varphi_{yE} + 3\varphi_{yF} + 1.8\phi_{xBF} - 1.8\phi_{xCF}) \\ &= -0.5\varphi_{yE} - 0.5\varphi_{yF} - 0.3\phi_{xBE} + 0.3\phi_{xCF}, \end{aligned}$$

$$\begin{aligned} X_{zKL} &= -(1/l_{KL})(M_{yKL} + M_{yLK}) = -(1/6.0)(3\varphi_{yK} + 3\varphi_{yL} + 1.8\phi_{xBE} - 1.8\phi_{xCF}) \\ &= -0.5\varphi_{yK} - 0.5\varphi_{yL} - 0.3\phi_{xBE} + 0.3\phi_{xCF}, \end{aligned}$$

$$\begin{aligned} X_{zFC} &= -(1/l_{CF})(M_{xCF} + M_{xFC}) = -(1/2/5.4)(3\varphi_{xF} + 2\phi_{xCF}) \\ &= -0.667\varphi_{xF} - 0.444\phi_{xCF}, \end{aligned}$$

$$\begin{aligned} X_{zLI} &= -(1/l_{IL})(M_{xIL} + M_{xLI}) = -(0.8/4.2)(3\varphi_{xL} + 2.572\phi_{xCF} - 57.2\mu) \\ &= -0.571\varphi_{xL} - 0.490\phi_{xCF} + 10.895\mu, \end{aligned}$$

$$X_{zFE} = X_{zEF} = -0.5\varphi_{yE} - 0.5\varphi_{yF} - 0.3\phi_{xBE} + 0.3\phi_{xCF},$$

$$X_{zLK} = X_{zKL} = -0.5\varphi_{yK} - 0.5\varphi_{yL} - 0.3\phi_{xBE} + 0.3\phi_{xCF}.$$

### 3) Elastic Equations, see Table 21

#### i) Joint Equilibrium Equations

About  $x$  axis :

$$\begin{aligned} \text{At joint D, } & M_{xDA} + M_{xDE} + M_{xDJ} = 0. \\ \text{„ E, } & M_{xEB} + M_{xED} + M_{xEF} + M_{xEK} = 0. \\ \text{„ F, } & M_{xFC} + M_{xFE} + M_{xFL} = 0. \\ \text{„ J, } & M_{xJD} + M_{xJG} + M_{xJK} = 0. \\ \text{„ K, } & M_{xKE} + M_{xKH} + M_{xKJ} + M_{xKL} = 0. \\ \text{„ L, } & M_{xLF} + M_{xLI} + M_{xLK} = 0. \end{aligned}$$

About  $y$  axis :

$$\begin{aligned} \text{At joint D, } & M_{yDA} + M_{yDE} + M_{yDJ} = 0. \\ \text{„ E, } & M_{yEB} + M_{yED} + M_{yEF} + M_{yEK} = 0. \\ \text{„ F, } & M_{yFC} + M_{yFE} + M_{yFL} = 0. \\ \text{„ J, } & M_{yJD} + M_{yJG} + M_{yJK} = 0. \\ \text{„ K, } & M_{yKE} + M_{yKH} + M_{yKJ} + M_{yKL} = 0. \\ \text{„ L, } & M_{yLF} + M_{yLI} + M_{yLK} = 0. \end{aligned}$$

About  $z$  axis :

$$\text{At joint D, } M_{zDA} + M_{zDE} + M_{zDJ} = 0.$$

$$// \quad \text{E, } M_{zEB} + M_{zED} + M_{zEF} + M_{zEK} = 0.$$

$$// \quad \text{F, } M_{zFC} + M_{zFE} + M_{zFL} = 0.$$

$$// \quad \text{J, } M_{zJD} + M_{zJG} + M_{zJK} = 0.$$

$$// \quad \text{K, } M_{zKE} + M_{zKH} + M_{zKJ} + M_{zKL} = 0.$$

$$// \quad \text{L, } M_{zLF} + M_{zLI} + M_{zLK} = 0.$$

ii) Horizontal Shear Equations

In  $z$  direction :

$$\text{For the 1st column-frame : } X_{zDA} + X_{zJG} + X_{zDE} + X_{zJK} = 0.$$

$$// \quad \text{2nd} \quad // \quad : X_{zEB} + X_{zKH} - X_{zED} - X_{zKJ} + X_{zEF} + X_{zKL} = 0.$$

$$// \quad \text{3rd} \quad // \quad : X_{zFC} + X_{zLI} - X_{zFE} - X_{zLK} = 0.$$

In  $x$  direction :

$$\text{For the 1st row-frame : } -X_{xDA} - X_{xEb} - X_{xFC} + X_{xDJ} + X_{xEK} + X_{xFL} = 0.$$

$$// \quad \text{2nd} \quad // \quad : X_{xJG} + X_{xKH} + X_{xLH} + X_{xJD} + X_{xKE} + X_{xLF} = 0.$$

The simultaneous equations in Table 21 are solved, and we have the rotations;

$$\text{at joint D, } \varphi_{xD} = -0.020, 71, \quad \varphi_{yD} = +0.974, 7, \quad \varphi_{zD} = -0.619, 4,$$

$$// \quad \text{E, } \varphi_{xE} = -0.700, 5, \quad \varphi_{yE} = +1.829, \quad \varphi_{zE} = -0.285, 1,$$

$$// \quad \text{F, } \varphi_{xF} = -6.073, \quad \varphi_{yF} = +1.966, \quad \varphi_{zF} = -0.663, 6,$$

$$// \quad \text{J, } \varphi_{xJ} = -0.010, 19, \quad \varphi_{yJ} = +0.981, 3, \quad \varphi_{zJ} = +0.323, 4,$$

$$// \quad \text{K, } \varphi_{xK} = -0.601, 2, \quad \varphi_{yK} = +1.809, \quad \varphi_{zK} = -1.299,$$

$$// \quad \text{L, } \varphi_{xL} = +0.015, 21, \quad \varphi_{yL} = +1.997, \quad \varphi_{zL} = -2.025,$$

and revolutions

$$\phi_{xAD} = -0.071, 53, \quad \phi_{xBE} = +2.988, \quad \phi_{xCF} = +13.383,$$

$$\phi_{zAD} = +3.027, \quad \phi_{zGJ} = -0.258, 7.$$

Using these  $\varphi$ 's and  $\phi$ 's, end-moments are determined. Results are read from Table 22.

Table 21 Elastic equations

Eq.	Left-hand side											
	$\varphi_{xD}$	$\varphi_{xE}$	$\varphi_{xF}$	$\varphi_{xJ}$	$\varphi_{xK}$	$\varphi_{xL}$	$\varphi_{yD}$	$\varphi_{yE}$	$\varphi_{yF}$	$\varphi_{yJ}$	$\varphi_{yK}$	$\varphi_{yL}$
1	5.280	-0.280		1.500								
2	-0.280	5.920	-0.240		1.500							
3		-0.240	5.640			1.500						
4	1.500			5.280	-0.280							
5		1.500		-0.280	5.920	-0.240						
6			1.500		-0.240	4.840						
7							4.600	1.200		1.000		
8							1.200	6.600	1.000		1.000	
9								1.000	4.200			1.000
10							1.000			4.600	1.200	
11								1.000		1.200	6.640	1.000
12									1.000		1.000	4.160
13												
14												
15												
16												
17												
18												
19	1.000			1.167			1.260	1.260		1.260	1.260	
20		1.200			1.543		-1.620	-0.720	0.900	-1.620	-0.720	0.900
21			1.200			1.029		-0.900	-0.900		-0.900	-0.900
22							-0.840	-0.840	-0.840	-0.840	-0.840	-0.840
23							0.720	0.720	0.720	0.720	0.720	0.720

Table 21—continued—

$\phi_{zD}$	$\phi_{zE}$	$\phi_{zF}$	$\phi_{zJ}$	$\phi_{zK}$	$\phi_{zL}$	$\phi_{xAD}$	$\phi_{xBE}$	$\phi_{xCF}$	$\phi_{zAD}$	$\phi_{zGJ}$	Right-hand side (multiplier : $\mu$ )
						1.000 1.167	1.200 1.543	1.200			-18.000 4.880
						1.260 1.260 1.260	-1.620 -0.720 0.900 -1.620	1.029 -0.900 -0.900	-0.840 -0.840 -0.840 -0.840	0.720 0.720 0.720 0.720	
4.280 1.000	1.000 6.680 1.000	1.000 4.680	-0.280	-0.280	-0.280	1.260	-0.720 0.900	-0.900 -0.900	-0.840 -0.840 1.000 0.933 0.933	0.720 0.720	
-0.280	-0.280	-0.280	4.280 1.000	1.000 1.280 1.000	1.000 4.280	3.339 -2.268	-2.268 6.118	-1.080		1.000 0.857 0.857	-10.000 -10.000
1.000	0.933	0.933	1.000	0.857	0.857		-1.080	2.761	3.046 -1.210	-1.210 2.683	19.611

**Table 22** Values of end-moments (multiplier :  $\mu$ )

Member	A D		B E		C F		D E	
End	A	D	B	E	C	F	D	E
$M_x$	-0.092	-0.113	+2.745	+1.904	+8.772	+1.484	+0.190	-0.190
$M_y$	-0.195	+0.195	-0.366	+0.366	-0.393	+0.393	-0.397	+0.629
$M_z$	+2.408	+1.788	+2.484	+2.142	+2.030	+1.233	-1.524	-1.190

Member	E F		D J		E K		F L	
End	E	F	D	J	E	K	F	L
$M_x$	+1.289	-1.289	-0.077	-0.062	-3.003	-2.854	-0.196	+8.936
$M_y$	-3.732	-3.595	-0.202	+0.208	+2.738	+2.718	+3.200	+3.231
$M_z$	-1.234	-1.612	-0.264	+0.264	+0.284	-0.284	+0.381	-0.381

Member	J K		K L		G J		H K	
End	J	K	K	L	G	J	H	K
$M_x$	+0.165	-0.165	-0.148	+0.148	-0.094	-0.104	+3.890	+3.168
$M_y$	-0.405	+0.588	-3.741	-3.553	-0.196	+0.196	-0.434	+0.434
$M_z$	-0.652	-2.275	+5.377	+4.651	+0.065	+0.388	-1.521	-2.820

Member	I L	
End	I	L
$M_x$	-9.099	-9.087
$M_y$	-0.320	+0.320
$M_z$	-2.247	-4.272

#### 4) Comparison with the Current Solutions

To solve this frame by the current two-dimensional method, we treat two constituent plane frames in Fig. 30 separately and the results and the errors are shown in Table 23. The moment diagrams are drawn by broken lines in Figs. 31, 32, 33 and 34.

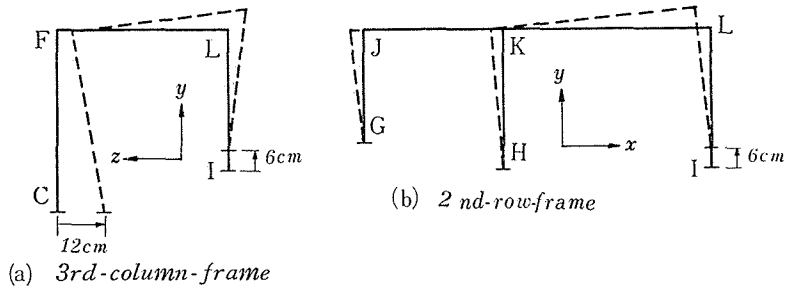


Fig.30 Deflected skeleton

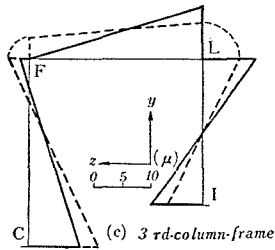
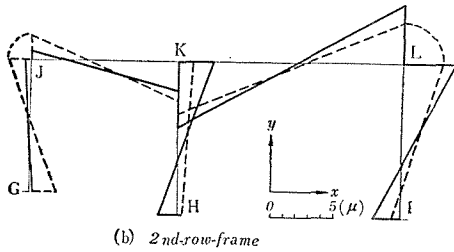
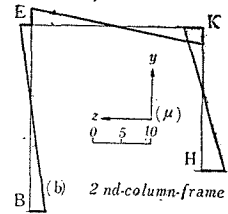
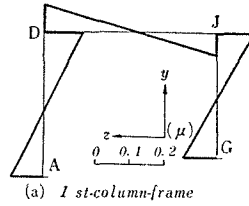
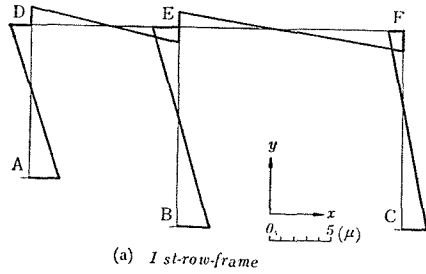
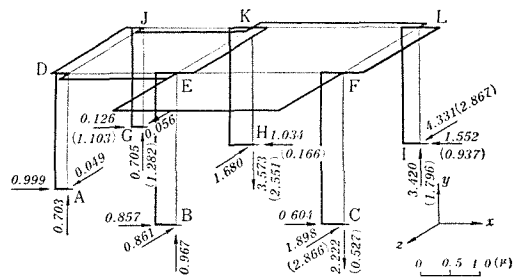
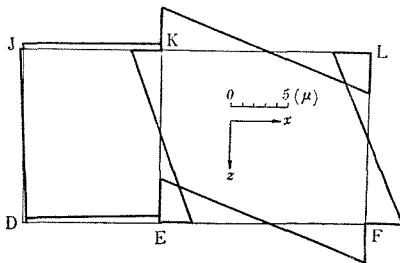
Table 23 Percentage errors of end-moments and reactions

 (a) The 2nd-row-frame (multiplier :  $\mu$ )

Member	G J		J K		H K		K L		I L	
End	G	J	J	K	H	K	K	L	I	L
$M$	+2.077	+1.893	-1.893	-3.234	+0.413	-1.112	+4.347	+3.269	-0.665	-3.269
$M_z$	+0.065	+0.388	-0.652	-2.275	-1.521	-2.820	+5.377	+4.651	-2.247	-4.272
$(M-M_z)/M_z$	(+3095)	(+388)	(+190)	+42.2	—	-60.6	-19.2	-29.7	-70.4	-23.5 (%)
$H$	+1.103				-0.166				-0.937	
$H_x$	+0.126				-1.034				-1.552	
$(H-H_x)/H_x$	(+775)				-83.9				-39.6	(%)
$V$	+1.282				-2.551				+1.796	
$V_y$	+0.705				-3.573				+3.420	
$(V-V_y)/V_y$	+81.8				-28.6				-47.5	(%)

 (b) The 3rd-column-frame (multiplier :  $\mu$ )

Member	C F		F L		I L	
End	C	F	F	L	I	L
$M$	+11.916	+3.563	-3.563	+6.200	-5.841	-6.203
$M_x$	+8.772	+1.484	-0.196	+8.936	-9.099	-9.087
$(M-M_x)/M_x$	+35.8				-35.8	(%)
$H$	-2.866				+2.867	
$H_z$	-1.898				+4.331	
$(H-H_z)/H_z$	+51.0				-33.4	(%)
$V$	-0.527				+1.796	
$V_y$	-2.222				+3.420	
$(V-V_y)/V_y$	-76.3				-47.5	(%)

Fig. 31  $M_z$  diagramFig. 32  $M_x$  diagram

See, for example, Fig. 31. The effect of the vertical settlement at support I does not reach far away beyond the bay considered, and we find that the column at the left in the adjacent bay is affected very little as shown by full line. This tendency does not appear at all in the results obtained by the conventional solution which are shown by broken lines.

It should be especially noticed that the twisting moment of the remarkable magnitude is acting upon the member EF, see Fig. 34.



## Chapter IV. Analysis of Grid Works

20. A Square Grid Supported Simply at the Corners,  
Concentrated Load on the Center

A grid work<sup>6), 8)</sup> can be regarded as a rigid frame in space from which the columns are taken off. Therefore, the author's method is equally applicable to its analysis as before. The loads are assumed here, as usual, to act perpendicularly to the structural plane, hence the horizontal shear equations are not required.

The analyses of the grid works have been presented by many investigators. Among these, Prof. T. FUKUDA's exact method<sup>7)</sup> is the most famous in this country. On this account, the illustrative examples treated below are taken from his paper mainly to show the accuracies of the author's solutions. Fig. 35<sup>\*)</sup> shows the grid work to be considered. The condition of symmetry gives the following relations, that is, the grid is dealt with four unknowns.

$$\varphi : \begin{cases} \varphi_{xA} = \varphi_{xB} = \varphi_{zA} = \varphi_{zC} \equiv \varphi_1, \\ \varphi_{xC} = \varphi_{xD} = \varphi_{zB} = \varphi_{zD} = -\varphi_1, \\ \varphi_{xE} = \varphi_{xF} \equiv \varphi_2, \\ \varphi_{xI} = \varphi_{zH} = -\varphi_2. \end{cases}$$

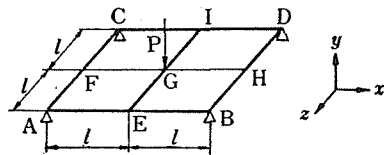


Fig. 35

$$\psi : \begin{cases} \psi_{xAF} = \psi_{xBH} = \psi_{zAE} = \psi_{zCI} \equiv \psi_1, \\ \psi_{xFC} = \psi_{xHD} = \psi_{zEB} = \psi_{zID} = -\psi_1, \\ \psi_{zEG} = \psi_{zFG} \equiv \psi_2, \\ \psi_{xGI} = \psi_{zGH} = -\psi_2. \end{cases}$$

## 1) Expressions of End-Moments

Due to the condition of symmetry, only the following end-moment expressions are required, where  $k_b$  and  $k_t$  mean the stiffness ratios for bending and for torsion respectively.

$$\begin{aligned} M_{zAE} &= k_b(2\varphi_1 + \psi_1), & M_{zAF} &= 2\beta k_t(\varphi_1 - \varphi_2), \\ M_{xEG} &= k_b(2\varphi_2 + \psi_2), & M_{zEA} &= k_b(\varphi_1 + \psi_1), & M_{xGE} &= k_b(\varphi_2 + \psi_2). \end{aligned}$$

The remainders are expressed by these:

$$M_{xAE} = M_{zAF}, \quad M_{xAF} = M_{zAE}, \quad M_{xBE} = M_{zAF}, \text{ etc.}$$

\*) FUKUDA: II Abschnitt, § 2, 2)

## 2) Elastic Equations

## i) Joint Equilibrium Equations

$$\text{At joint A, } \sum M_{xA} = M_{xAE} + M_{xAf} = 0.$$

$$\text{„ E, } \sum M_{xE} = M_{xEA} + M_{xEb} + M_{xEg} = 0.$$

## ii) Vertical Shear Equations

$$\text{At joint E, } \sum X_{yE} = X_{yEA} - X_{yEb} - X_{yEG} = 0.$$

$$\text{„ G, } \sum X_{yG} = X_{yGE} + X_{yGF} - X_{yGH} - X_{yGI} = P,$$

where,

$$\begin{aligned} X_{yEA} &= -(1/l)(M_{zEA} + M_{zAE}), & X_{yEb} &= -(1/l)(M_{zEb} + M_{zBE}), \\ X_{yEG} &= -(1/l)(M_{zEG} + M_{zGE}), & X_{yGE} &= -(1/l)(M_{zGE} + M_{zEG}), \\ X_{yGF} &= -(1/l)(M_{zGF} + M_{zFG}), & X_{yGH} &= -(1/l)(M_{zGH} + M_{zHG}), \\ X_{yGI} &= -(1/l)(M_{zGI} + M_{zIG}). \end{aligned}$$

Substituting the end-moments in 1), we have the simultaneous equations:

$$(2k_b + 2\beta k_t)\varphi_1 - 2\beta k_t\varphi_2 + k_b\phi_1 = 0,$$

$$4\beta k_t\varphi_1 - (2k_b + 4\beta k_t)\varphi_2 - k_b\phi_2 = 0,$$

$$6\varphi_1 - 3\varphi_2 + 4\phi_1 - 2\phi_2 = 0,$$

$$12\varphi_2 + 8\phi_2 = -(l/k_b)P.$$

Solution gives :

$$\begin{aligned} \varphi_1 &= \frac{Pl}{8k_b} \cdot \frac{k_b + 16\beta k_t}{k_b + 12\beta k_t}, & \varphi_2 &= \frac{Pl}{4k_b} \cdot \frac{k_b + 8\beta k_t}{k_b + 12\beta k_t}, \\ \phi_1 &= -\frac{Pl}{4k_b} \cdot \frac{k_b + 15\beta k_t}{k_b + 12\beta k_t}, & \phi_2 &= -\frac{Pl}{2k_b} \cdot \frac{k_b + 9\beta k_t}{k_b + 12\beta k_t}. \end{aligned}$$

## 3) Values of End-Moments

Using these, we have from 1) :

$$M_{zAE} = \frac{Pl}{4} \cdot \frac{1}{k_b + 12\beta k_t} (k_b + 16\beta k_t - k_b - 15\beta k_t) = \frac{Pl}{4} \cdot \frac{\beta k_t}{k_b + 12\beta k_t},$$

$$M_{zEA} = \frac{Pl}{8} \cdot \frac{1}{k_b + 12\beta k_t} (k_b + 16\beta k_t - 2k_b - 30\beta k_t) = -\frac{Pl}{8} \cdot \frac{k_b + 14\beta k_t}{k_b + 12\beta k_t},$$

$$M_{zGE} = \frac{-Pl}{4} \cdot \frac{1}{k_b + 12\beta k_t} (-k_b - 8\beta k_t + 2k_b + 18\beta k_t) = -\frac{Pl}{4} \cdot \frac{k_b + 10\beta k_t}{k_b + 12\beta k_t},$$

$$M_{xEG} = \frac{-Pl}{2} \cdot \frac{1}{k_b + 12\beta k_t} (-k_b - 8\beta k_t + k_b + 9\beta k_t) = -\frac{Pl}{2} \cdot \frac{\beta k_t}{k_b + 12\beta k_t},$$

$$M_{zAF} = \frac{Pl}{4} \cdot \frac{\beta k_t}{k_b(k_b + 12\beta k_t)} \cdot (k_b + 16\beta k_t - 2k_b - 16\beta k_t) = -\frac{Pl}{4} \cdot \frac{\beta k_t}{k_b + 12\beta k_t}.$$

Prof. FUKUDA's values are as follows, where  $\alpha = EI/GH$  and  $H$  represents the coefficient of twisting resistance; note that in his analysis eight unknowns being employed.

$$M_{zAE} = \frac{Pl}{16(\alpha+3)}, \quad M_{zEA} = -\frac{Pl(2\alpha+7)}{16(\alpha+3)},$$

$$M_{xGE} = -\frac{Pl(2\alpha+5)}{8(\alpha+3)}, \quad M_{xEG} = -\frac{Pl}{8(\alpha+3)},$$

$$M_{zAF} = -\frac{Pl}{16(\alpha+3)}.$$

The relation between  $\alpha$  and the author's factor  $\beta$  is:

$$\begin{aligned} \alpha = \frac{EI}{GH} &= \frac{2(m+1)}{m} \cdot \frac{I}{H} = \frac{2(m+1)}{m} \cdot \frac{\frac{I}{l}}{\frac{H}{l}} = \frac{2(m+1)}{m} \cdot \frac{k_b}{k_t} \\ &= \frac{1}{4} \cdot \frac{8(m+1)}{m} \cdot \frac{k_b}{k_t} = \frac{1}{4} \frac{k_b}{\beta k_t}. \\ \therefore \alpha &= \frac{k_b}{4\beta k_t}. \end{aligned}$$

Referring to this relation, it is seen that Prof. FUKUDA's solutions just come to the authors' except signs. It is to be noticed that the author's method needs only four independent unknowns, half of the FUKUDA's method, and that the slope-deflection method is a powerful means also in dealing with the grid works. The

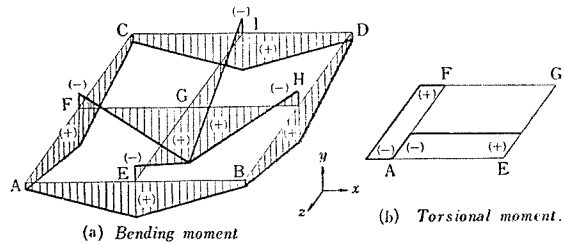


Fig. 36 Moment diagram

moment diagrams are shown in Fig. 36.

21. A Square Grid Supported Simply at the Corners,  
Concentrated Load on an Edge Joint. Fig. 37.\*)

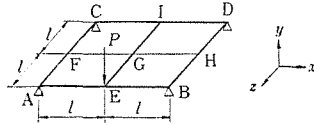


Fig. 37

From symmetry, we have the following nine  $\varphi$ 's and four  $\psi$ 's as unknowns:

$$\varphi : \left\{ \begin{array}{l} \text{At joint A, } \varphi_{xA}, \varphi_{zA} \\ \text{'' E, } \varphi_{xE} \\ \text{'' F, } \varphi_{xF}, \varphi_{zF} \\ \text{'' G, } \varphi_{xG} \\ \text{'' C, } \varphi_{xC}, \varphi_{zC} \\ \text{'' I, } \varphi_{xI} \end{array} \right. \quad \psi : \left\{ \begin{array}{l} \text{For member AE, } \psi_{xAE} \\ \text{'' FG, } \psi_{zFG} \\ \text{'' CI, } \psi_{zCI} \\ \text{'' AF, } \psi_{xAf} \end{array} \right.$$

The remainders are expressed by these :

$$\varphi_{xH} = \varphi_{xF}, \quad \varphi_{zD} = -\varphi_{zC}, \quad \text{etc.},$$

and

$$\psi_{xEG} = \psi_{xAf} + \psi_{zFG} - \psi_{xAe}, \quad \psi_{xGI} = -\psi_{zFG} - \psi_{xAf} + \psi_{zCI}, \quad \text{etc.}$$

1) Expressions of End-Moments

$$\begin{array}{ll} \text{At joint A:} & \left\{ \begin{array}{l} M_{xAf} = kb(2\varphi_{xA} + \varphi_{xF} + \psi_{xAf}), \\ M_{zAf} = 2\beta k_l(\varphi_{zA} - \psi_{zF}), \end{array} \right. \quad \left\{ \begin{array}{l} M_{xAB} = 2\beta k_l(\varphi_{xA} - \psi_{xE}), \\ M_{zAE} = kb(2\varphi_{zA} + \psi_{zAE}). \end{array} \right. \\ \text{'' B:} & \left\{ \begin{array}{l} M_{xBH} = M_{xAf}, \\ M_{zBH} = -M_{zAf}, \end{array} \right. \quad \left\{ \begin{array}{l} M_{xBE} = M_{xAe}, \\ M_{zBE} = -M_{zAE}. \end{array} \right. \\ \text{'' C:} & \left\{ \begin{array}{l} M_{xCf} = kb(2\varphi_{xC} + \varphi_{xF} - \psi_{xAf}), \\ M_{zCf} = -M_{zFC}, \end{array} \right. \quad \left\{ \begin{array}{l} M_{xCI} = 2\beta k_l(\varphi_{xC} - \varphi_{xI}), \\ M_{zCI} = kb(2\varphi_{zC} + \psi_{zCI}). \end{array} \right. \\ \text{'' D:} & \left\{ \begin{array}{l} M_{xDH} = M_{xCf}, \\ M_{zDH} = -M_{zCf}, \end{array} \right. \quad \left\{ \begin{array}{l} M_{xDI} = M_{xCI}, \\ M_{zDI} = -M_{zCI}. \end{array} \right. \end{array}$$

\* ) FUKUDA : II Abschnitt, §2, 4)

$$\begin{aligned}
\text{At joint E: } & \begin{cases} M_{xEA} = -M_{xAE}, & M_{xEB} = -M_{xAE}, \\ M_{xEG} = kb(2\varphi_{xE} + \varphi_{xG} + \phi_{xAF} + \phi_{zFG} - \phi_{zAE}), \\ M_{zEA} = kb(\varphi_{zA} + \phi_{zAE}), & M_{zEB} = -M_{zEA}, & M_{zEG} = 0. \end{cases} \\
// \quad \text{F: } & \begin{cases} M_{xFA} = kb(2\varphi_{xF} + \varphi_{xA} + \phi_{xAF}), & M_{xFC} = kb(2\varphi_{xF} + \varphi_{xC} - \phi_{xAF}), \\ M_{xFG} = 2\beta k_l(\varphi_{xF} - \varphi_{xG}), \\ M_{zFA} = -M_{zAF}, & M_{zFC} = 2\beta k_l(\varphi_{zF} - \varphi_{zC}), & M_{zFG} = kb(2\varphi_{zF} + \phi_{zFG}). \end{cases} \\
// \quad \text{G: } & \begin{cases} M_{xGE} = kb(2\varphi_{xG} + \varphi_{xE} + \phi_{xAF} + \phi_{zFG} - \phi_{zAE}), \\ M_{xGF} = -M_{xFG}, & M_{xGH} = -M_{xFG}, \\ M_{xGI} = kb(2\varphi_{xG} + \varphi_{xI} - \phi_{zFG} - \phi_{xAF} + \phi_{zCI}), \\ M_{zGE} = 0, & M_{zGF} = kb(\varphi_{zF} + \phi_{zFG}), \\ M_{zGH} = -M_{zGF}, & M_{zGI} = 0. \end{cases} \\
// \quad \text{H: } & \begin{cases} M_{xHB} = M_{xFA}, & M_{xHD} = M_{xFC}, & M_{xHG} = M_{xFG}, \\ M_{zHB} = -M_{zFA}, & M_{zHD} = -M_{zFC}, & M_{zHG} = -M_{zFG}. \end{cases} \\
// \quad \text{I: } & \begin{cases} M_{xIC} = -M_{xCi}, & M_{xID} = -M_{xCi}, \\ M_{xIG} = kb(2\varphi_{xI} + \varphi_{xG} - \phi_{zFG} - \phi_{xAF} + \phi_{zCI}), \\ M_{zIC} = kb(\varphi_{zC} + \phi_{zCI}), & M_{zID} = -M_{zIC}, & M_{zIG} = 0. \end{cases}
\end{aligned}$$

## 2) Elastic Equations

## i) Joint Equilibrium Equations

$$\text{About } x \text{ axis: } \begin{cases} \sum M_{xA} = M_{xAF} + M_{xAE} = 0, \\ \sum M_{xC} = M_{xCF} + M_{xCi} = 0, \\ \sum M_{xE} = M_{xEA} + M_{xEB} + M_{xEG} = 0, \\ \sum M_{xF} = M_{xFA} + M_{xFC} + M_{xFG} = 0, \\ \sum M_{xG} = M_{xGF} + M_{xGH} + M_{xGE} + M_{xGI} = 0, \\ \sum M_{xI} = M_{xIC} + M_{xID} + M_{xIG} = 0. \end{cases}$$

$$\text{About } z \text{ axis: } \begin{cases} \sum M_{zA} = M_{zAE} + M_{zAF} = 0, \\ \sum M_{zC} = M_{zCF} + M_{zCi} = 0, \\ \sum M_{zF} = M_{zFA} + M_{zFC} + M_{zFG} = 0. \end{cases}$$

## ii) Vertical Shear Equations

At joint E:  $X_{yEA} - X_{yEB} - X_{yEG} - P = 0$ .

// F:  $X_{yFA} - X_{yFC} - X_{yFG} = 0$ .

// G:  $X_{yGF} - X_{yGH} + X_{yGE} - X_{yGI} = 0$ .

// I:  $X_{yIC} - X_{yID} + X_{yIG} = 0$ .

where,

$$\begin{aligned}
 X_{yEA} &= -(1/l)(M_{zAE} + M_{zEA}), & X_{yEB} &= -(1/l)(M_{zEB} + M_{zBE}), \\
 X_{yEG} &= -(1/l)(M_{xEG} + M_{xGE}), & X_{yFA} &= -(1/l)(M_{xAF} + M_{xFA}), \\
 X_{yFC} &= -(1/l)(M_{xFC} + M_{xCF}), & X_{yFG} &= -(1/l)(M_{zFG} + M_{zGF}), \\
 X_{yGF} &= -(1/l)(M_{zFG} + M_{zGF}), & X_{yGH} &= -(1/l)(M_{zGH} + M_{zHG}), \\
 X_{yGE} &= -(1/l)(M_{xEG} + M_{xGE}), & X_{yGI} &= -(1/l)(M_{xGI} + M_{xIG}), \\
 X_{yIC} &= -(1/l)(M_{zCI} + M_{zIC}), & X_{yID} &= -(1/l)(M_{zID} + M_{zDI}), \\
 X_{yIG} &= -(1/l)(M_{xGI} + M_{xIG}).
 \end{aligned}$$

Using expressions in 1), we have the simultaneous equations shown in Table 24, which produces the following solutions :

**Table 24** Elastic equations

Eq.	Left-hand side													Right-hand side
	$\varphi xA$	$\varphi xC$	$\varphi xE$	$\varphi xF$	$\varphi xG$	$\varphi xI$	$\varphi zA$	$\varphi zC$	$\varphi zF$	$\psi xAF$	$\psi zAE$	$\psi zCI$	$\psi zFG$	
1	$2a+1$		-1	$a$						$a$				0
2		$2a+1$		$a$		-1				$-a$				0
3	-2		$2a+2$		$a$					$a$	$-a$		$a$	0
4	$a$	$a$		$4a+1$	-1									0
5			$a$	-2	$4a+2$	$a$					$-a$	$a$		0
6		-2			$a$	$2a+2$				$-a$		$a$	$-a$	0
7							$2a+1$		-1		$a$			0
8								$2a+1$	-1			$a$		0
9							-1	-1	$2a+2$				$a$	0
10			3		3		-6			2	-6		2	$Pl/k_b$
11	3	3							3	-4			2	0
12			3			3			6	-4	2	2	-8	0
13					3	3		6		-2		6	-2	0

$$\begin{aligned}
\varphi_{xA} &= n(-6 - 15a - 5a^2 + 4a^3 + a^4), \\
\varphi_{xC} &= -n(6 + 23a + 25a^2 + 8a^3 + a^4), \\
\varphi_{xE} &= -n(6 + 45a + 86a^2 + 46a^3 + 6a^4), \\
\varphi_{xF} &= -n(6 + 13a + 2a^2), \\
\varphi_{xG} &= -n(6 + 49 + 92a^2 + 38a^3 + 4a^4), \\
\varphi_{xI} &= -n(6 + 29a + 46a^2 + 22a^3 + 2a^4), \\
\varphi_{zA} &= n(96 + 314a + 311a^2 + 108a^3 + 11a^4), \\
\varphi_{zC} &= n(96 + 206a + 77a^2 - a^4), \\
\varphi_{zF} &= n(96 + 248a + 164a^2 + 36a^3 + 2a^4), \\
\phi_{xAF} &= -n(12 + 38a + 38a^2 + 15a^3 + 2a^4), \\
\phi_{xAE} &= -n(258 + 775a + 694a^2 + 225a^3 + 22a^4), \\
\phi_{xCi} &= -n(150 + 325a + 118a^2 - 3a^3 - 2a^4), \\
\phi_{zFG} &= -n(168 + 436a + 292a^2 + 66a^3 + 4a^4),
\end{aligned}$$

where,

$$\begin{aligned}
\mu &= Pl/24(2 + a)(6 + a)(2 + 4a + a^2), \\
n &= \mu/kb.
\end{aligned}$$

### 3) Moments and Reactions

From eqs. (a) and (f), end-moments are determined.

$$\begin{aligned}
\text{At joint A: } & \begin{cases} M_{xAF} = -\mu(30 + 81a + 50a^2 + 7a^3), \\ M_{xAE} = \mu(30 + 81a + 50a^2 + 7a^3), \\ M_{zAF} = 3\mu(22 + 49a + 24a^2 + 3a^3), \\ M_{zAE} = -3\mu(22 + 49a + 24a^2 + 3a^3). \end{cases} \\
\text{C: } & \begin{cases} M_{xCF} = -\mu(6 + 21a + 14a^2 + a^3), \\ M_{xCi} = \mu(6 + 21a + 14a^2 + a^3), \\ M_{zCi} = 3\mu(14 + 29a + 12a^2 + a^3). \end{cases} \\
\text{E: } & \begin{cases} M_{xEG} = 2\mu(30 + 81a + 50a^2 + 7a^3), \\ M_{zEA} = -\mu(162 + 461a + 383a^2 + 117a^3 + 11a^4). \end{cases}
\end{aligned}$$

$$\text{At joint F: } \begin{cases} M_{xFA} = -\mu(30 + 79a + 47a^2 + 11a^3 + a^4), \\ M_{xFC} = \mu(-6 - 11a + 9a^2 + 7a^3 + a^4), \\ M_{xFG} = 2\mu(18 + 45a + 19a^2 + 2a^3), \\ M_{zFC} = 3\mu(14 + 29a + 12a^2 + a^3), \\ M_{zFG} = 6\mu(4 + 10a + 6a^2 + a^3). \end{cases}$$

$$\text{G: } \begin{cases} M_{xGE} = 2\mu(30 + 79a + 47a^2 + 11a^3 + a^4), \\ M_{xGI} = -2\mu(-6 - 11a + 9a^2 + 7a^3 + a^4), \\ M_{zGF} = -2\mu(36 + 94a + 64a^2 + 15a^3 + a^4). \end{cases}$$

$$\text{I: } \begin{cases} M_{xIG} = 2\mu(6 + 21a + 14a^2 + a^3), \\ M_{zIC} = -\mu(54 + 119a + 41a^2 - 3a^3 - a^4). \end{cases}$$

Putting  $a = 2\alpha$ , we see that these are in agreement with Prof. FUKUDA'S solutions which are obtained from twenty-five equations. Consider now the case where all members are of steel ( $m=4$ ) with square sections, i.e.,  $a = 2.992, 6$  and  $\mu = 0.000, 042, 670 \text{ Pl}$ . Then we have:

$$\text{At joint A: } \begin{cases} M_{xAF} = -0.037, 06 \text{ Pl} & (\text{B}) \\ M_{xAE} = +0.037, 06 \text{ Pl} & (\text{T}) \\ M_{zAF} = +0.056, 98 \text{ Pl} & (\text{T}) \\ M_{zAE} = -0.056, 98 \text{ Pl} & (\text{B}) \end{cases}$$

$$\text{C: } \begin{cases} M_{xCF} = -0.009, 04 \text{ Pl} & (\text{B}) \\ M_{xCI} = +0.009, 04 \text{ Pl} & (\text{T}) \\ M_{zCI} = +0.028, 96 \text{ Pl} & (\text{B}) \end{cases}$$

$$\text{E: } \begin{cases} M_{xEG} = +0.074, 12 \text{ Pl} & (\text{B}) \\ M_{zEA} = -0.362, 87 \text{ Pl} & (\text{B}) \end{cases}$$

$$\text{F: } \begin{cases} M_{xFA} = -0.043, 09 \text{ Pl} & (\text{B}) \\ M_{xFC} = +0.012, 22 \text{ Pl} & (\text{B}) \\ M_{xFG} = +0.030, 87 \text{ Pl} & (\text{T}) \\ M_{zFC} = +0.028, 96 \text{ Pl} & (\text{T}) \\ M_{zFG} = +0.028, 02 \text{ Pl} & (\text{B}) \end{cases}$$



$$\begin{aligned}
 // \quad G : & \begin{cases} M_{xGE} = +0.086, 18Pl & (B) \\ M_{xGI} = -0.024, 44Pl & (B) \\ M_{zGF} = -0.111, 35Pl & (B) \end{cases} \\
 // \quad I : & \begin{cases} M_{xIG} = +0.018, 09Pl & (B) \\ M_{zIC} = -0.025, 78Pl & (B) \end{cases}
 \end{aligned}$$

Note: (B): Bending moment, (T): Torsional moment.

These coincide with the values shown by Prof. FUKUDA except signs. Fig. 38 shows the moment diagram.

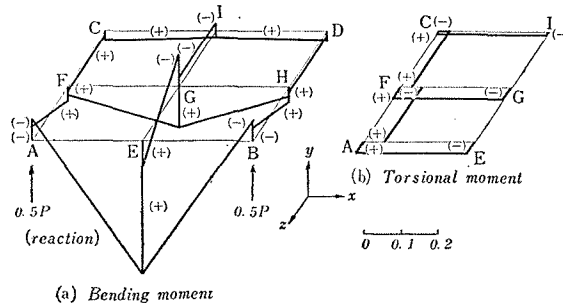


Fig. 38 Moment diagram (multiplier :  $Pl$ )

Vertical reactions are calculated as follows :

$$\begin{aligned}
 A &= -(1/l)(M_{xAF} + M_{xFA}) - (1/l)(M_{zAE} + M_{zEA}) \\
 &= -(1/l)(-0.037, 06 - 0.043, 09)Pl - (1/l)(-0.056, 98 - 0.362, 87)Pl = 0.5P, \\
 C &= (1/l)(M_{xFC} + M_{xCF}) - (1/l)(M_{zCI} + M_{zIC}) \\
 &= (1/l)(0.012, 22 - 0.009, 04)Pl - (1/l)(0.028, 96 - 0.025, 78)Pl = 0.
 \end{aligned}$$

#### 4) Deflections

Obtaining end-moments and  $\phi$ 's, we readily determine the deformation of the structure, which are shown in Fig. 39. It will be worth while to mention that this diagram gives the influence diagram for the deflection at the joint considered if we put  $P = 1$ .

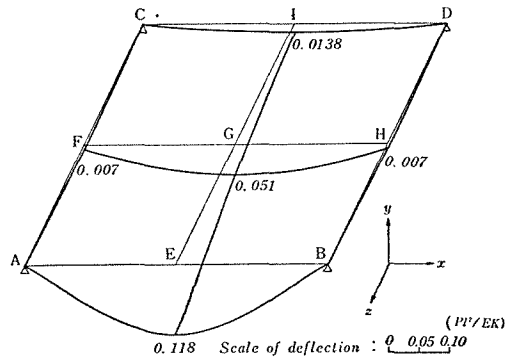


Fig. 39 Deflection diagram

22. A Square Grid of  $3 \times 3$  Panels Supported Simply at Corners, Concentrated Loads on the Inner Joints. Fig. 40.\*)

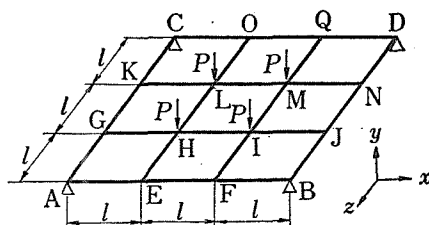


Fig. 40

By symmetry, the following six unknowns are enough.

$$\varphi : \begin{cases} \varphi_{zA} = \varphi_{xA} \equiv \varphi_1, \\ \varphi_{zE} = \varphi_{xG} \equiv \varphi_2, \\ \varphi_{xE} = \varphi_{zG} \equiv \varphi_3, \\ \varphi_{zH} = \varphi_{xH} \equiv \varphi_4. \end{cases} \quad \psi : \begin{cases} \psi_{zAE} = \psi_{xAG} \equiv \psi_1, \\ \psi_{xEH} = \psi_{zGH} \equiv \psi_2, \\ (\psi_{zEF} = \psi_{xGK} = \psi_{zHI} = \psi_{zHL} = 0). \end{cases}$$

1) Expressions of End-Moments

$$\begin{aligned} \text{At joint A: } & M_{zAE} = kb(2\varphi_1 + \varphi_2 + \psi_1), \quad M_{xAG} = 2\beta k l(\varphi_1 - \varphi_3). \\ \text{" E: } & \begin{cases} M_{zEA} = kb(\varphi_1 + 2\varphi_2 + \psi_1), & M_{xEA} = 2\beta k l(\varphi_3 - \varphi_1), \\ M_{zEF} = kb\varphi_2, & M_{zEH} = 2\beta k l(\varphi_2 - \varphi_4), \\ M_{xEH} = kb(2\varphi_3 + \varphi_4 + \psi_2), & M_{xEF} = 0. \end{cases} \\ \text{" H: } & \begin{cases} M_{zHG} = kb(\varphi_3 + 2\varphi_4 + \psi_2), & M_{zHE} = 2\beta k l(\varphi_4 - \varphi_2), \\ M_{zHI} = kb\varphi_4, & M_{zHL} = 0, & M_{xHE} = kb(\varphi_3 + 2\varphi_4 + \psi_2). \end{cases} \\ \text{" F: } & M_{zFE} = -kb\varphi_2. \\ \text{" I: } & M_{zIH} = -kb\varphi_4. \end{aligned} \quad (a)$$

2) Elastic Equations

i) Joint Equilibrium Equations

$$\begin{aligned} \text{At joint A, } & \sum M_{zA} = M_{zAE} + M_{zAG} = 0. \\ \text{" E, } & \begin{cases} \sum M_{zE} = M_{zEA} + M_{zEF} + M_{zEH} = 0, \\ \sum M_{xE} = M_{xEA} + M_{xEF} + M_{xEH} = 0. \end{cases} \\ \text{" H, } & \sum M_{zH} = M_{zHG} + M_{zHE} + M_{zHI} + M_{zHL} = 0. \end{aligned} \quad (b)$$

\*) FUKUDA: II Abschnitt, §3, 2)

## ii) Vertical Shear Equations

$$\left. \begin{aligned} \text{At joint E, } \sum X_{yE} &= X_{yEA} - X_{yEH} - X_{yEF} = 0, \\ \text{H, } \sum X_{yH} &= X_{yHG} + X_{yHE} - X_{yHI} - X_{yHL} = P, \end{aligned} \right\} \quad (c)$$

where,

$$\left. \begin{aligned} X_{yEA} &= -(1/l)(M_{zAE} + M_{zEA}), & X_{yEH} &= -(1/l)(M_{xEH} + M_{xHE}), \\ X_{yEF} &= -(1/l)(M_{zEF} + M_{zFE}), & X_{yHG} &= -(1/l)(M_{zGH} + M_{zHG}), \\ X_{yHE} &= -(1/l)(M_{xHE} + M_{xEH}), & X_{yHI} &= -(1/l)(M_{zHI} + M_{zIH}), \\ X_{yHL} &= -(1/l)(M_{xHL} + M_{xLH}). \end{aligned} \right\} \quad (d)$$

Here, we have the equations shown in Table 25, which produces the following results :

Table 25 Elastic equations

Eq.	Left-hand side						Right-hand side
	$\varphi_1$	$\varphi_2$	$\varphi_3$	$\varphi_4$	$\phi_1$	$\phi_2$	
(1)	$2(kb + \beta kt)$	$kb$	$-2\beta kt$		$kb$		0
(2)	$kb$	$3kb + 2\beta kt$		$-2\beta kt$	$kb$		0
(3)	$-2\beta kt$		$2(kb + \beta kt)$	$kb$		$kb$	0
(4)		$-2\beta kt$	$kb$	$3kb + 2\beta kt$		$kb$	0
(5)	3	3	-3	-3	2	-2	0
(6)			3	3		2	$-(Pl/2kb)$

$$\left. \begin{aligned} \varphi_1 &= \frac{Pl}{kb}, & \varphi_2 &= \frac{Pl}{2kb}, & \varphi_3 &= \frac{Pl}{kb}, \\ \varphi_4 &= \frac{Pl}{2kb}, & \phi_1 &= -\frac{5Pl}{2kb}, & \phi_2 &= -\frac{5Pl}{2kb}. \end{aligned} \right\} \quad (e)$$

## 3) End-Moments

Eqs. (a) and (e) determine the end-moments;

$$\left. \begin{aligned} M_{zAE} &= kb \left( \frac{2Pl}{kb} + \frac{Pl}{2kb} - \frac{5Pl}{2kb} \right) = 0, \\ M_{zEA} &= kb \left( \frac{Pl}{kb} + \frac{2Pl}{2kb} - \frac{5Pl}{2kb} \right) = -\frac{1}{2}Pl, \end{aligned} \right\}$$

$$\begin{aligned}
 M_{zEF} &= k_b \left( \frac{Pl}{2k_b} \right) = \frac{1}{2} Pl, \\
 M_{xEH} &= k_b \left( \frac{2Pl}{k_b} + \frac{Pl}{2k_b} - \frac{5Pl}{2k_b} \right) = 0, \\
 M_{zHI} &= k_b \frac{Pl}{2k_b} = \frac{1}{2} Pl, \\
 M_{zAG} &= 2\beta k_t \left( \frac{Pl}{k_b} - \frac{Pl}{k_b} \right) = 0, \\
 M_{zEH} &= 2\beta k_t \left( \frac{Pl}{2k_b} - \frac{Pl}{2k_b} \right) = 0.
 \end{aligned} \tag{f}$$

These coincide again with Prof. FUKUDA's solutions which he found from thirteen equations.

In this case, as Prof. FUKUDA points out, all the torsional moments vanish and hence the girders act as if they were simply supported at their extreme ends; refer to Fig. 41.

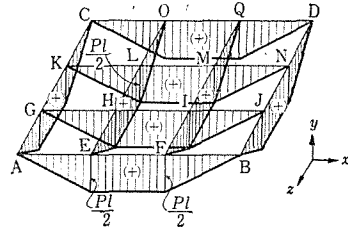


Fig. 41 Moment diagram

23. *A Square Grid of  $3 \times 3$  Panels Fixed at the Periphery, Concentrated Loads on the Inner Joints. Fig. 42.\*)*

This problem is rather simple to analyse than the preceding, because the independent unknowns are only two, i. e.,

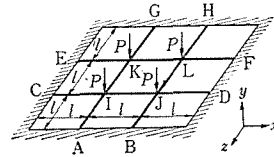


Fig. 42

$$\varphi_{zI} = \varphi_{zJ} = \varphi_{zK} = \varphi_{zL} = \varphi_{xI} = \varphi_{xJ} = \varphi_{xK} = \varphi_{xL} \equiv \varphi,$$

$$\phi_{xAI} = \phi_{zCI} \equiv \phi.$$

Thus we have the end-moment expressions :

$$M_{zCI} = k_b(\varphi + \phi), \quad M_{zIC} = k_b(2\varphi + \phi),$$

$$M_{zIA} = 2\beta k_t \varphi, \quad M_{zIJ} = k_b \varphi, \quad M_{zIK} = 0.$$

\*) FUKUDA: II Abschnitt, § 4, 2)

These must satisfy the equilibrium conditions below:

$$\sum M_{zI} = M_{zIC} + M_{zIA} + M_{zIJ} + M_{zIK} = 0,$$

$$\sum X_{yI} = X_{yIC} + X_{yIA} - X_{yIJ} - X_{yIK} = P,$$

which give the following simultaneous equations :

$$(3k_b + 2\beta k_t)\varphi + k_b\psi = 0,$$

$$3k_b\varphi + 2k_t\psi + \frac{Pl}{2} = 0.$$

Solution gives

$$\varphi = \frac{Pl}{2k_b} \cdot \frac{\alpha}{3\alpha + 1}, \quad \psi = \frac{Pl}{4k_b} \cdot \frac{6\alpha + 1}{3\alpha + 1},$$

where

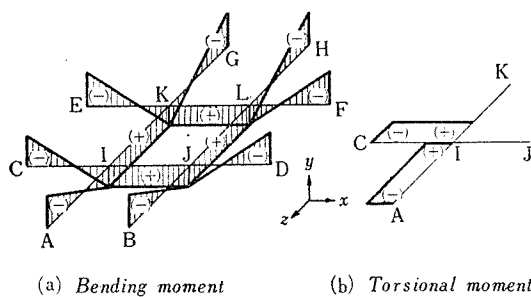
$$\alpha = k_b/4\beta k_t.$$

Introducing these in eqs. (a), the end-moments are determined :

$$M_{zCI} = -\frac{Pl}{4} \cdot \frac{4\alpha + 1}{3\alpha + 1}, \quad M_{zIC} = -\frac{Pl}{4} \cdot \frac{2\alpha + 1}{3\alpha + 1},$$

$$M_{zIJ} = \frac{Pl}{2} \cdot \frac{\alpha}{3\alpha + 1}, \quad M_{zIA} = \frac{Pl}{4} \cdot \frac{1}{3\alpha + 1}.$$

These results agree with those of Prof. FUKUDA which are obtained by solving seven equations simultaneously. Fig. 43 shows the moment diagram.



**Fig. 43** Moment diagram

### Summary

Some of the features of the foregoing investigations may be summarized as follows:

(1) To analyse the rigid frames in space, the so-called classical methods which take the stress functions for unknowns are considered not to be useful in practice, because they require numerous condition equations. In addition, we are very often confused in drawing moment diagrams, because those methods require to pay constant attentions in reading the sign conventions adopted. To this, the author's method, depending upon the slope-deflection theory, requires far smaller number of condition equations, about half of those of the classical methods, because it takes end deformations for unknowns, and the moment diagrams are mechanically drawn without confusion.

(2) For the frames without sways, the current two-dimensional analysis may be applied with good accuracy except the cases in which there exist large unbalances among the stiffnesses of members; for such exceptional cases we must treat the frame three-dimensionally.

(3) For the frames with sways, which is the usual case, the conventional two-dimensional treatment produces results containing large errors; the errors will appear either on the safe side or on the dangerous side due to the arrangement of members and loading conditions. Hence, these frames must be analysed three-dimensionally.

(4) The author's method is successfully applied to the analysis of the grid works.

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